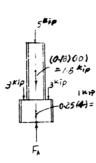
**1–1.** Determine the resultant internal normal force acting on the cross section through point A in each column. In (a), segment BC weighs 180 lb/ft and segment CD weighs 250 lb/ft. In (b), the column has a mass of 200 kg/m.

(a)

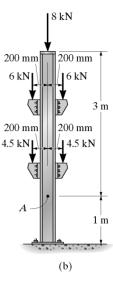
+ ↑ Σ 
$$F_y = 0$$
;  $F_A - 1.0 - 3 - 3 - 1.8 - 5 = 0$   
 $F_A = 13.8 \text{ kip}$  Ans



10 ft 8 in. 8 in. 3 kip C 4 ft A 4 ft

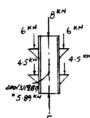
(a)

5 kip



(b)

$$+ \uparrow \Sigma F_y = 0;$$
  $F_A - 4.5 - 4.5 - 5.89 - 6 - 6 - 8 = 0$   
 $F_A = 34.9 \text{ kN}$  Ans



**1–2.** Determine the resultant internal torque acting on the cross sections through points C and D of the shaft. The shaft is fixed at B.



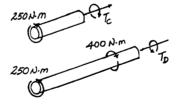
Equations of Equilibrium:

$$(+ 250 - T_C = 0)$$

$$T_C = 250 \text{ N} \cdot \text{m}$$
 Ans

$$(+ 250 - 400 + T_0 = 0)$$

$$T_D = 150 \text{ N} \cdot \text{m}$$
 A



**1–3.** Determine the resultant internal torque acting on the cross sections through points B and C.

 $\Sigma M_x = 0;$   $T_B + 350 - 500 = 0$ 

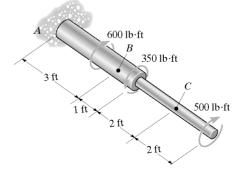
 $T_B = 150 \text{ lb} \cdot \text{ft}$  Ans

 $\Sigma M_x = 0; \qquad T_C - 500 = 0$ 

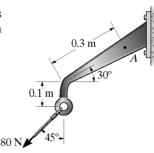
 $T_C = 500 \text{ lb} \cdot \text{ft}$  Ans

8 350 16-ft 500 16-ft





\*1–4. A force of 80 N is supported by the bracket as shown. Determine the resultant internal loadings acting on the section through point A.



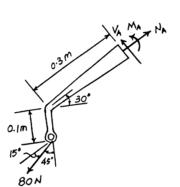
Equations of Equilibrium:

$$+_{A}\Sigma F_{x'} = 0;$$
  $N_{A} - 80 \cos 15^{\circ} = 0$   
 $N_{A} = 77.3 \text{ N}$  Ans

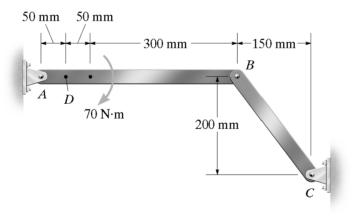
$$\nabla + \Sigma F_{y'} = 0;$$
  $V_A - 8 \text{ 0sin } 15^\circ = 0$   
 $V_A = 20.7 \text{ N}$  Ans

$$(+ \Sigma M_A = 0: M_A + 80 \cos 45^{\circ} (0.3\cos 30^{\circ}) - 80\sin 45^{\circ} (0.1 + 0.3\sin 30^{\circ}) = 0 M_A = -0.555 \text{ N} \cdot \text{m}$$
 Ans

Negative sign indicates that  $M_A$  acts in the opposite direction to that shown on FBD.



**1–5.** Determine the resultant internal loadings acting on the cross section through point D of member AB.



Segment AD:

$$\stackrel{+}{\to} \Sigma F_x = 0;$$
  $N_D + 131.25 = 0;$   $N_D = -131 \text{ N}$ 

Ans

$$+ \oint \sum F_y = 0;$$
  $V_D + 175 = 0;$   $V_D = -17$ 

An

$$(+ \Sigma M_D = 0;$$
  $M_D + 175(0.05) = 0;$   $M_D = -8.75 \text{ N} \cdot \text{m}$  Ans

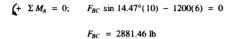
175 H VB ND

**1–6.** The beam AB is pin supported at A and supported by a cable BC. Determine the resultant internal loadings acting on the cross section at point D.

$$\theta = \tan^{-1}\left(\frac{6}{8}\right) = 36.87^{\circ}$$

$$\phi = \tan^{-1}\left(\frac{10}{8}\right) - 36.87^{\circ} = 14.47^{\circ}$$

Member AB:



Segment BD:

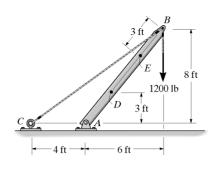
$$+ \Sigma F_x = 0$$
;  $-N_D - 2881.46 \cos 14.47^\circ - 1200 \cos 36.87^\circ = 0$   
 $N_D = -3750 \text{ lb} = -3.75 \text{ kip}$  Ans

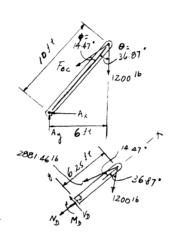
$$V_D + 2881.46 \sin 14.47^{\circ} - 1200 \sin 36.87^{\circ} = 0$$

$$V_D = 0 \qquad \qquad \text{Ans}$$

(+ 
$$\Sigma M_D = 0$$
; 2881.46 sin 14.47°(6.25) - 1200 sin 36.87°(6.25) -  $M_D = 0$   
 $M_D = 0$  Ans

Notice that member AB is the two-force member; therefore the shear force and moment are zero.





