ATTITUDE DETERMINATION & CONTROL		
select attitude control concept	1a	2009 - April
dumping of Angular momentum, calculate force		2009 - April
PD control law - P control law		2009 - April
Graph of attitude vs time		2009 - April
minimum thrust force		2008 - March
effect of gravity gradient disturbance torque	4b	2008 - March
can gps or galilieo determine attitude of Iss	4c	2008 - March
thrust needed/ angular momentum	5a	2008 - January
calculate torque needed	5b	2008 - January
discuss stability of attitude	5c	2008 - January
maximum distubance torque gravity gradient	2a	2007 - February
can it be used?	2b	2007 - February
how long does slew manouvre take	2c	2007 - February
draw PD control system	2d	2007 - February
possible only to have gyroscopes as sensors?	2e	2007 - February
force needed to dump angular momentum	3a	2006 - April
equation of PD control law	3b	2006 - April
what criteria apply to make closed loop system stable		2006 - April
draw block diagram for pitch axis closed loop system	3d	2006 - April
5 controlers to be investigated/analysed: find best controler	2	2005 - April
mass moment of inertia	4a	2005 - January
rotational rate about x-axis	4b	2005 - January
rotational rate after deployment	4c	2005 - January
maximum distubance torque on the spacecraft	4a	2003 - August
torque needed to allow the slew maneuver	4b	2003 - August

### **STRUCTURES & MECHANISMS**

3a 2009 - April
3b 2009 - April
3c 2009 - April
3d 2009 - April
5a 2003 - August
5b 2003 - August
5c 2003 - August
5d 2003 - August
5e 2003 - August

#### THERMAL CONTROL/THERMAL DESIGN

thermal conductivity ratio alpha/epsilon of radiator coating	
solar reflection factor of SC	
maximum fraction of each element covered by SC if	
solar radiant flux density	
maximum temperature of the foil	
what uncertanty margines in thermal analysis?	
ratio of alpha/epsilon with uncertainty margin	
effective radiative coupling of reflector dish to space	
effective solar absorptance	
establish heat balance/ calculate temperature	
maximum temerature of dish	

1a	2009 - January
1b	, 2009 - January
1c	2009 - January
1d	2009 - January
5a	2007 - February
5b	2007 - February
5c	2007 - February
5d	2007 - February
1a	2006 - April
1b	2006 - April
1c	2006 - April
1d	2006 - April

define optimum thermal layout of bottom surface/justify design	3	2005 - April
heat balance/ investigate feasibility from thermal point of view	1a	2005 - January
draft a thermal network	1b	2005 - January
conduction	1c	2005 - January
radiative couplings to space	1d	2005 - January
linearise conductive coupling to radiative coupling	1e	2005 - January
power absorbed	1f	2005 - January
temperature	1g	2005 - January
temperature	1h	2005 - January
gradient	1i	2005 - January
heat flow into tank	2a	2004 - January
hat flow into payload module	2b	2004 - January
which is another important heat flow?	2c	2004 - January
total heat flow into payload module	2d	2004 - January
evaporation rate of helium	2e	2004 - January
how many days until helium is evaporated	2f	2004 - January
any ideas of significant impove of the thermal design?	2g	2004 - January
solar absorptivity at 25 °	6a	2003 - August
solar absorbtivity at 150°	6b	2003 - August
fraction of OSRs and fraction of solar cells on solar array		2003 - August
tilt angle to keep temperature at 150°		2003 - August
thermal design of the illuminated panel side edge		2003 - August

