

Mechanics of Materials

$$\sigma = \frac{F}{A} = \left[\frac{N}{m^2} \right] = [Pa] \quad [MPa] = [Pa] \cdot 10^6 = [N/mm^2]$$

σ = tension [Pa]

F = force [N]

A = surface [m²]

$$\sigma = E \cdot \epsilon$$

$$\epsilon = \frac{\Delta L}{L_0}$$

$\Delta L = \delta$ = change in length [m]

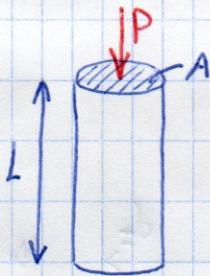
L_0 = original length [m]

ϵ = ~~strain~~ Strain ratio [-]

$$\Rightarrow \frac{F}{A} = E \cdot \epsilon \Rightarrow \frac{F}{A} = E \cdot \frac{\Delta L}{L_0} \Rightarrow \Delta L = \frac{F \cdot L_0}{A E}$$

Watch out with superposition!

Example:



$P = 1 \text{ kN}$

$A = 2 \text{ m}^2$

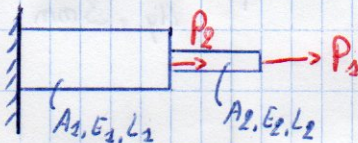
$L = 1 \text{ m}$

$E = 10 \text{ MPa}$

$$\sigma = \frac{P}{A} = \frac{1000}{2} = 500 \text{ Pa}$$

$$\Delta L = \frac{P \cdot L}{A E} = \frac{1000 \cdot 1}{2 \cdot 10 \cdot 10^6} = 5 \cdot 10^{-5} = 0,00005 \text{ m}$$

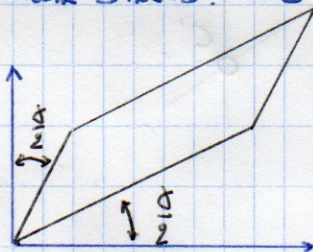
$$\delta = \Delta L = 0,05 \text{ mm}$$



! $A_1 \neq A_2, E_1 \neq E_2, L_1 \neq L_2$!

Shear stress: $\tau = G \cdot \gamma$

$$G = \frac{E}{2(1+\nu)}$$



τ = shear stress [Pa]

G =

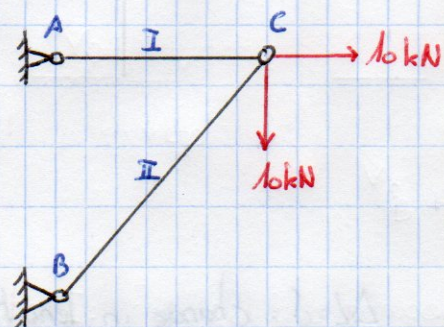
γ =

V = shear force [N]

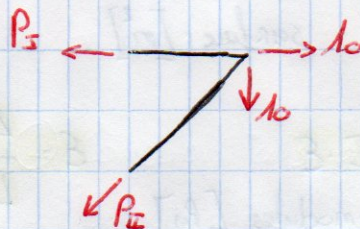
static undetermined constraints:

Williot diagram:

Example:



First determine the internal forces:

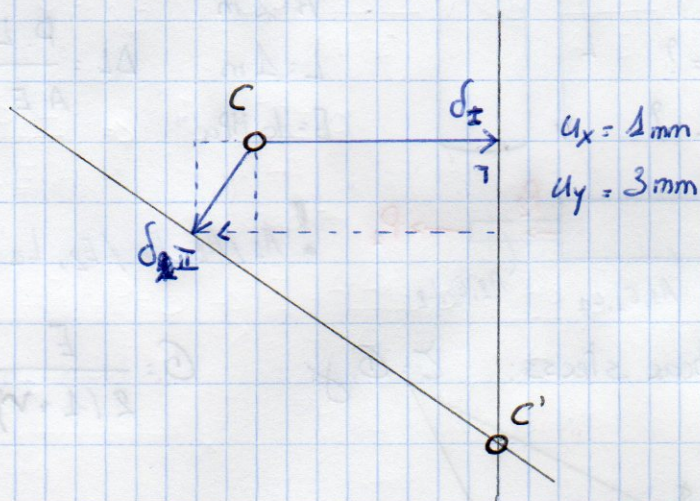


$$P_I = 20 \text{ kN}$$
$$P_{II} = -10\sqrt{2} \text{ kN}$$

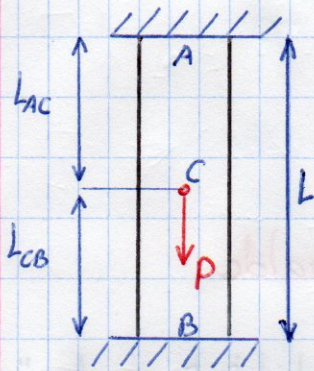
Secondly compute displacements:

$$\delta_I = \frac{P_I L_I}{A_I E_I} = 1 \text{ mm}$$

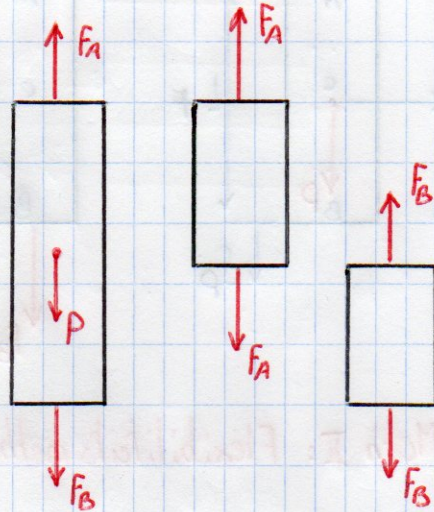
$$\delta_{II} = \frac{P_{II} L_{II}}{A_{II} E_{II}} = -\sqrt{2} \text{ mm}$$



Statically indeterminate structures



A, E are given



$$\sum F = 0 \quad -F_B + F_A - P = 0 \quad \textcircled{I}$$

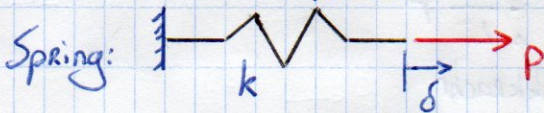
$$\delta_{AB} = \delta_A - \delta_B = 0 \quad \delta = \frac{P \cdot L}{A \cdot E}$$

$$= \frac{F_A \cdot L_{AC}}{E \cdot A} + \frac{F_B \cdot L_{CB}}{E \cdot A} = 0 \quad \textcircled{II}$$

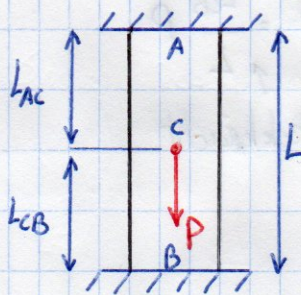
$$\text{Uit } \textcircled{I} \text{ en } \textcircled{II} : F_A = \frac{L_{CB}}{L} P, \quad F_B = -\frac{L_{AC}}{L} P$$

trekkkracht

dukkkracht



$$P = k \cdot \delta \rightarrow \delta = \frac{P}{k}; \quad k = \frac{AE}{L}; \quad \frac{1}{k} = \frac{L}{AE}$$



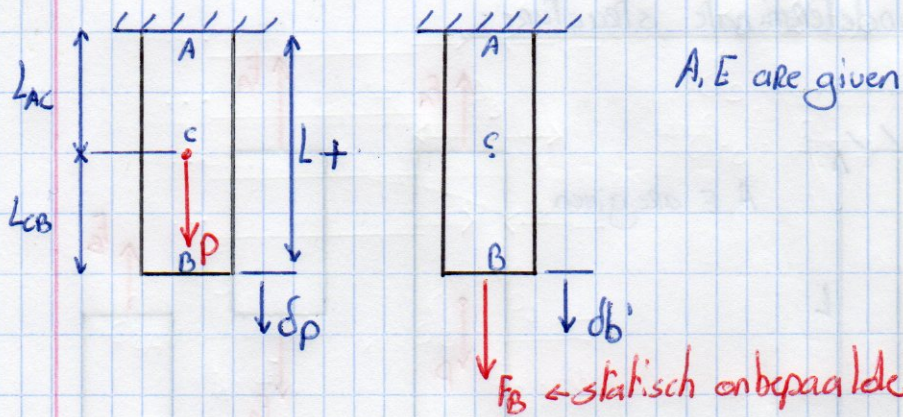
A, E are given **Meth I:** ① Evenwicht:

- Maak FBD
- Stel evenwichtsgl op (-F_B + F_A - P = 0)

② Compatibiliteit:

- Tekening met de verplaatsingen
- Compatibiliteitsvoorwaarden ($\delta_{AB} = \delta_A - \delta_B = 0$)
- Gebruik de krachtverplaatsingrelaties

③ Bereken de reactiekrachten uit ① en ②.