**1–7.** Solve Prob. 1–6 for the resultant internal loadings acting at point E.

$$\theta = \tan^{-1}\left(\frac{6}{8}\right) = 36.87^{\circ}$$

$$\phi = \tan^{-1}\left(\frac{10}{8}\right) - 36.87^{\circ} = 14.47^{\circ}$$

Member AB:

$$C + \Sigma M_A = 0;$$
  $F_{BC} \sin 14.47^{\circ}(10) - 1200(6) = 0$   $F_{BC} = 2881.46 \text{ lb}$ 

Segment BE:

$$+\Sigma F_x = 0;$$
  $-N_E - 2881.46 \cos 14.47^\circ - 1200 \cos 36.87^\circ = 0$ 

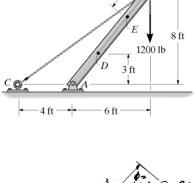
$$N_E = -3750 \text{ lb} = -3.75 \text{ kip}$$

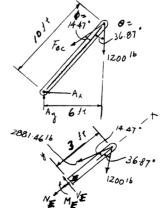
$$+ \Sigma F_y = 0;$$
  $V_E + 2881.46 \sin 14.47^\circ - 1200 \sin 36.87^\circ = 0$ 

$$E = 0$$
 Ar

$$\zeta$$
+  $\Sigma M_E = 0$ ; 2881.46 sin 14.47°(3) - 1200 sin 36.87°(3) -  $M_E = 0$ 

$$M_E = 0$$
 Ans





Notice that member AB is the two-force member; therefore the shear force and moment are zero.

\*1-8. The boom DF of the jib crane and the column DEhave a uniform weight of 50 lb/ft. If the hoist and load weigh 300 lb, determine the resultant internal loadings in the crane on cross sections through points A, B, and C.

#### Equations of Equilibrium: For point A

$$\stackrel{+}{\leftarrow} \Sigma F = 0$$

$$N_A = 0$$

$$+ \uparrow \Sigma F_{\nu} = 0;$$
  $V_{\lambda} - 150 - 300 = 0$ 

$$\int_{A} + \sum M_A = 0;$$
  $-M_A - 150(1.5) - 300(3) = 0$   $M_A = -1125 \text{ lb} \cdot \text{ft} = -1.125 \text{ kip} \cdot \text{ft}$  Ans

Negative sign indicates that MA acts in the opposite direction to that shown on FBD.

# Equations of Equilibrium: For point B

$$\stackrel{\star}{\leftarrow} \Sigma F_{x} = 0;$$

$$N_B = 0$$

$$+ \uparrow \Sigma F_v = 0;$$
  $V_B - 550 - 300 = 0$ 

$$V_B = 850 \text{ lb}$$

$$(+ \Sigma M_B = 0; -M_B - 550(5.5) - 300(11) = 0$$
  
 $M_B = -6325 \text{ lb} \cdot \text{ft} = -6.325 \text{ kip} \cdot \text{ft}$  Ans

Negative sign indicates that  $M_B$  acts in the opposite direction to that shown on FBD.

# Equations of Equilibrium: For point C

$$\stackrel{+}{\leftarrow} \Sigma F_r = 0;$$

$$V_C = 0$$

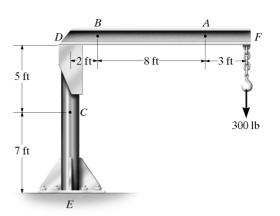
$$+ \uparrow \Sigma F_{y} = 0;$$
  $-N_{C} - 250 - 650 - 300 = 0$ 

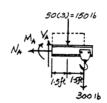
$$N_C = -1200 \text{ lb} = -1.20 \text{ kip}$$

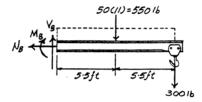
$$\int_C + \sum M_C = 0; \qquad -M_C - 650(6.5) - 300(13) = 0$$

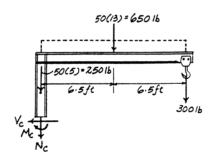
$$M_C = -8125 \text{ lb} \cdot \text{ft} = -8.125 \text{ kip} \cdot \text{ft} \quad \text{Ans}$$

Negative signs indicate that N<sub>C</sub> and M<sub>C</sub> act in the opposite direction to that shown on FBD.









**1–9.** The force  $F = 80 \, \text{lb}$  acts on the gear tooth. Determine the resultant internal loadings on the root of the tooth, i.e., at the centroid point A of section a-a.