# COST RISK RELIABILITY

calculate SE		2009 - April
RDT&E and TFU		2009 - April
fullfils required reliability with probablilty of 90%?		2009 - April
determine reliabiity		2009 - January
discuss the assumption of constant failure rate	5b	2009 - January
system reliabiity	5c	2009 - January
determine standard error	5a	2008 - March
calculate total cost	5b	2008 - March
average cost price of single unit	5c	2008 - March
mean time to failure	7a	2008 - January
probablity of 15 week survial	7b	2008 - January
discuss two approches to increase reliability	7c	2008 - January
determine availibility		2008 - January
Standard error for summed RDT&E cost	5b	2007 - March
average time to build 5 spacecraft	5d	2007 - March
mean time to failure	4a	2007 - February
reliability of this computer system	4b	2007 - February
calculate percentage margin	4c	2007 - February
average time to build spacecraft	4d	2007 - February
calculate standard error	4e	2007 - February
various reliabilities to ensure total reliability of 0.9	5a	2006 - April
failure rate pf power subsystem	5b	2006 - April
reliability of propulsion subsystem	5c	2006 - April
reliability of subsystem/	5d	2006 - April
reliability of subsystem/		2006 - April
MTTF for structure subsystem/minimum test length	5f	2006 - April
reliabilty of solar cell/blocking diode	3a	2005 - January
reliability of single string	3b	2005 - January
reliability of single solar panel	Зс	2005 - January
probability of successful deployment	3d	2005 - January

probability of successfull power provision		2005 - January
discuss effect of solar cell open circuit failure	3f	2005 - January
probability of successfull power provision two more strings	3g	2005 - January
total mission cost	3a	2003 - August
probability of complete success	3b	2003 - August
associated risk	3c	2003 - August
overall launch success probability	3d	2003 - August
do 3d again with other data	3e	2003 - August

## COMMAND AND DATA HANDLING /TELECOMMUNICATIONS

data volume2a2009Januaryfree space loss2b2009Januaryrequired transmit power2c2009Januarydata volume of panoramic photo1a2008Januarydata volume of temperature set1b2008Januarygain of transmitting antenna2a2008Januarywhat is free space loss2b2008Januarycalculate transmitt power2c2008Januarydata volume2a2009Aprilfree space loss2d2009AprilEb/No (given free space loss)2e2009AprilCalculate gain of helix antenna2a2007MarchCalculate free space loss2b2007MarchCalculate free space loss2b2007MarchCalculate required data rate3a2007Februarycalculate required bandwidth3b2007Februarydetermine space losc3c2007Februarysignal round trip/ accuracy in time4a2006Aprildistance from earth surface4c2006Aprilpower density at earth surface4c2005Januaryspace loss right above ground station2c2005Januaryspace loss right above ground station2c2005Januarynumber of bits needed for one pixel5a2005Januarynumber of bits needed for one pixel5a2005Januarynumber of bits per			
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effects of convolutional encoding and reed solomon5c2004 - Januarydata rate a maximum distance1a2003 - Augustestimate total amount of databits received1b2003 - August	number of bits needed for one pixel	5a	2004 - January
data rate a maximum distance1a2003 - Augustestimate total amount of databits received1b2003 - August	number of bits per second delivered by camera		,
estimate total amount of databits received 1b 2003 - August	effects of convolutional encoding and reed solomon		
estimate amount of data without coding 1c 2003 - August	estimate total amount of databits received	1b	2003 - August
	estimate amount of data without coding	1c	2003 - August

### PROPULSION

exhaust velocity (solar photovoltaic)
characteristic exhaust velocity (solar photovoltaic)
propellant mass + photovoltaic system mass ?

4b 2009 - April4c 2009 - April4d 2009 - April

calculate total mass of kick stage	4a	2009 - January
estimate mass moments	4c	2009 - January
List&discuss adv/disadv of liquid hydrogen as prop	6a	2008 - January
thrust level&total operation time for lunar transfer	6b	2008 - January
estimate propellant tank volume	6c	2008 - January
Mass of propellant	1a	2007 - February
Required hydrogen tank volume	1b	2007 - February
collector frontal diameter	1c	2007 - February
hydrogen temperature for true exhaust velocity of	1d	2007 - February
thrust efficiency	1e	2007 - February
size of spherical pressure tank	4e	2006 - January
true exhaust velocity	4f	2006 - January
what is electro thermal rocket propulsion	5a	2005 - January
thrust needed to provide for attitude control	5b	2005 - January
nozzel pressure ratio needed to obtain certain exhaust velocity	5c	2005 - January
effective exhaust velocity in vacuum	5d	2005 - January
power to raise mass flow rate to required temperature	5e	2005 - January
jet power /discuss difference with calculated power in e	5f	2005 - January
propellant mass	2a	2003 - August
thrust duration	2b	2003 - August
optimum exhaust velocity	2c	2003 - August
total power needed for thrusters	2d	2003 - August
discuss effect of too less power on exhaust velocity	2e	2003 - August
discuss if ion engine is able to change exhaust velocity	2f	2003 - August