Meth II: Flexibiliteitsmethode: ① Compabiliteits en de "statisch onbepaalde"

- Superpositie van 2 systemen

De bekende verplaatsing δ_b bij de statisch onbepaalde ondersteuningskracht = verplaatsing δ_b t.g.v. alleen externe belasting + verplaatsing δ_b' t.g.v. de statisch onbepaalde kracht.

$$0 = \delta_b = \delta_p + \delta_b', \quad \delta = \frac{P \cdot L}{A \cdot E}, \quad \delta_p = \frac{P \cdot L_{AC}}{A \cdot E}, \quad \delta_b' = \frac{F_b \cdot L}{A \cdot E}$$

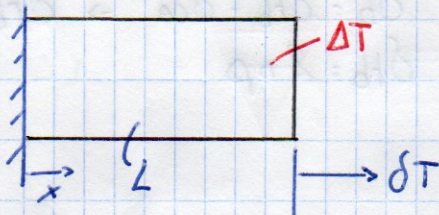
$$0 = \frac{P \cdot L_{AC}}{A \cdot E} + \frac{F_b \cdot L}{A \cdot E} \Rightarrow F_b = \frac{L_{AC}}{L} P$$

↑
drukkracht

$$\text{Evenwicht: } -F_b + F_A - P = 0 \Rightarrow F_A = \frac{L - L_{AC}}{L} P = \frac{L_{CB}}{L} P$$

↑
trekkracht

Thermische belasting:

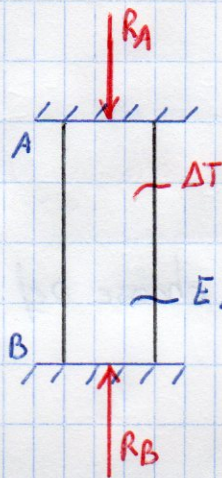


ΔT : Temperaturverandering
 α : thermische expansie coefficient
 ϵ_T : thermische rek
 $\epsilon_T = \alpha \cdot \Delta T$

Verlenging staaf: $\delta T = \alpha \cdot \Delta T \cdot L$, als $\alpha = \alpha(x)$

~~Evenwicht: $R_A - R_B = 0$~~

dan $\delta T = \int_0^L \alpha \Delta T dx$



Evenwicht: $R_A - R_B = 0$

Compatibiliteit:

- verlenging $\delta_{AB} = \delta T + \delta F = 0$
- verplaatsing door $\Delta T = \alpha \cdot \Delta T \cdot L$
- verlenging t.g.v. interne krachten $\delta F = \frac{-R_B \cdot L}{A \cdot E}$

$$\delta_{AB} = \alpha \cdot \Delta T \cdot L - \frac{R_B \cdot L}{A \cdot E} = 0 \Rightarrow$$

$$R_B = \frac{\alpha \cdot \Delta T \cdot L \cdot (A \cdot E)}{L} = \alpha \cdot \Delta T \cdot (A \cdot E)$$

Turnbuckle = tb

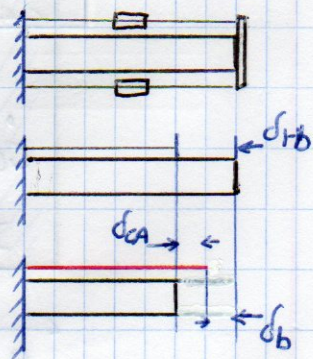
$$\delta_{tb} = 2 \cdot n \cdot p$$

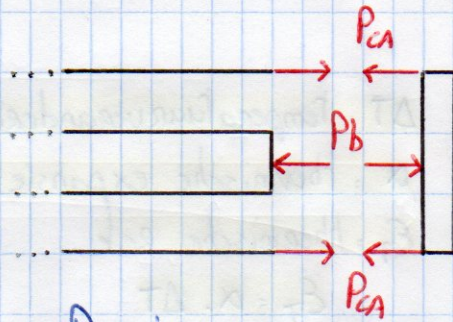
n = aantal omwentelingen

p = speed

δ_{ca} = verlenging kabel

δ_b = verkorting staaf





$$P_b = 2 P_{ca}$$

$$\delta_b = \delta_{tb} - \delta_{ca} \rightarrow \delta_{tb} = \delta_b + \delta_{ca}$$

$$\delta_{tb} = 2np$$

$$\delta_{ca} = \frac{P_{ca} \cdot L}{A_{ca} \cdot E_{ca}}, \quad \delta_b = \frac{P_b \cdot L}{E_b \cdot A_b}$$

$$2np = \frac{P_b \cdot L}{E_b \cdot A_b} + \frac{P_{ca} \cdot L}{A_{ca} \cdot E_{ca}} \quad P_{ca} = \dots$$

$$P_b = \dots$$

Zwaartepunt & Traagheidsmoment (I)

$$\hookrightarrow \bar{x} = \frac{\sum \tilde{x} A}{\sum A}, \quad \bar{y} = \frac{\sum \tilde{y} A}{\sum A} \rightarrow \text{choose ref. axis}$$

$$I \rightarrow I_x = \int_A y^2 dA, \quad I_y = \int_A x^2 dA$$

$$I_x = \bar{I}_x + A d_y^2, \quad I_y = \bar{I}_y + A d_x^2 \leftarrow \text{Als fig. niet symmetrisch is!}$$

Example:

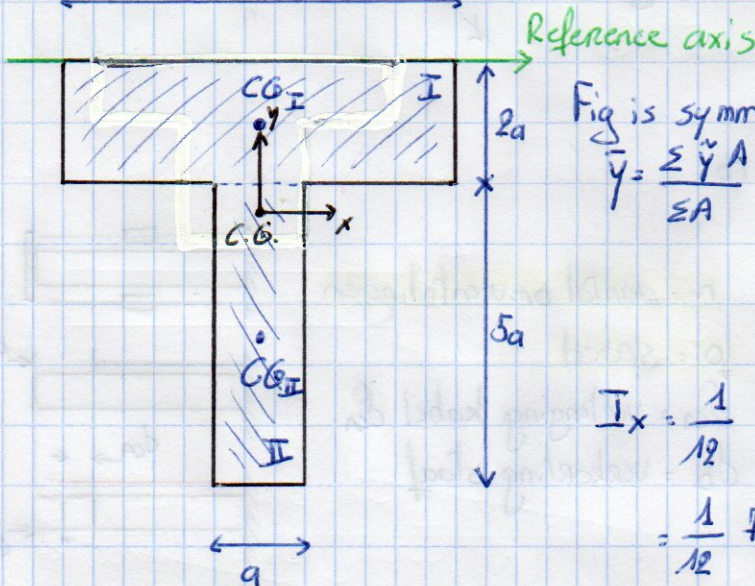


Fig is symm.:

$$\bar{y} = \frac{\sum \tilde{y} A}{\sum A} = \frac{\frac{1}{2} \cdot 2a \cdot 2a \cdot 7a + (\frac{1}{2} (5a+2a) 5a \cdot a)}{2a \cdot 7a + 5a \cdot a}$$

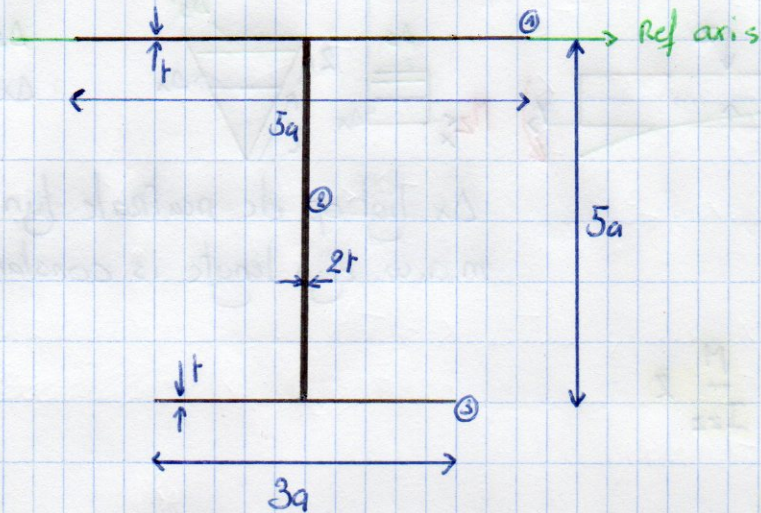
$$= \frac{73}{38} a$$

$$I_x = \frac{1}{12} b h^3 + A d_y^2$$

$$= \frac{1}{12} 7a (2a)^3 + 7a \cdot 2a \left(\frac{73}{38} a - \frac{1}{2} \cdot 2a \right)^2 + \frac{1}{12} a (5a)^3 + a \cdot 5a \left(\frac{73}{38} a - \frac{1}{2} a \right)^2$$

Inertia moment on thin walled structure (t):

Example



Zwaartepunt:
$$\bar{y} = \frac{\sum \bar{y} A}{\sum A} = \frac{\frac{1}{2}t \cdot 3a \cdot t + (t + \frac{1}{2}5a) \cdot 5a \cdot 2t + (\frac{3}{2}t + 5a) \cdot 3a \cdot t}{3a \cdot t + 5a \cdot 2t + 3a \cdot t}$$

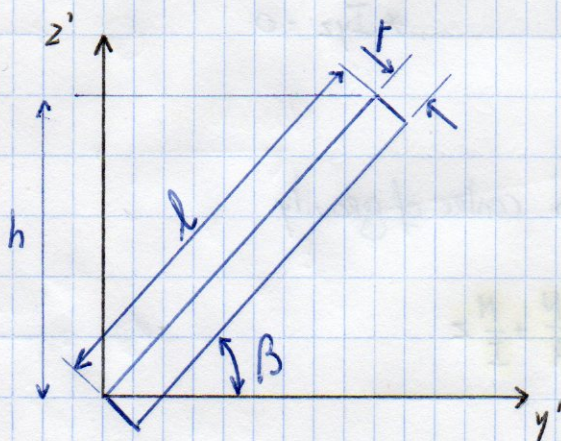
$$= \frac{\frac{31}{2} a^2 t + 40 a^2 t}{18 a t} = \frac{40 a^2 t}{18 a t} = \frac{20}{9} a = \bar{y}$$

Inertia:
$$I_x = \frac{1}{12} b h^3 + A d y^2$$

$$= \frac{1}{12} 3a \cdot t^3 + 3a \cdot t \left(\frac{20}{9} a\right)^2 + \frac{1}{12} 2t \cdot 5a^3 + 2t \cdot 5a \cdot \left(\frac{1}{2} 5a - \frac{20}{9} a\right)^2$$