# Equations of Equilibrium: For section a-a

$$+ \sum F_{r'} = 0;$$

$$V_A - 80 \cos 15^\circ = 0$$

$$V_{\rm c} = 77.31$$

Ans

$$\times + \Sigma F_{i'} = 0$$
;

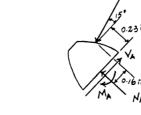
$$N_A - 80 \sin 15^\circ = 0$$

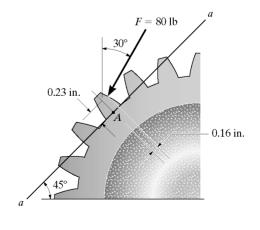
$$N_A = 20.7 \text{ lb}$$

$$\begin{cases} + \sum M_A = 0; & -M_A - 80 \sin 15^{\circ}(0.16) \\ & + 80 \cos 15^{\circ}(0.23) = 0 \end{cases}$$

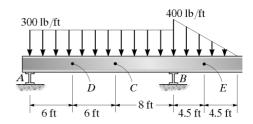
$$+80 \cos 15^{\circ}(0.23) = 0$$

$$M_A = 14.5 \text{ lb} \cdot \text{in}.$$





**1–10.** The beam supports the distributed load shown. Determine the resultant internal loadings on the cross section through point C. Assume the reactions at the supports A and B are vertical.



### Support Reactions:

$$A + \sum M_A = 0;$$
  $B_y(20) - 6(10) - 1.8(23) = 0$   
 $B_y = 5.07 \text{ kip}$ 

$$+ \uparrow \Sigma F_y = 0;$$
  $A_y + 5.07 - 6 - 1.8 = 0$   
 $A_y = 2.73 \text{ kip}$ 

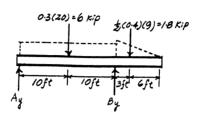
Equations of Equilibrium: For point C

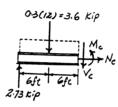
$$\stackrel{+}{\rightarrow} \Sigma F_x = 0;$$
  $N_C = 0$  Ans

+ ↑ Σ 
$$F_y = 0$$
; 2.73 – 3.60 –  $V_C = 0$   
 $V_C = -0.870 \text{ kip}$  Ans

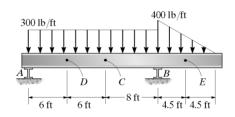
$$M_C = 0;$$
  $M_C + 3.60(6) - 2.73 (12) = 0$   
 $M_C = 11.2 \text{ kip} \cdot \text{ft}$  Ans

Negative sign indicates that  $V_C$  acts in the opposite direction to that shown on FBD.





**1–11.** The beam supports the distributed load shown. Determine the resultant internal loadings on the cross sections through points D and E. Assume the reactions at the supports A and B are vertical.



# Equations of Equilibrium: For point D

$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad N_D = 0 \qquad \text{Ans}$$

+ 
$$\uparrow \Sigma F_y = 0$$
; 2.73 – 1.8 –  $V_D = 0$   
 $V_D = 0.930 \text{ kip}$  Ans

$$(+\Sigma M_D = 0; M_D + 1.8(3) - 2.73(6) = 0$$
  
 $M_D = 11.0 \text{ kip} \cdot \text{ft}$  Ans



$$\int_{A} + \sum M_A = 0;$$
  $B_y(20) - 6(10) - 1.8(23) = 0$   $B_y = 5.07 \text{ kip}$ 

+ 
$$\uparrow \Sigma F_y = 0$$
;  $A_y + 5.07 - 6 - 1.8 = 0$   
 $A_y = 2.73 \text{ kip}$ 

### Equations of Equilibrium : For point E

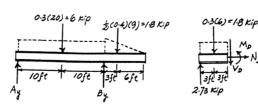
$$\stackrel{\leftarrow}{\leftarrow} \Sigma F_x = 0; \qquad N_E = 0 \qquad \text{Ans}$$

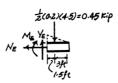
$$+ \uparrow \Sigma F_y = 0; \qquad V_E - 0.45 = 0$$

$$V_E = 0.450 \text{ kip} \qquad \text{Ans}$$

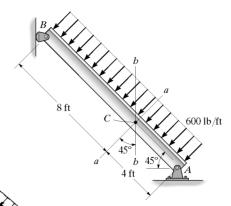
$$M_E = 0;$$
  $-M_E - 0.45(1.5) = 0$   $M_E = -0.675 \text{ kip} \cdot \text{ft}$  Ans

Negative sign indicates that  $\mathbf{M}_{\mathcal{E}}$  acts in the opposite direction to that shown on FBD.





\*1–12. Determine the resultant internal loadings acting on (a) section a–a and (b) section b–b. Each section is located through the centroid, point C.



 $V + \Sigma F_x = 0;$   $N_C + 5.091 \sin 45^\circ = 0$ 

 $N_C = -3.60 \, \text{kip}$  Ans

 $A \Sigma F_{y} = 0;$   $V_{C} + 5.091 \cos 45^{\circ} - 2.4 = 0$ 

 $V_C = -1.20 \text{ kip}$  Ans

 $(+ \Sigma M_C = 0; -M_C - 2.4(2) + 5.091 \cos 45^{\circ}(4) = 0$ 

 $M_C = 9.60 \text{ kip} \cdot \text{ft}$  Ans

(b)

 $\int + \sum M_C = 0;$ 

 $\stackrel{+}{\leftarrow} \Sigma F_x = 0; \qquad N_C + 2.4 \cos 45^\circ = 0$ 

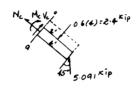
 $N_C = -1.70 \text{ kip}$  Ans

 $+ \uparrow \Sigma F_v = 0;$   $V_C + 5.091 - 2.4 \sin 45^\circ = 0$ 

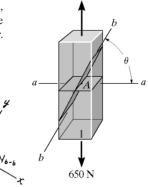
 $V_{\rm C} = -3.39 \, {\rm kip}$  Ans

 $-M_C - 2.4(2) + 5.091\cos 45^{\circ}(4) = 0$ 

 $M_C = 9.60 \text{ kip} \cdot \text{ft}$  Ans



**1–13.** Determine the resultant internal normal and shear forces in the member at (a) section a–a and (b) section b–b, each of which passes through point A. Take  $\theta = 60^{\circ}$ . The 650-N load is applied along the centroidal axis of the member.