### DESIGN AND SIZING OF SATELLITE BUS

empty vehicle mass	4a	2009 - April
propellant mass + photovoltaic system mass ?	4d	2009 - April
calculate total mass of kick stage	4a	2009 - January
estimate SC mass with margins	4b	2009 - January
estimate mass moments	4c	2009 - January
determine mass moment of inertia	1a	2008 - March
discuss steps to generate first mass budget	1b	2008 - March
discuss adv/disadv of cylindycal shape	1c	2008 - March
total propellant mass for transfer stage	4a	2008 - January
remaining mass in moon rover vehicle	4b	2008 - January
set up mass budget	4c	2008 - January
Mass of propellant	1a	2007 - February
Required hydrogen tank volume	1b	2007 - February
collector frontal diameter	1c	2007 - February
hydrogen temperature for true exhaust velocity of	1d	2007 - February
thrust efficiency	1e	2007 - February
set up mass budget taking design margin into account	5a	2005 - April
Calculate SE for mass of all subsystems	5b	2005 - April
Mass of water tank	5c	2005 - April

ELECTRICAL POWER SUPLY		
required power from solar aray required solar aray area mass of EPS area of solar array	2b 2c 3a	2008 2008 2008 2008
mass of solar aray/power system/space craft		200

2a 2008 - March 2b 2008 - March 2c 2008 - March 3a 2008 - January 3b 2008 - January

battery capacity		2009 - April
mass of battery (given capacity)	2c	2009 - April
determine duration of daylight and eclipse period	3a	2007 - March
minimum power provided during daylight	3b	2007 - March
required array area	3c	2007 - March
required battery capacity	3d	2007 - March
required battery capacity	3e	2007 - March
solar panel area	2a	2006 - April
mass of solar panel	2b	2006 - April
power to be delivered by battery during eclipse	2c	2006 - April
battery energy storage capacity	2d	2006 - April
volume and mass of battery	2e	2006 - April
solar panel area including intermediate steps	3	2004 - January
battery power during eclipse	4a	2004 - January
depth of discharge	4b	2004 - January
energy storage capacity in Wh	4c	2004 - January
mass and volume of battery	4d	2004 - January

SPACECRAFT SYSTEMS ENGINEERING		
SMART	3a	2009 - January
which delta v is needed	3b	2009 - January
calculate data rate	3c	2009 - January
required storage capacity/required bit rate	3d	2009 - January
Sum of RDT&E cost	5a	2007 - March
Standard error for summed RDT&E cost	5b	2007 - March
Prppellant tank mass	5c	2007 - March
average time to build 5 spacecraft	5d	2007 - March
mean time to failure	4a	2007 - February
reliability of this computer system	4b	2007 - February
calculate percentage margin	4c	2007 - February
average time to build spacecraft	4d	2007 - February
calculate standard error	<u>4e</u>	2007 - February

SPACE MISSION CONCEPT EXPLORATION		
advantages/dsadvantages of different flight trajectories	8a	2008 - January
discuss possibility of direct two way communcicatin	8b	2008 - January
EARTH OBSERVATION		
wavelength of frequency	4a	2007 - March
3db beam width	4b	2007 - March
radius of instantaneous coverage	4c	2007 - March
signal round trip/ accuracy in time	4a	2006 - April
diameter of antenna foot print area	4b	2006 - April
power density at earth surface	4c	2006 - April
power received back by the spacecraft	4d	2006 - April