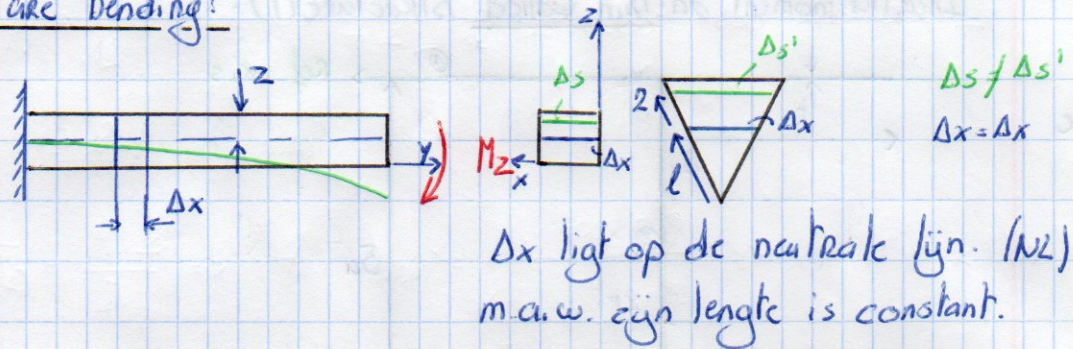
$$+ \frac{1}{12} 3a \cdot t^3 + 3a \cdot t \left(5a - \frac{20}{9} a\right)^2$$

$$= \dots = \frac{625}{9} a^3 t$$



$$I_y = \frac{1}{12} t l^3 \sin^2 \beta$$

Pure bending:



$$\sigma = \frac{E}{l} z = \frac{M}{I_{zz}} z$$

σ = tension

E = E-modulus

l = length from bottom to NL

z = distance from NL to the point

M = moment

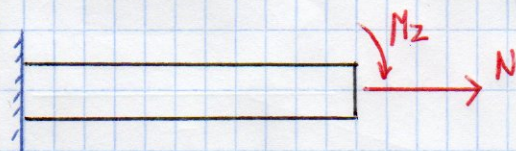
I_{zz} = inertia

} look at drawing above

Solving strategie pure bending:

- ① FBD
- ② V- en M-lines
- ③ zwpt and inertia
- ④ location / beam-axis, cross-cut
- ⑤ $\sigma_{max} = \frac{M}{I} z_{max}$

Combination normal force & bending



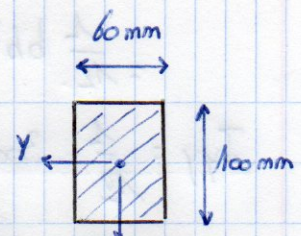
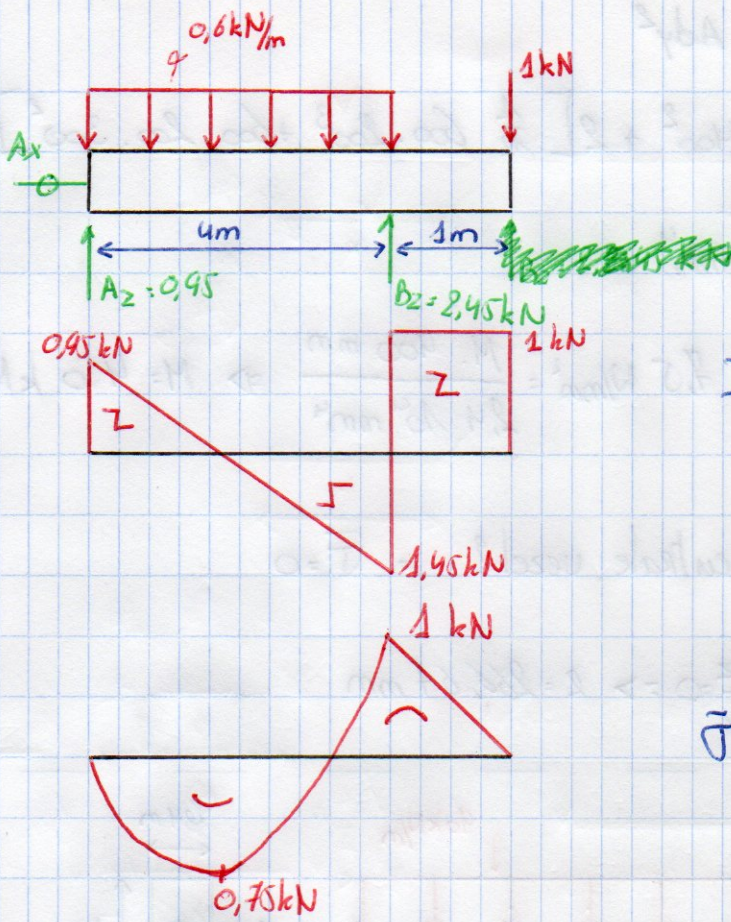
- * Bending in x-z-plane
- * $I_{yz} = 0$

→ y-axis thru NC

→ normal force grasps in centre of gravity

$$\sigma = \frac{N}{A}; \sigma = \frac{M}{I} z \rightarrow \sigma = \frac{N}{A} + \frac{M}{I} z$$

Example:



$$I_{yy} = \frac{1}{12} b h^3$$

$$= \frac{1}{12} \cdot 0.06 \cdot 0.1^3$$

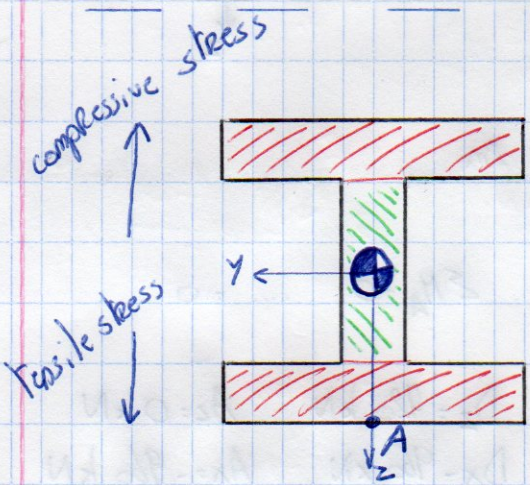
$$= 5 \cdot 10^{-6} \text{ mm}^4$$

$$\bar{\sigma} = \frac{\bar{M} \cdot \bar{z}}{I_{yy}}$$

$$= \frac{1 \cdot 10^3 \cdot 0.05}{5 \cdot 10^{-6}}$$

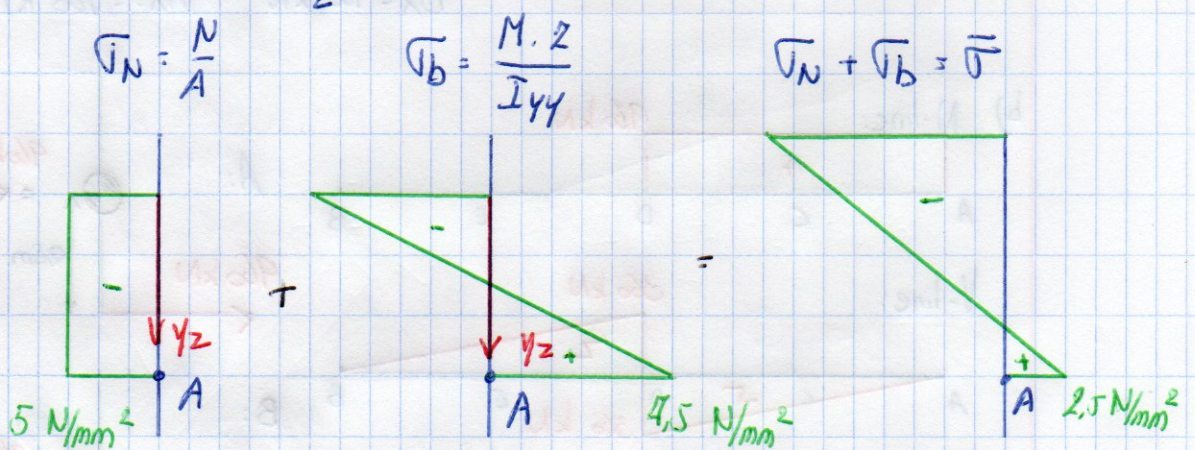
$$= 10 \text{ MPa}$$

Example:



$A = 0.36 \text{ m}^2$ $F = 1800 \text{ kN}$
 ↳ Some where in cross-section

$\bar{\sigma} = 2.5 \text{ N/mm}^2$ in tensile stress



$$\frac{1800000}{0.36 \cdot 10^6} = \sigma_N$$

$$5 + x = 2.5$$

$$x = -2.5$$

$$= \frac{1}{12} b b^3 + A d y^2$$

$$I_{yy} = \frac{1}{12} \cdot 300 \cdot 400^3 + 2 \left[\frac{1}{12} 600 \cdot 200^3 + 600 \cdot 200 \cdot 300^2 \right]$$

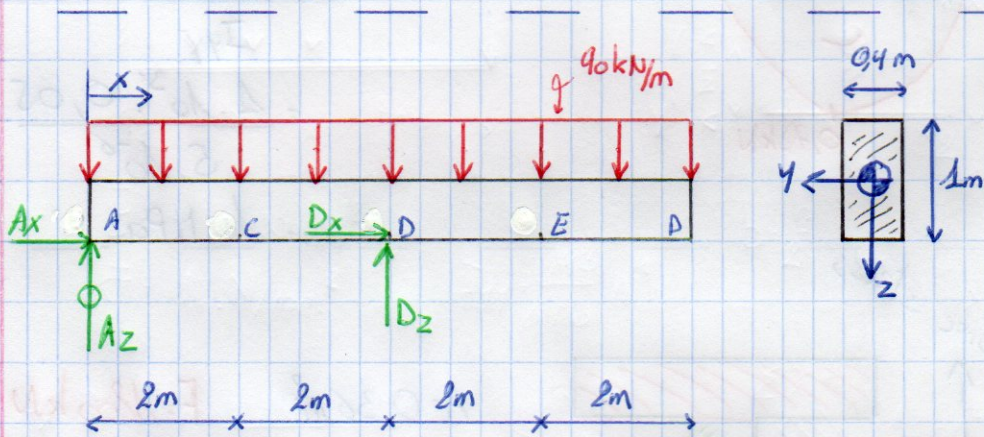
$$= 2,4 \cdot 10^4 \text{ mm}^4$$

$$\sigma_b = \frac{M \cdot z}{I_{yy}} \Rightarrow 7,5 \text{ N/mm}^2 = \frac{M \cdot 400 \text{ mm}}{2,4 \cdot 10^4 \text{ mm}^4} \Rightarrow M = 450 \text{ kNm}$$