650 N

Equations of Equilibrium: For section a-a

+ ↑ Σ  $F_y = 0$ ; 650 -  $N_{a-a} = 0$  $N_{a-a} = 650 \text{ N}$  Ans

 $\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad V_{a-a} = 0 \qquad \text{Ans}$ 

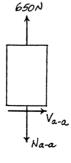
Equations of Equilibrium: For section b-b

 $\sqrt{\Sigma} F_y = 0;$  650cos 30° -  $V_{b-b} = 0$  $V_{b-b} = 563 \text{ N}$ 

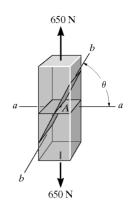
 $L_{-k} = 563 \text{ N}$  Ans

Ans

 $+\Sigma F_x = 0;$   $N_{b-b} - 650 \sin 30^\circ = 0$  $N_{b-b} = 325 \text{ N}$ 



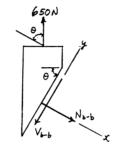
**1–14.** Determine the resultant internal normal and shear forces in the member at section b–b, each as a function of  $\theta$ . Plot these results for  $0^{\circ} \le \theta \le 90^{\circ}$ . The 650-N load is applied along the centroidal axis of the member.

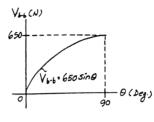


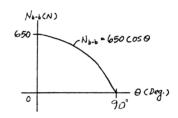
Equations of Equilibrium: For section b-b

$$\begin{array}{c} \searrow + \Sigma F_x = 0; \qquad N_{b-b} - 650 \cos \theta = 0 \\ \\ N_{b-b} = 650 \cos \theta & \text{Ans} \end{array}$$

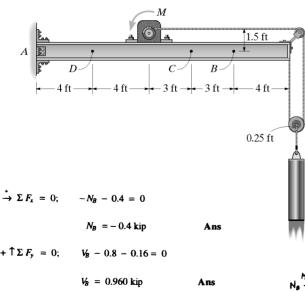
$$+ \Sigma F_y = 0;$$
  $-V_{b-b} + 650 \sin \theta = 0$   
  $V_{b-b} = 650 \sin \theta$  Ans







**1–15.** The 800-lb load is being hoisted at a constant speed using the motor M, which has a weight of 90 lb. Determine the resultant internal loadings acting on the cross section through point B in the beam. The beam has a weight of 40 lb/ft and is fixed to the wall at A.

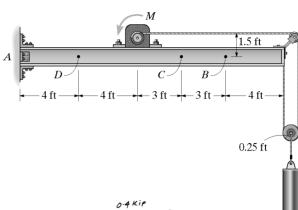


 $-M_B - 0.16(2) - 0.8(4.25) + 0.4(1.5) = 0$ 

Ans

 $M_B = -3.12 \text{ kip} \cdot \text{ ft}$ 

\*1–16. Determine the resultant internal loadings acting on the cross section through points C and D of the beam in Prob. 1–15.



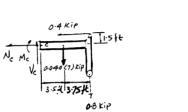
For point C:

$$\stackrel{\leftarrow}{\leftarrow} \Sigma F_x = 0; \quad N_C + 0.4 = 0; \quad N_C = -0.4 \text{ kip} \qquad \text{Ans}$$

$$+ \uparrow \Sigma F_y = 0; \quad V_C - 0.8 - 0.04 (7) = 0; \quad V_C = 1.08 \text{ kip} \qquad \text{Ans}$$

$$\left( + \Sigma M_C = 0; \quad - M_C - 0.8(7.25) - 0.04(7)(3.5) + 0.4(1.5) = 0 \right)$$

$$M_C = -6.18 \text{ kip} \cdot \text{ft} \qquad \text{Ans}$$



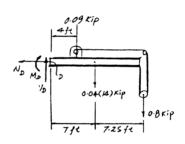
For point D:

$$\stackrel{\leftarrow}{\leftarrow} \Sigma F_x = 0; \qquad N_D = 0 \qquad \text{Ans}$$

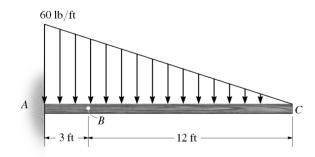
$$+ \uparrow \Sigma F_y = 0; \qquad V_D - 0.09 - 0.04(14) - 0.8 = 0; \qquad V_D = 1.45 \text{ kip} \qquad \text{Ans}$$

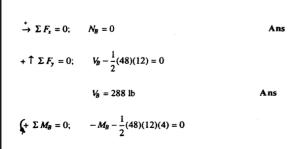
$$\stackrel{\leftarrow}{\leftarrow} \Sigma M_D = 0; \qquad - M_D - 0.09(4) - 0.04(14)(7) - 0.8(14.25) = 0$$

$$M_D = -15.7 \text{ kip} \cdot \text{ ft} \qquad \text{Ans}$$

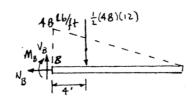


**1–17.** Determine the resultant internal loadings acting on the cross section at point B.

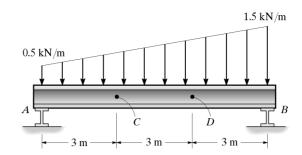




 $M_B = -1152 \text{ lb} \cdot \text{ft} = -1.15 \text{ kip} \cdot \text{ft}$ 



**1–18.** The beam supports the distributed load shown. Determine the resultant internal loadings acting on the cross section through point C. Assume the reactions at the supports A and B are vertical.



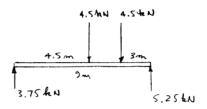
$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad N_C = 0$$
 Ans

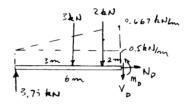
+ 
$$\downarrow \Sigma F_y = 0$$
;  $V_C + 0.5 + 1.5 - 3.75 = 0$ 

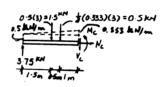
$$V_C = 1.75 \text{ kN}$$
 Ans

$$\sum M_C = 0;$$
  $M_C + 0.5(1) + 1.5(1.5) - 3.75(3) = 0$ 

$$M_C = 8.50 \text{ kN} \cdot \text{m}$$
 Ans







**1–19.** Determine the resultant internal loadings acting on the cross section through point D in Prob. 1–18.

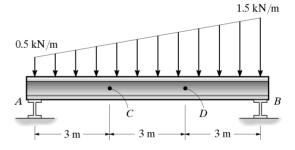
$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad N_D = 0 \qquad \text{Ans}$$

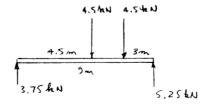
$$+ \uparrow \Sigma F_{y} = 0;$$
  $3.75 - 3 - 2 - V_{D} = 0$ 

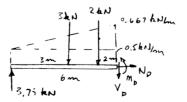
$$V_D = -1.25 \text{ kN}$$
 Ans

$$(+ \Sigma M_D = 0; M_D + 2(2) + 3(3) - 3.75(6) = 0$$

$$M_D = 9.50 \,\mathrm{kN \cdot m}$$
 Ans







\*1–20. The wishbone construction of the power pole supports the three lines, each exerting a force of 800 lb on the bracing struts. If the struts are pin connected at A, B, and C, determine the resultant internal loadings at cross sections through points D, E, and F.

Support Reaction: FBD(a) and (b).

$$(+\Sigma M_A = 0; B_y(4) + B_x(3) - 800(4) = 0$$
 [1]

$$(+\Sigma M_C = 0; B_x(3) + 800(4) - B_y(4) - 800(4) = 0$$
 [2]

Solving Eq. [1] and [2] yields

$$B_{y} = 400.0 \text{ lb}$$
  $B_{x} = 533.33 \text{ lb}$ 

From FBD (a)

$$\stackrel{+}{\to} \Sigma F_x = 0; \qquad 533.33 - A_x = 0 \qquad A_x = 533.33 \text{ lb} 
+ \uparrow \Sigma F_y = 0; \qquad A_y - 800 - 400.0 = 0 \qquad A_y = 1200 \text{ lb}$$

From FBD (b)

$$\overset{\star}{\to} \Sigma F_x = 0;$$
  $C_x - 533.33 = 0$   $C_x = 533.33 \text{ lb}$   
+  $\uparrow \Sigma F_y = 0;$   $C_y + 400.0 - 800 - 800 = 0$   $C_y = 1200 \text{ lb}$ 

Equations of Equilibrium: For point D [FBD(c)].

$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad V_D = 0 \qquad \text{Ans}$$

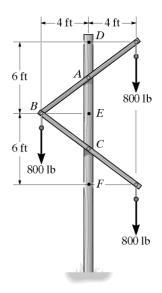
$$+ \uparrow \Sigma F_y = 0; \qquad N_D = 0 \qquad \text{Ans}$$

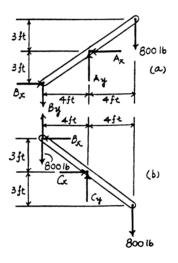
$$(+ \Sigma M_D = 0; \qquad M_D = 0 \qquad \text{Ans}$$

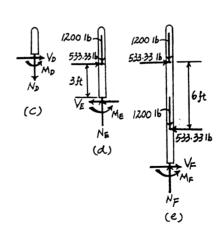
For point E [FBD(d)].

For point F [FBD(e)].

$$\stackrel{+}{\to} \Sigma F_x = 0; \qquad V_F + 533.33 - 533.33 = 0 \qquad V_F = 0 \qquad \text{Ans} \\
+ \uparrow \Sigma F_y = 0; \qquad N_F - 1200 - 1200 = 0 \qquad N_F = 2400 \text{ lb} \qquad \text{Ans} \\
(+ \Sigma M_F = 0; \qquad M_F - 533.33(6) = 0 \qquad M_F = 3200 \text{ lb} \cdot \text{ft} \qquad \text{Ans}$$







**1–21.** The drum lifter suspends the 500-lb drum. The linkage is pin connected to the plate at A and B. The gripping action on the drum chime is such that only horizontal and vertical forces are exerted on the drum at G and H. Determine the resultant internal loadings on the cross section through point I.