Waar ligt de neutrale vezel? $= \bar{y} = 0$

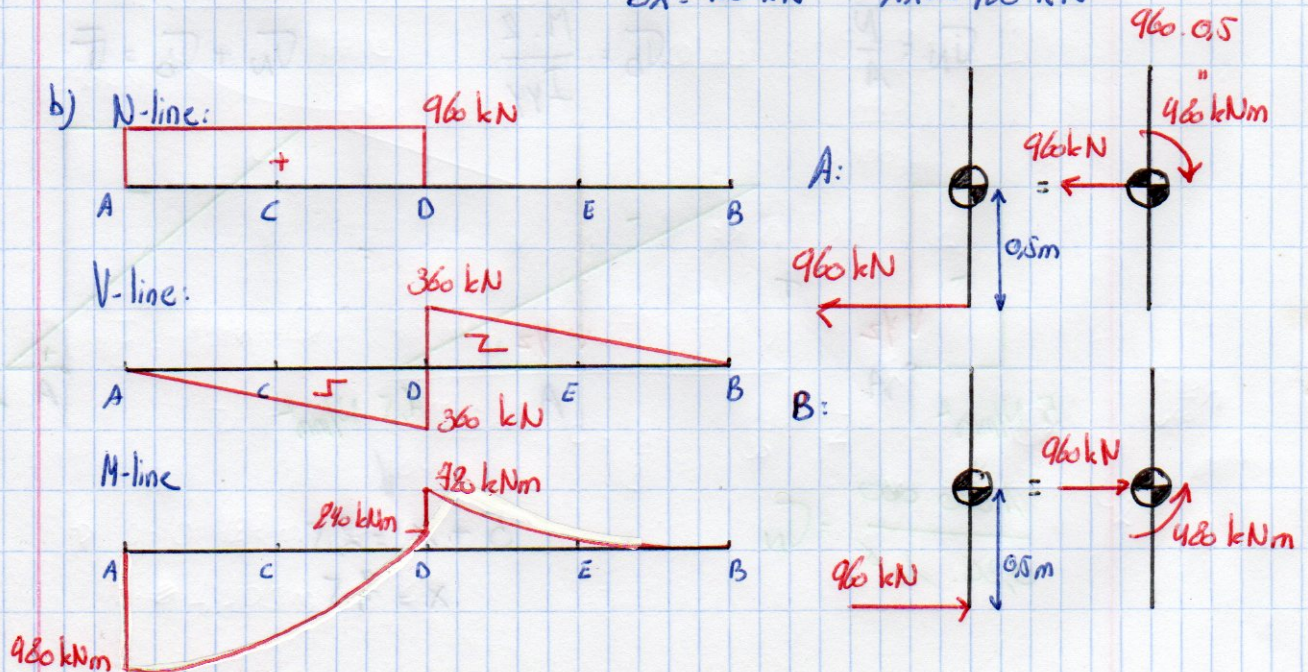
$$\bar{y} = \frac{N}{A} + \frac{M \cdot z}{I_{yy}} = 0 \Rightarrow z = 266,67 \text{ mm}$$

Example

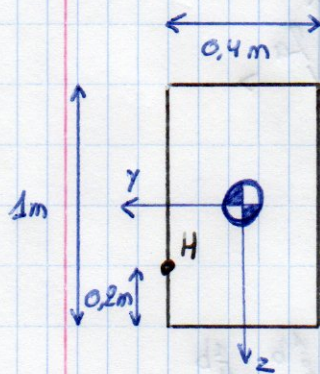


a) $\sum F_y = 0 \quad A_z + D_z = 90 \cdot 8$ $\sum M_A = 0 \quad \dots = 0$
 $\sum F_x = 0 \quad A_x + D_x = 0$

$D_z = 720 \text{ kN} \quad A_z = 0 \text{ kN}$
 $D_x = 960 \text{ kN} \quad A_x = -960 \text{ kN}$



c) Determine the normal and shear stress acting at point H for the cross-sections at C and E



$$A = 0,4 \text{ m}^2 = \frac{2}{5} \text{ m}^2$$

$$I_{yy} = \frac{1}{12} 0,4 \cdot 1^3 = \frac{1}{30} \text{ m}^4$$

$$z_H = 0,5 - 0,2 = 0,3 \text{ m} = \frac{3}{10} \text{ m}$$

$$\bar{\sigma} = \frac{N}{A} + \frac{M \cdot z}{I_{yy}}$$

$$\bar{\sigma}_C(H) = \frac{960 \cdot 10^3}{\frac{2}{5}} + \frac{180 \cdot 10^3 \cdot \frac{3}{10}}{\frac{1}{30}} = 5,1 \text{ MPa}$$

$$\bar{\sigma}_E(H) = \frac{0}{\frac{2}{5}} - \frac{180 \cdot 10^3 \cdot \frac{3}{10}}{\frac{1}{30}} = -1,62 \text{ MPa} \quad \text{Compressive force!}$$

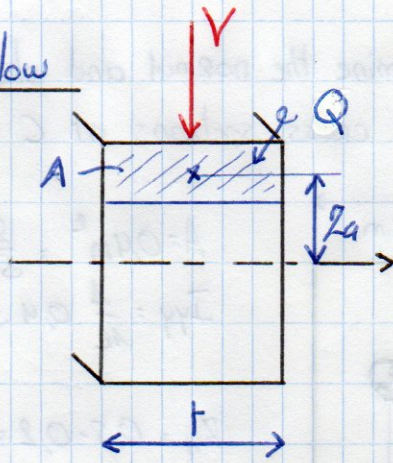
$$\tau_C(H) =$$

$$\tau_E(H) =$$

Shear stress & shear flow

$$\tau = \frac{V \cdot Q^a}{I \cdot t^a}$$

$$Q^a = A \cdot z_a$$



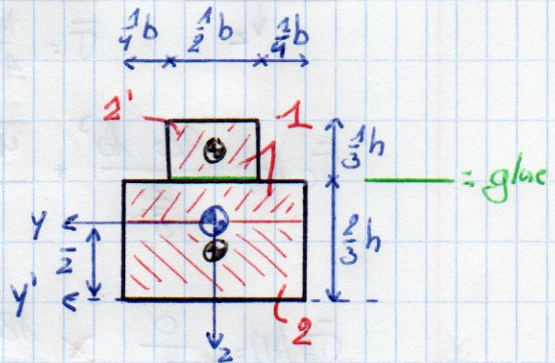
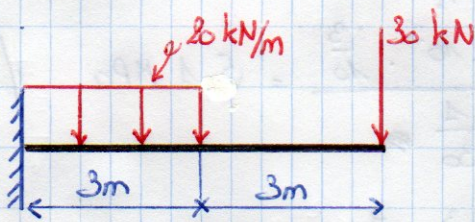
τ = shear stress

V = shear force

I = inertia moment

t = thickness

Example:



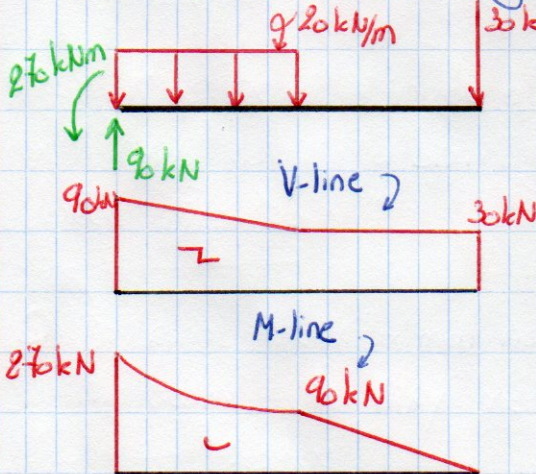
$$\bar{\tau} = 0,33 \text{ N/mm}^2$$

$$\bar{\sigma} = 10 \text{ N/mm}^2$$

$$J_{yy} = \frac{193}{6480} h^4$$

$$b = \frac{h}{2}$$

a) Determine the minimum height, h



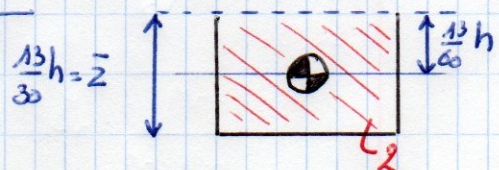
$$\bar{z} = \frac{\sum x \cdot A}{A} = \frac{\frac{13}{36} b h^2}{\frac{5}{6} b h} = \frac{13}{30} h$$

$$\bar{\sigma} = \frac{M_{\max} \cdot \bar{z}}{J_{yy}}$$

$$0,33 = \frac{270 \cdot 1000 \cdot \frac{17}{30} h}{\frac{193}{6480} h^4}$$

$$\Rightarrow h = 0,7 \text{ m} \quad 1$$

$$\bar{\tau} = \frac{V_{\max} \cdot Q}{I \cdot t} = \frac{90 \cdot 1000 \cdot (b \cdot \frac{13}{30} h \cdot \frac{13}{60} h)}{\frac{193}{6480} h^4 \cdot b}$$



$$\Rightarrow h = 0,927 \text{ m} \leftarrow 2$$

$$h = 1,105 \text{ m} \leftarrow 1'$$

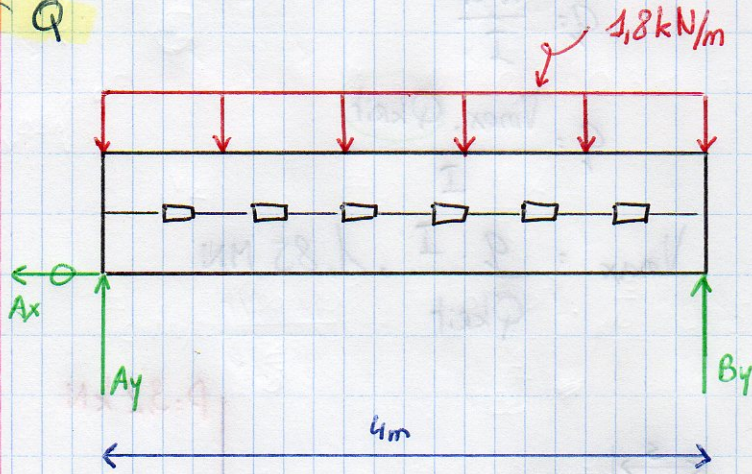
$$q = \frac{V \cdot Q}{I}$$

q = shear flow [N/m]