Equations of Equilibrium: Members AC and BD are two-force members.

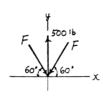
$$+ \uparrow \Sigma F_{y} = 0;$$
 500 - 2F sin 60° = 0  
F = 288.7 lb

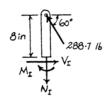
Equations of Equilibrium: For point I

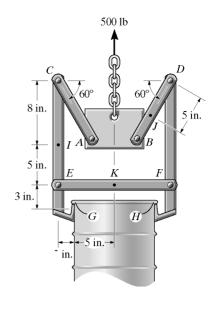
$$\stackrel{+}{\to} \Sigma F_x = 0;$$
  $V_I - 288.7 \cos 60^\circ = 0$   $V_I = 144 \text{ lb}$  Ans

$$+ \uparrow \Sigma F_y = 0;$$
 288.7 sin 60° -  $N_I = 0$   
 $N_I = 250 \text{ lb}$  Ans

$$(+\Sigma M_l = 0;$$
 288.7cos 60°(8) -  $M_l = 0$   
 $M_l = 1154.7$  lb·in. = 1.15 kip·in. Ans







**1–22.** Determine the resultant internal loadings on the cross sections through points K and J on the drum lifter in Prob. 1–21.

Equations of Equilibrium: Members AC and BD are two force members.

$$+ \uparrow \Sigma F_y = 0;$$
 500 - 2F sin 60° = 0  
F = 288.7 lb

Equations of Equilibrium: For point J

$$F_{ij} + \Sigma F_{j'} = 0;$$
  $V_{j} = 0$  Ans  $V_{j} = 0$   $V_{j} = 0$   $V_{j} = -289 \text{ lb}$  Ans

$$(+\Sigma M_f = 0;$$
  $M_f = 0$  Ans

Negative sign indicates that  $N_J$  acts in the opposite direction to that shown on FBD.

Support Reactions: For member DFH

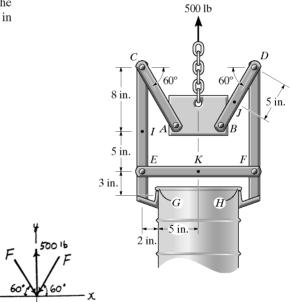
$$F_{EF}(3) - 288.7\cos 60^{\circ}(16)$$
  
+ 288.7 sin 60°(2) = 0  
 $F_{EF} = 603.1 \text{ lb}$ 

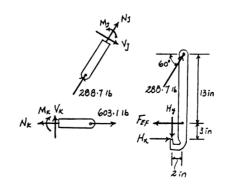
Equations of Equilibrium: For point K

$$\stackrel{\leftarrow}{\leftarrow} \Sigma F_x = 0; \quad N_K - 603. = 0 \quad N_K = 603 \text{ lb} \quad \text{Ans}$$

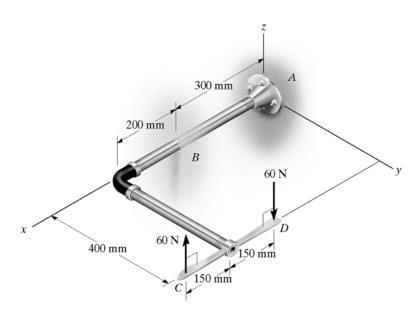
$$+ \uparrow \Sigma F_y = 0; \quad V_K = 0 \quad \text{Ans}$$

$$\stackrel{\leftarrow}{\leftarrow} \Sigma M_K = 0; \quad M_K = 0 \quad \text{Ans}$$





**1–23.** The pipe has a mass of 12 kg/m. If it is fixed to the wall at A, determine the resultant internal loadings acting on the cross section at B. Neglect the weight of the wrench CD.



$$\Sigma F_x = 0; \qquad (N_B)_x = 0$$

$$\Sigma F_{y} = 0; \qquad (V_{B})_{y} = 0$$

$$\sum F_z = 0;$$

 $\Sigma M_{y} = 0;$ 

$$(V_B)_z - 60 + 60 - (0.2)(12)(9.81) - (0.4)(12)(9.81) = 0$$

$$(V_B)_z = 70.6 \text{ N}$$

Ans

$$\Sigma M_x = 0;$$
  $(T_B)_x + 60(0.4) - 60(0.4) - (0.4)(12)(9.81)(0.2) = 0$ 

 $(M_B)_y + (0.2)(12)(9.81)(0.1) + (0.4)(12)(9.81)(0.2) - 60(0.3) = 0$ 

$$(M_B)_y = 6.23 \text{ N} \cdot \text{m}$$

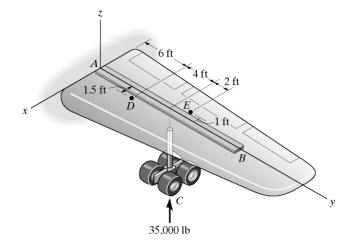
 $(T_B)_x = 9.42 \text{ N} \cdot \text{m}$ 

Ans

$$\Sigma M_z = 0; \qquad (M_B)_z = 0$$

Ans

\*1–24. The main beam AB supports the load on the wing of the airplane. The loads consist of the wheel reaction of 35,000 lb at C, the 1200-lb weight of fuel in the tank of the wing, having a center of gravity at D, and the 400-lb weight of the wing, having a center of gravity at E. If it is fixed to the fuselage at A, determine the resultant internal loadings on the beam at this point. Assume that the wing does not transfer any of the loads to the fuselage, except through the beam.



$$\sum F_x = 0; \qquad (V_A)_x = 0$$

Ans

$$\Sigma F_{v} = 0; \qquad (N_{A})_{v} = 0$$

Ans

$$\Sigma F_z = 0;$$
  $(V_A)_z - 1200 - 400 + 35000 = 0$ 

$$(V_A)_z = -33.4 \text{ kip}$$

Ans

$$\Sigma M_x = 0;$$
  $(M_A)_x - 1200(6) + 35000(10) - 400(12) = 0$ 

$$(M_A)_x = 338 \text{ kip} \cdot \text{ft}$$

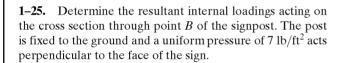
Ans

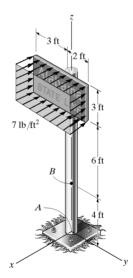
$$\Sigma M_y = 0;$$
  $(T_A)_y + 1200(1.5) - 400(1) = 0$ 

 $(T_A)_v = -1.40 \text{ kip} \cdot \text{ft}$ 

$$\sum M_z = 0; \qquad (M_A)_z = 0$$

Ans





$$\Sigma F_x = 0;$$
  $(V_B)_x - 105 = 0;$ 

$$(V_B)_x = 105 \text{ lb}$$

$$\Sigma F_{y} = 0; \qquad (V_{B})_{y} = 0$$

Ans

$$\sum F_z = 0; \qquad (N_B)_z = 0$$

Ans

$$\sum M_x = 0; \qquad (M_B)_x = 0$$

.

Ans

$$\Sigma M_{y} = 0;$$

$$(M_B)_v - 105(7.5) = 0;$$

$$(M_B)_v = 788 \text{ lb} \cdot \text{ft}$$

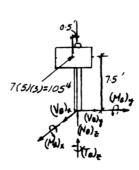
A ne

$$\Sigma M_2 = 0;$$

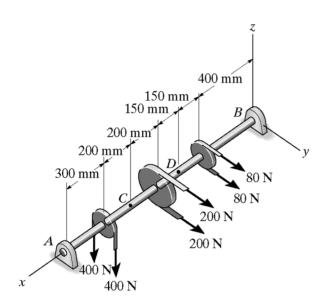
$$(T_B)_z - 105(0.5) = 0;$$

$$(T_B)_z = 52.5 \text{ lb} \cdot \text{ft}$$

Ans



**1–26.** The shaft is supported at its ends by two bearings A and B and is subjected to the forces applied to the pulleys fixed to the shaft. Determine the resultant internal loadings acting on the cross section through point D. The 400-N forces act in the -z direction and the 200-N and 80-N forces act in the +y direction. The journal bearings at A and B exert only y and z components of force on the shaft.



## Support Reactions:

$$\Sigma M_z = 0;$$
 160(0.4) + 400(0.7) -  $A_y$  (1.4) = 0  
 $A_y = 245.71 \text{ N}$ 

$$\Sigma F_y = 0;$$
  $-245.71 - B_y + 400 + 160 = 0$   
 $B_y = 314.29 \text{ N}$ 

$$\Sigma M_v = 0;$$
 800(1.1)  $-A_z$  (1.4) = 0  $A_z = 628.57 \text{ N}$ 

$$\Sigma F_z = 0;$$
  $B_z + 628.57 - 800 = 0$   $B_z = 171.43 \text{ N}$ 

#### Equations of Equilibrium: For point D

$$\Sigma F_x = 0; \qquad (N_D)_x = 0 \qquad \text{Ans}$$

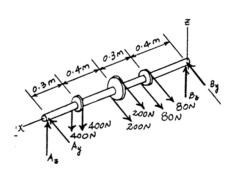
$$\Sigma F_y = 0;$$
  $(V_D)_y - 314.29 + 160 = 0$   $(V_D)_y = 154 \text{ N}$  Ans

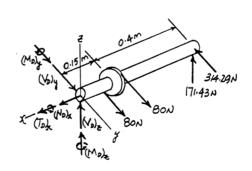
$$\Sigma F_z = 0;$$
  $171.43 + (V_D)_z = 0$   $(V_D)_z = -171 \text{ N}$  Ans

$$\sum M_x = 0; \qquad (T_D)_x = 0 \qquad \text{Ans}$$

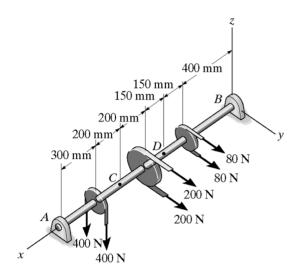
$$\Sigma M_y = 0;$$
 171.43 (0.55) +  $(M_D)_y = 0$   
 $(M_D)_y = -94.3 \text{ N} \cdot \text{m}$  Ans

$$\Sigma M_z = 0;$$
 314.29(0.55) - 160(0.15) +  $(M_D)_z = 0$   
 $(M_D)_z = -149 \text{ N} \cdot \text{m}$  Ans





**1–27.** The shaft is supported at its ends by two bearings A and B and is subjected to the forces applied to the pulleys fixed to the shaft. Determine the resultant internal loadings acting on the cross section through point C. The 400-N forces act in the -z direction and the 200-N and 80-N forces act in the +y direction. The journal bearings at A and B exert only y and z components of force on the shaft.