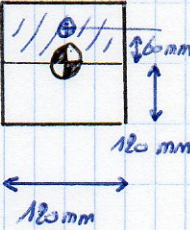
V = shear force

Q

Example:



□ = dowel
 $\bar{\sigma} = 5 \text{ kN}$



Q: minimum nr of "dowels"

$$\sum F_y = 0$$

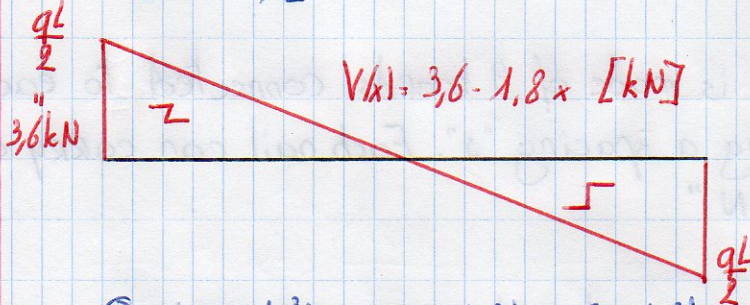
$$\sum M_A = 0$$

$$B_y = 3,6 \text{ kN}$$

$$A_y + B_y = 1,8 \cdot 4$$

$$A_y = 3,6 \text{ kN}$$

$$I = \frac{1}{12} b h^3 = \frac{1}{12} \cdot 120 \cdot 240^3 = 138,24 \cdot 10^6 \text{ mm}^4$$



$$Q = (120 \cdot 10^{-3}) \cdot (120 \cdot 10^{-3}) \cdot (60 \cdot 10^{-3}) = 8,64 \cdot 10^{-3} \text{ [m}^3]$$

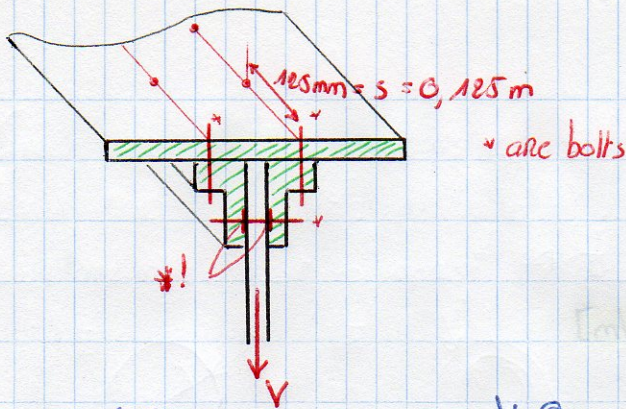
$$q(x) = \frac{V(x) \cdot Q}{I} = 22,5 - 11,25x \text{ [kN/m]}$$

$$F(x) = \int_0^x q(x) dx = 22,5x - 5,625x^2 \text{ [kN]}$$

\rightarrow nr of dowels for 2m: $\frac{22,5 \text{ kN}}{5 \text{ kN}} = 4,5 \rightarrow$ 5 dowels

for 4m: $5 \cdot 2 =$ 10 dowels

Example:



given: $\rightarrow I = \frac{30 \cdot 434}{3} b^4$

$$q = \frac{V \cdot Q}{I}$$

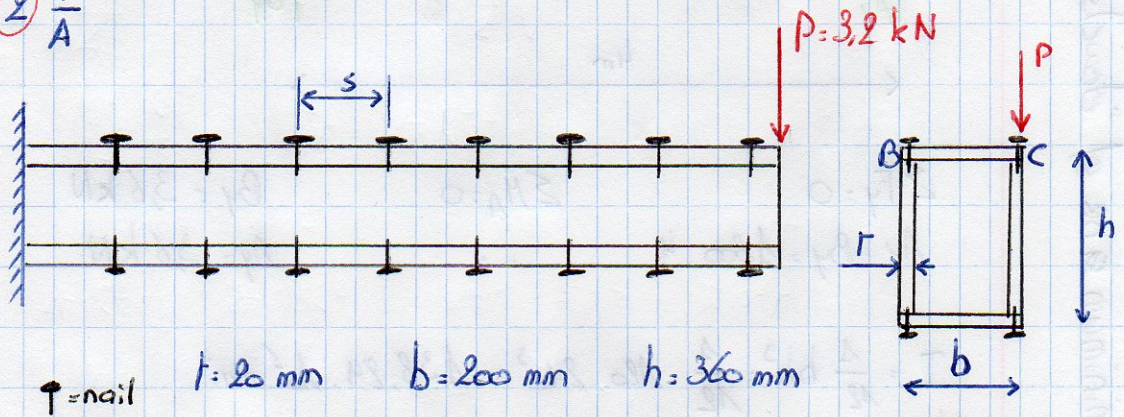
$$q = \frac{V_{max} \cdot Q_{krit}}{I}$$

$Q_{krit} = 4,75b \cdot t \cdot 3$

$$V_{max} = \frac{q \cdot I}{Q_{krit}} = 1,85 \text{ MN}$$

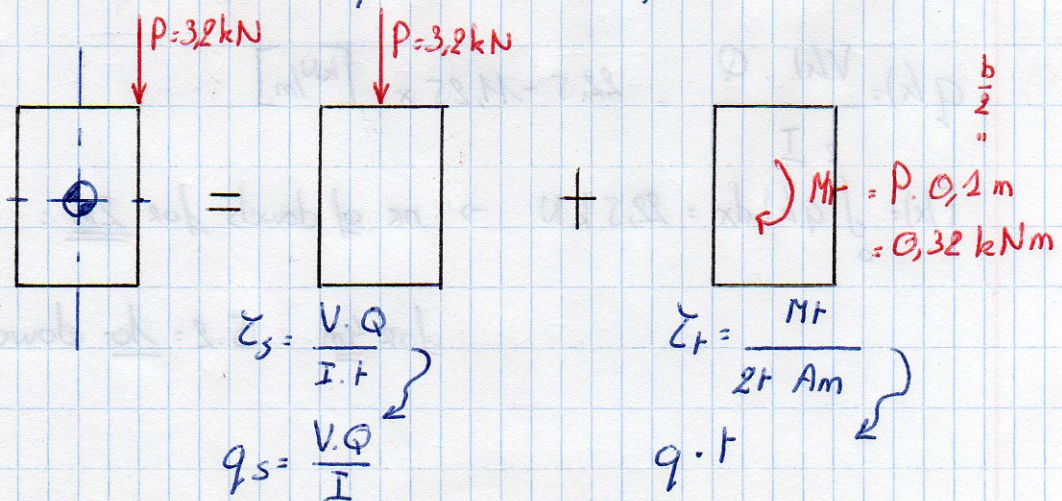
$q = 2 \cdot \frac{F}{A}$

Example:



A built up beam is made of 4 boards connected to each other with nails having a spacing "s". Each nail can carry a shear force "F_n = 250 N"

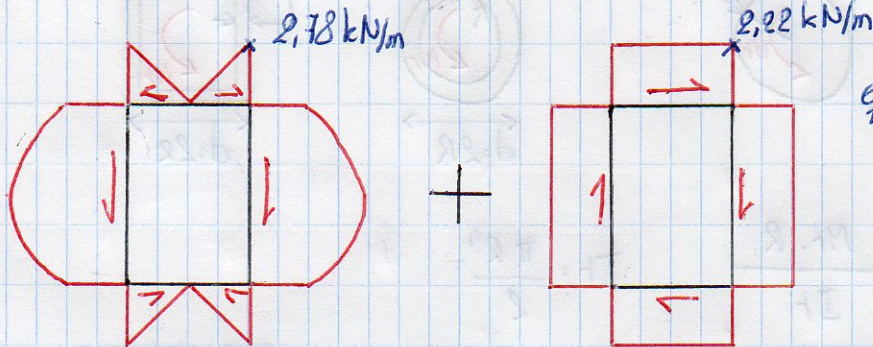
Q: Determine the maximal nail spacing "s_B" and "s_C", such that the beam can carry the vertical force P = 3,2 kN



$$A_m = h \cdot b = 360 \cdot 200 = 72 \cdot 10^3 \text{ mm}^2$$

$$Q = t \cdot b \cdot \frac{h}{2} = t b h = \frac{20 \cdot 200 \cdot 360}{2} = 720 \cdot 10^3 \text{ mm}^3$$

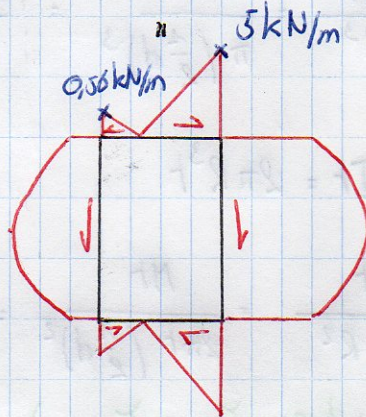
$$I = 2 \left(\frac{t h^3}{12} \right) + 2 \left(b \cdot t \cdot \left(\frac{h}{2} \right)^2 \right) = 4,1472 \cdot 10^8 \text{ mm}^4$$



$$q \cdot t = \frac{320}{2 \cdot 72 \cdot 10^3} = 0,00222 \text{ N/mm}$$

$$q_s = \frac{V \cdot Q}{I} = \frac{P \cdot Q}{I}$$

$$= \frac{320 \cdot 720 \cdot 10^3}{4,1472 \cdot 10^8 \cdot 2} = 2,78 \text{ kN/m}$$



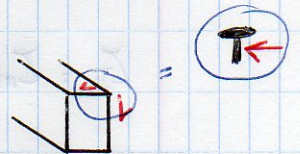
$$q \cdot t = \frac{M_t}{2 A_m}$$

$$= \frac{320 \text{ [kNmm]}}{2 \cdot 72 \cdot 10^3 \text{ [mm}^2\text{]}}$$

$$= 0,00222 \text{ kN/mm} = 2,22 \text{ kN/m}$$

$$F_N = q_B \cdot S_B \Rightarrow S_B = \frac{F_N}{q_B} = \frac{250}{5000} = 0,05 \text{ m}$$

$$F_N = q_C \cdot S_C \Rightarrow S_C = \frac{F_N}{q_C} = \frac{250}{560} = 0,45 \text{ m}$$



$$\tau = \frac{M_t \cdot R}{I_t} = \frac{M_t}{A_m \cdot r}$$

Rigid

τ = shear stress

M_t = torsion moment

r = radius (inner)

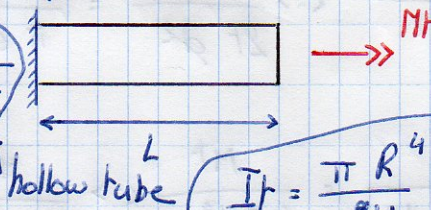
I_t = moment of inertia to torsion

R = radius (outer)

L = length of bar

G = torsion stiffness

$\Delta \varphi_x$ =



$$I_t = \frac{\pi R^4}{24} \text{ for rigid bar}$$

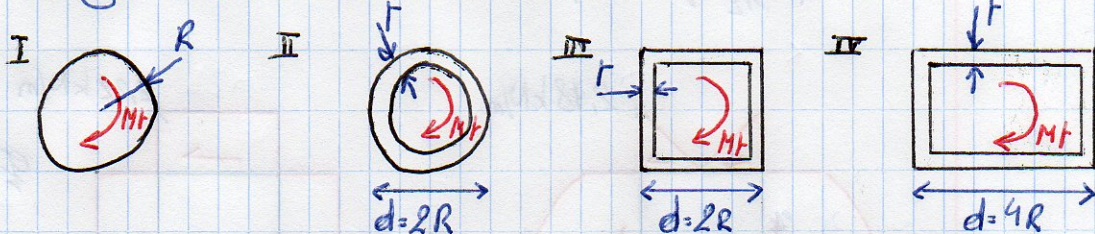
$$A_m = \pi R^2$$

$$I_t = 2\pi R^3 t \text{ hollow bar}$$

$$\Delta \varphi_x = \frac{M_t \cdot L}{G \cdot I_t}$$

Example:

Given are 4 front views of diff. bars. In the center of each bar is a M_T . Which bar is the best to resist M_T ? Rank arrange: best \rightarrow worst ~~1-2-3-4~~



ζ_{\max}

$$\zeta_{II} = \frac{M_T \cdot R}{I_T}$$

$$I_T = \frac{\pi R^4}{2}$$

$$= \frac{M_T \cdot R}{\frac{\pi R^4}{2}} = \frac{2 M_T}{\pi R^3} = \frac{2 M_T}{\pi \left(\frac{1}{2}d\right)^3} = \frac{16 M_T}{\pi d^3}$$

ζ_{\max}

$$\zeta_{IV} = \frac{M_T \cdot R}{I_T}$$

$$I_T = 2\pi R^3 t$$

$$= \frac{M_T \cdot R}{2\pi R^3 t} = \frac{M_T}{2\pi t R^2} = \frac{M_T}{2\pi t \left(\frac{1}{2}d\right)^2} = \frac{2 M_T}{\pi d^2 t}$$

(Now find t , where $\zeta_{IV} = \zeta_{\max}$)

$$\zeta_{IV} = \zeta_{\max} \Leftrightarrow \frac{2 M_T}{\pi d^2 t} = \frac{16 M_T}{\pi d^3} \Leftrightarrow t = \frac{d}{8} = 0,125 d$$

$$\zeta_{III} = \frac{M_T}{2t d^2} \Leftrightarrow \frac{M_T}{2t d^2} = \frac{16 M_T}{\pi d^3} \Leftrightarrow t = \frac{\pi d}{32} = 0,0917 d$$

$$\zeta_{II} = \frac{M_T}{2t d^2} = \frac{M_T}{4t d^2} \Leftrightarrow \frac{M_T}{4t d^2} = \frac{16 M_T}{\pi d^3} \Leftrightarrow t = \frac{\pi d}{64} = 0,05 d$$

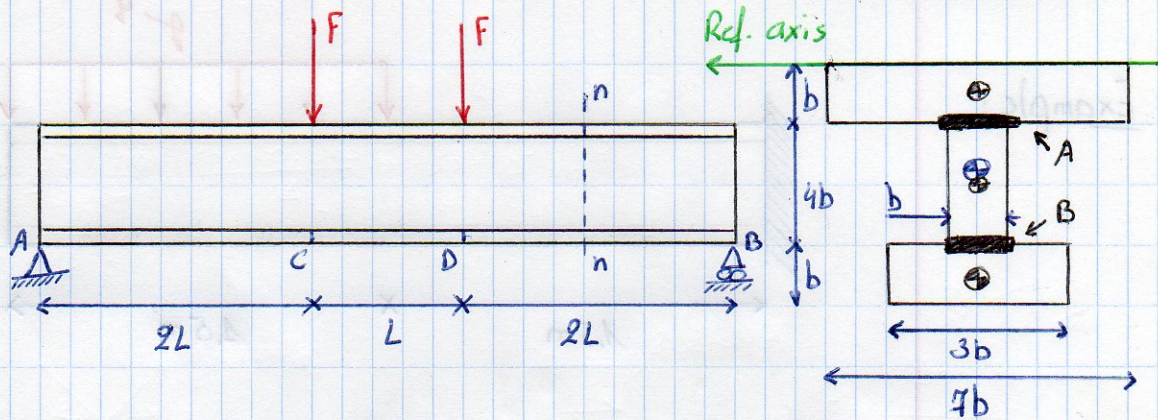
$$A_1 = \pi R^2 = \frac{\pi d^2}{4} = 0,25 \pi d^2$$

$$A_4 = 6 d t = \frac{3}{32} \pi d^2 = 0,09375 \pi d^2$$

$$\frac{A_4}{A_1} = 0,375$$

~~I - II - III - IV~~ IV - III - II - I

Example:



Q: Determine the shear stress in joint A at cross-section n-n.

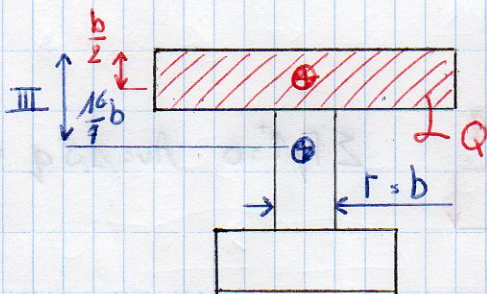
$$I. \bar{y} = \frac{\sum yA}{\sum A} = \frac{\frac{b}{2} \cdot 7b \cdot b + 3b \cdot 4b \cdot b + \frac{11}{2}b \cdot 3b \cdot b}{7b \cdot b + 4b \cdot b + 3b \cdot b} = \frac{32b^3}{14b^2} = \frac{16}{7}b$$

$$II. I = \sum \bar{I}_x + A d^2$$

$$= \frac{7b \cdot b^3}{12} + 7b^2 \left(\frac{16}{7}b - \frac{b}{2} \right)^2 + \frac{b \cdot (4b)^3}{12} + 4b^2 \left(3b - \frac{16}{7}b \right)^2$$

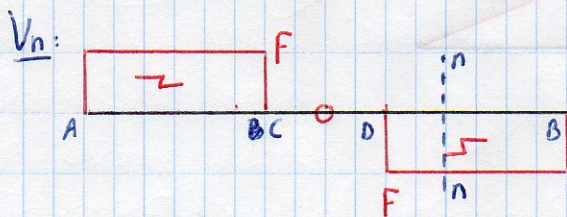
$$+ \frac{3b \cdot b^3}{12} + 3b^2 \left(\frac{11}{2}b - \frac{16}{7}b \right)^2$$

$$= \frac{1292}{21} b^4$$



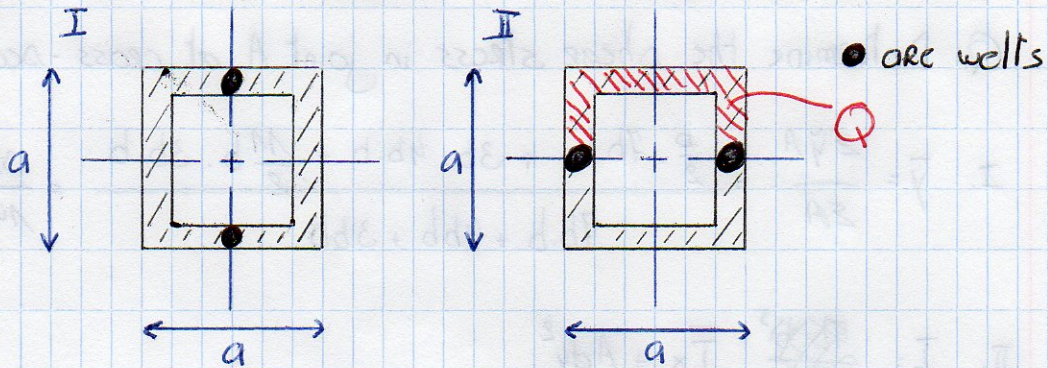
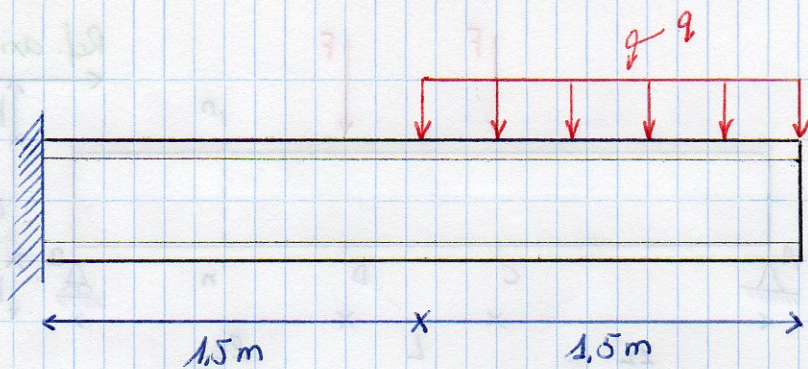
$$Q = y' A' = \left(\frac{16}{7}b - \frac{b}{2} \right) \cdot 7b^2 = \frac{25}{2} b^3$$

$$\tau = \frac{V_n \cdot Q}{I \cdot r} = \frac{F \cdot \frac{25}{2} b^3}{\frac{1292}{21} b^4 \cdot b} = \frac{525}{2584} \frac{F}{b^2}$$



$$V_n = F$$

Example:



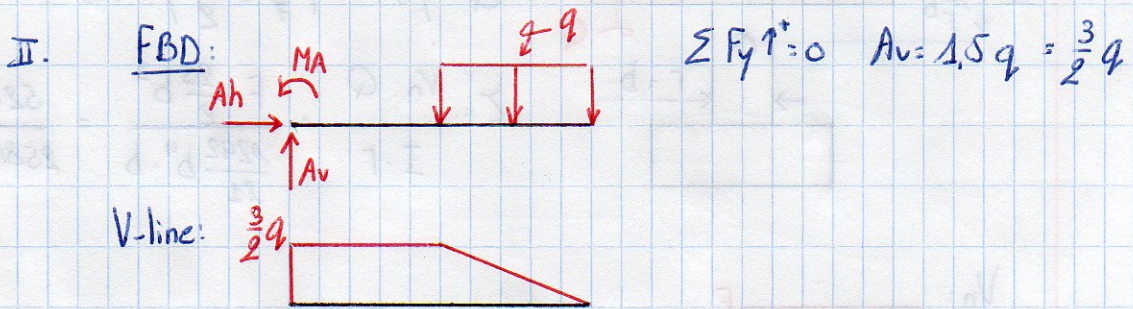
Q: Calculate the max. avg. shear flow along 1 welding joint for the cross-section in position I and II

I.
$$I = \bar{I}_x + A d y^2$$

$$= 2 \left[\frac{a t^3}{12} + a t \cdot \left(\frac{a}{2}\right)^2 \right] + 2 \left[\frac{t a^3}{12} + a t \cdot (0) \right]$$

thin walled

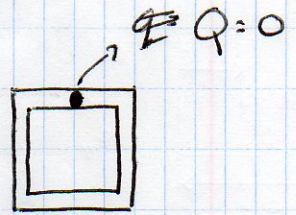
$$= \frac{2}{3} t a^3$$



III. For position I:

$$\Sigma = \frac{V \cdot Q}{I \cdot r} \Leftrightarrow \Sigma \cdot r = \frac{V \cdot Q}{I} = q$$

$$Q=0 \rightarrow \cancel{\Sigma q} = 0 \quad q=0$$



For position II:

Check figure \rightarrow

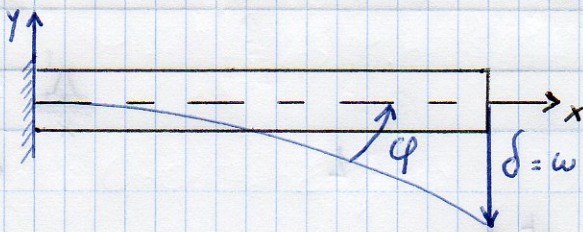
$$Q_{II} = y \cdot A' = 2 \left[r \cdot \frac{a}{2} \right] \cdot \frac{a}{4} + r \cdot a \cdot \frac{a}{2} = \frac{3}{4} r a^2$$

$$q = \frac{V \cdot Q}{I} = \frac{\frac{3}{2} q \cdot \frac{3}{4} r a^2}{\frac{2}{3} r a^3} = \frac{27}{16} \frac{q}{a} \quad \text{Is voor alle twee de "wells" !}$$

$$\text{Dus: } q_{II} = \frac{q}{2} = \frac{27}{32} \frac{q}{a}$$

The deflection curve

diff. equations

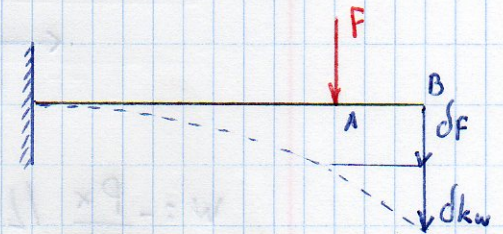
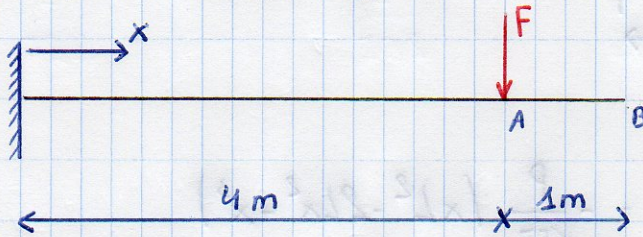


$$\begin{aligned}
 EI w''' &= q \\
 EI w''' &= -V \\
 EI w'' &= -M \\
 EI w' &= -EI \phi \\
 EI w &= EI \delta
 \end{aligned}$$

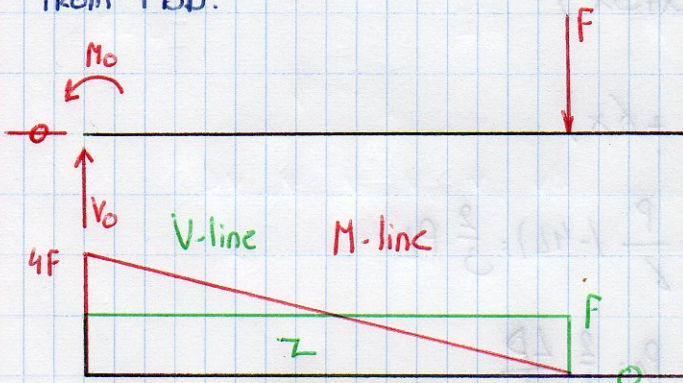
Use RVWN to solve the problem.

Example:

Let op het kwispel effect.



From FBD:



$$\begin{aligned}
 M_0 &= 4F \\
 V_0 &= F
 \end{aligned}$$

$$\begin{aligned}
 M(x) &= 4F - F \cdot x = EI w'' \\
 EI w' &= 4Fx - \frac{Fx^2}{2} + C_1
 \end{aligned}$$

RVWON: $x=0 \quad w=0 \Rightarrow C_2=0$
 $x=0 \quad \phi=0 \Rightarrow C_1=0$

$$EI w = \frac{4Fx^2}{2} - \frac{Fx^3}{6} + C_2 x + C_2$$

$$EI w = \frac{4Fx^2}{2} - \frac{Fx^3}{6}$$

$$\delta_F \Rightarrow EI w = \frac{4F \cdot 4^2}{2} - \frac{F \cdot 4^3}{6}$$

$$w = \frac{64}{3} \cdot \frac{F}{EI} = \delta_F$$

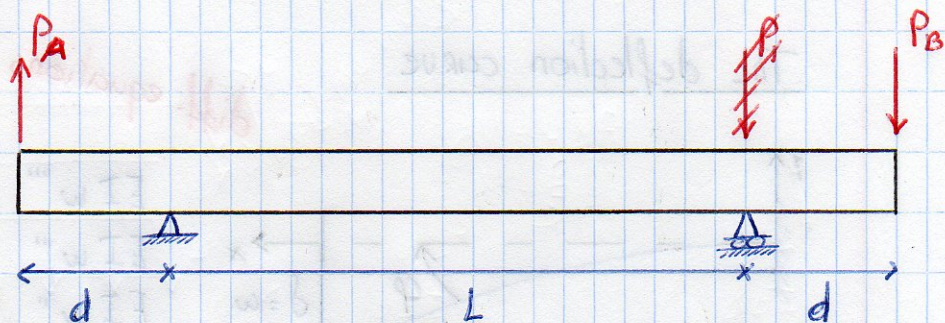
$$\begin{aligned}
 \delta_{kw} &= \phi_A \cdot L_{AB} \\
 &= 8 \frac{F}{EI} \cdot 1
 \end{aligned}$$

$$EI w' = 4F \cdot 4 - \frac{F \cdot 4^2}{2} \Leftrightarrow w' = \phi_A = \frac{8F}{EI}$$

$$\delta_B = \delta_F + \delta_{kw} = \frac{64}{3} \frac{F}{EI} + 8 \frac{F}{EI} = \frac{88}{3} \frac{F}{EI}$$

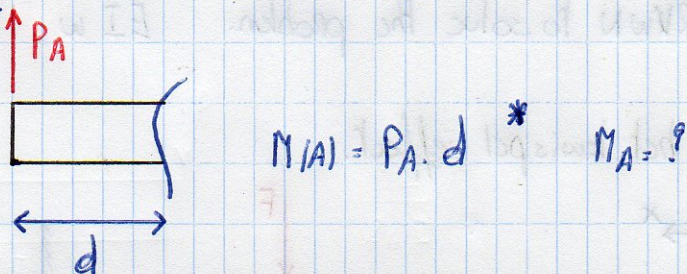
Q: Determine the abs. displacement of B.

Example:



$$w''(x) = -\frac{Px}{6EI} (L-x)^2 \text{ at } d$$

Answer:



$$w = -\frac{Px}{6EI} (L-x)^2 = -\frac{P}{6EI} (xL^2 - 2Lx^2 + x^3)$$

$$w' = -\frac{P}{6EI} (L^2 - 4Lx + 3x^2)$$

$$w'' = -\frac{P}{6EI} (L - 4L + 6x)$$

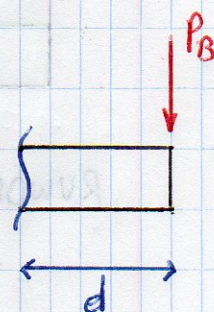
$$M_A = EI w''(x=0) = -\frac{P}{6} (-4L) = \frac{2}{3} PL$$

$$* \frac{2}{3} PL = P_A \cdot d \Rightarrow P_A = \frac{2}{3} \frac{PL}{d}$$

$$M_B^* = -P_B \cdot d$$

$$* M_B = EI w''(x=L) = -\frac{PL}{3} \quad -\frac{PL}{3} = -P_B \cdot d$$

$$\Rightarrow P_B = \frac{PL}{3d}$$



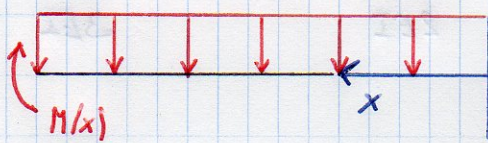
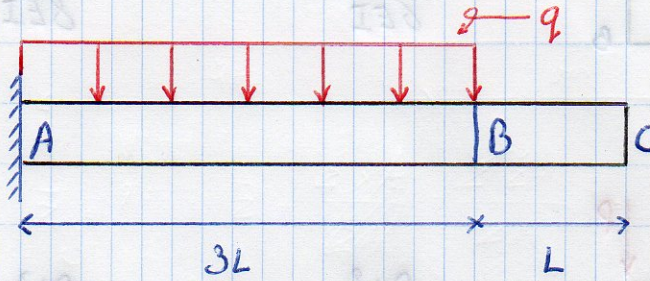
Q: Compute PA and PB.

Moment - Area Methode

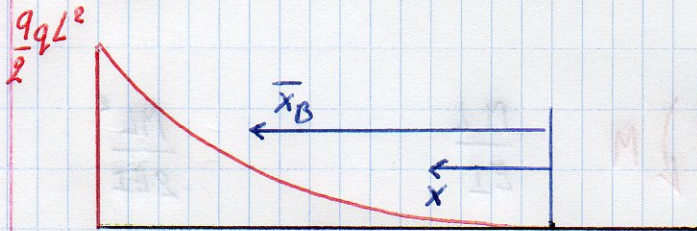
$$\theta_{B/A} = \int_A^B \frac{M}{EI} dx$$

$$w_{B/A} = \bar{x} \int_A^B \frac{M}{EI} dx$$

Example:



$$M(x) = \frac{qx^2}{2}$$



$$M_A = \int_B^A M(x) dx = \int_0^{3L} \frac{qx^2}{2} dx = \frac{q}{2} qL^2$$

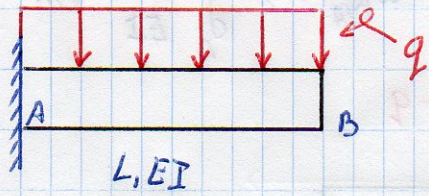
$$\bar{x} = \frac{\int_B^A x M(x) dx}{\int_B^A M(x) dx} = \frac{\int_0^{3L} \frac{qx^3}{2} dx}{\frac{q}{2} qL^2} = \frac{q}{4} L$$

$$w_{B/A} = \bar{x} \int_B^A \frac{M(x)}{EI} dx = \frac{1}{EI} \left[\frac{q}{4} L \cdot \frac{q}{2} qL^3 \right] = \frac{81}{8} \frac{qL^4}{EI}$$

$$\theta_{B/A} = \int_B^A \frac{M(x)}{EI} dx = \frac{q}{2} \frac{qL^3}{EI}$$

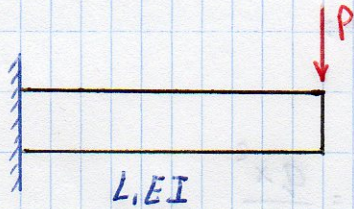
$$w_C = w_B + \theta_B \cdot L = \frac{81}{8} \frac{qL^4}{EI} + \frac{q}{2} \frac{qL^4}{EI} = \frac{117}{8} \frac{qL^4}{EI}$$

Vergeet-me-nietjes



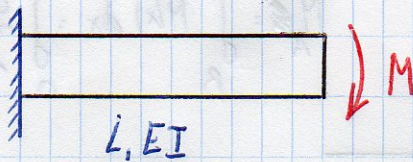
$$\frac{qL^3}{6EI}$$

$$w_B = \frac{qL^4}{8EI}$$



$$\frac{PL^2}{2EI}$$

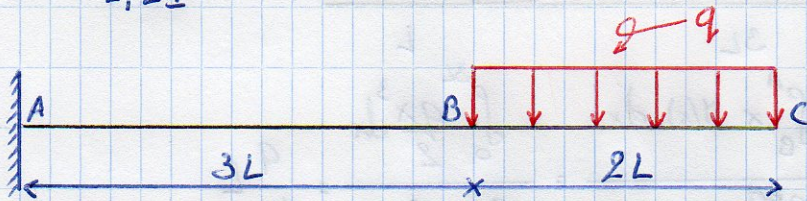
$$\frac{PL^3}{3EI}$$



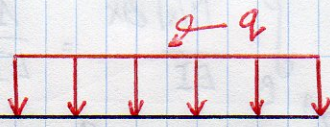
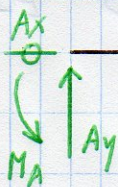
$$\frac{ML}{EI}$$

$$\frac{ML^2}{2EI}$$

Example:



FBD:

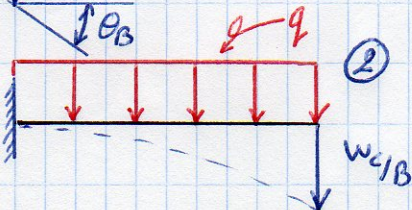
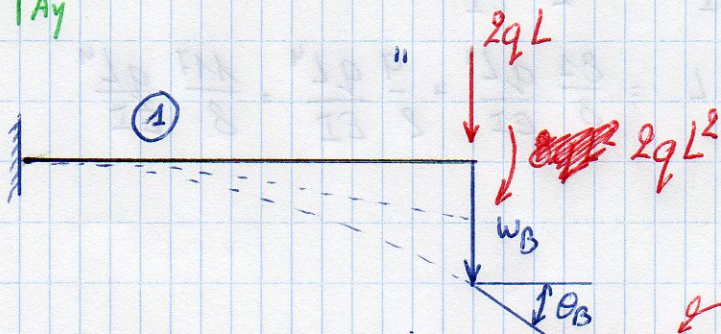


$$\sum F_y \uparrow = 0 \quad \sum F_x \rightarrow = 0$$

$$A_y = 2qL \quad A_x = 0$$

$$\sum M_A \uparrow = 0$$

$$M_A = 2qL \cdot 4L = 8qL^2$$



Q: $w_C = ?$

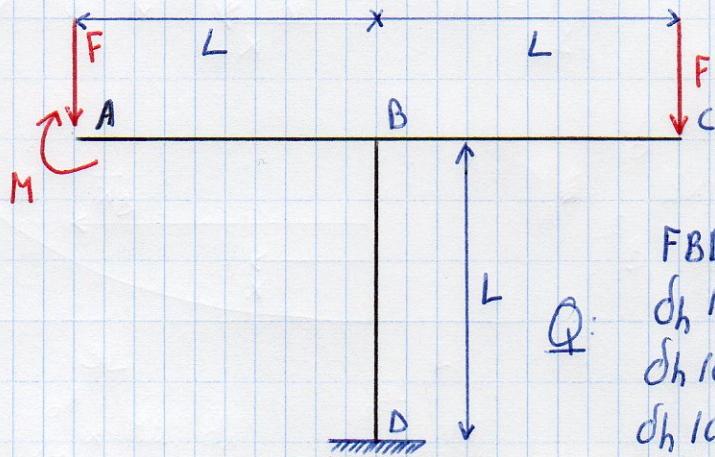
$$\textcircled{1} \quad w_B = \frac{2qL \cdot (3L)^3}{3EI} + \frac{2qL^2 (3L)^2}{2EI} = \frac{27qL^4}{EI}$$

$$\theta_B = \frac{2qL (3L)^2}{2EI} + \frac{2qL^2 (3L)}{EI} = \frac{15qL^3}{EI}$$

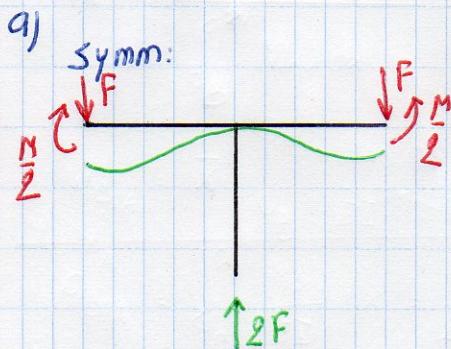
$$\textcircled{2} \quad w_{C/B} = \frac{q(2L)^4}{8EI} = \frac{2qL^4}{EI}$$

$$\textcircled{1} + \textcircled{2} \Rightarrow w_C = w_B + \theta_B \cdot 2L + w_{C/B} = \frac{59qL^4}{EI}$$

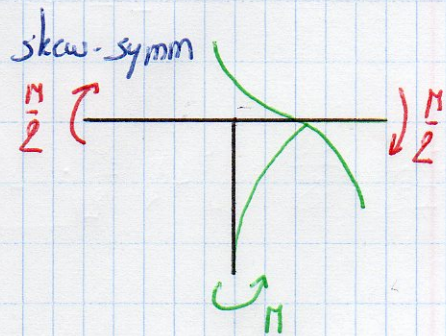
Example:



FBD?
 $\delta_h(C) + \delta_v(C)$ due skew-symm
 $\delta_h(C) + \delta_v(C)$ due symm
 $\delta_h(C) + \delta_v(C)$ due total



$$\begin{aligned} \delta_h^s(B) &= 0 \\ \delta_v^s(B) &= 0 \\ \phi_B^s &= 0 \end{aligned}$$



$$\begin{aligned} \delta_h^{ss}(B) &= \frac{ML^2}{2EI} \\ \delta_v^{ss}(B) &= 0 \\ \phi_B^{ss} &= \frac{ML}{EI} \end{aligned}$$

⋮