#### Support Reactions:

$$\Sigma M_z = 0;$$
  $160(0.4) + 400(0.7) - A_y(1.4) = 0$   
 $A_y = 245.71 \text{ N}$ 

$$\Sigma F_y = 0;$$
  $-245.71 - B_y + 400 + 160 = 0$   
 $B_y = 314.29 \text{ N}$ 

$$\Sigma M_y = 0;$$
  $800(1.1) - A_z(1.4) = 0$   $A_z = 628.57 \text{ N}$ 

$$\Sigma F_z = 0;$$
  $B_z + 628.57 - 800 = 0$   $B_z = 171.43 \text{ N}$ 

### Equations of Equilibrium: For point C

$$\Sigma F_{x} = 0; \qquad (N_C)_{x} = 0 \qquad \text{Ans}$$

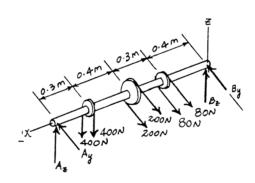
$$\Sigma F_y = 0;$$
  $-245.71 + (V_C)_y = 0$   $(V_C)_y = -246 \text{ N}$  Ans

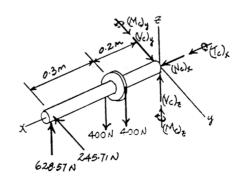
$$\Sigma F_z = 0;$$
 628.57 - 800 +  $(V_C)_z = 0$   
 $(V_C)_z = -171 \text{ N}$  Ans

$$\Sigma M_x = 0; \qquad (T_C)_x = 0 \qquad \text{Ans}$$

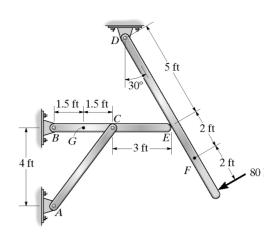
$$\Sigma M_y = 0;$$
  $(M_C)_y - 628.57(0.5) + 800(0.2) = 0$    
  $(M_C)_y = -154 \text{ N} \cdot \text{m}$  Ans

$$\Sigma M_z = 0;$$
  $(M_C)_z - 245.71(0.5) = 0$   $(M_C)_z = -123 \text{ N} \cdot \text{m}$  Ans





\*1-28. Determine the resultant internal loadings acting on the cross section of the frame at points F and G. The contact at E is smooth.



Member DEF:

$$(+ \Sigma M_D = 0; N_E(5) - 80(9) = 0$$

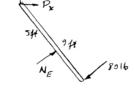
$$N_E = 144 \text{ lb}$$

Member BCE:

$$F_{AC}(\frac{4}{5})(3) - 144 \sin 30^{\circ} (6) = 0$$

$$F_{AC} = 180 \, \text{lb}$$

$$\stackrel{+}{\to} \Sigma F_x = 0;$$
  $B_x + 180 \left(\frac{3}{5}\right) - 144 \cos 30^\circ = 0$ 



$$B_x = 16.708 \text{ lb}$$

$$+ \uparrow \Sigma F_y = 0;$$
  $-B_y + 180(\frac{4}{5}) - 144 \sin 30^\circ = 0$ 

$$B_{\nu} = 72.0 \text{ lb}$$

For point F:  $+\sum F_x=0;$ 

$$+\Sigma F_y = 0;$$
  $V_F - 80 = 0;$   $V_F = 80 \text{ lb}$ 

(+ 
$$\Sigma M_F = 0$$
;  $M_F - 80 (2) = 0$ ;  $M_F = 160 \text{ lb} \cdot \text{ft}$  Ans

For point G:

$$\stackrel{+}{\to} \Sigma F_x = 0;$$
 16.708 -  $N_G = 0;$   $N_G =$ 

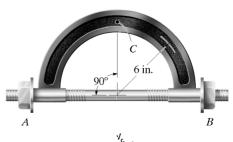
$$N_G = 16.7 \text{ lb}$$
 Ans

Ans

 $+ \uparrow \stackrel{\circ}{\Sigma} F_{\nu} = 0;$  $V_G - 72.0 = 0;$  $V_G = 72.0 \text{ lb}$ 

$$(+ \Sigma M_G = 0;$$
 72 (1.5) -  $M_G = 0;$   $M_G = 108 \text{ lb} \cdot \text{ft Ans}$ 

**1–29.** The bolt shank is subjected to a tension of 80 lb. Determine the resultant internal loadings acting on the cross section at point C.



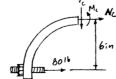
Segment AC:

$$\stackrel{+}{\to} \Sigma F_x = 0; \qquad N_C + 80 = 0; \qquad N_C = -80 \text{ lb}$$

Ans

Ans

$$(+ \Sigma M_C = 0; M_C + 80(6) = 0; M_C = -480 \text{ lb} \cdot \text{in}.$$
 Ans



1-30. The pipe has a mass of 12 kg/m. If it is fixed to the wall at A, determine the resultant internal loadings acting on the cross section through B.

## Equations of Equilibrium: For point B

$$\Sigma F_r = 0$$
;

$$(V_B)_x = 0$$

Ans

$$\Sigma F_y = 0;$$
  $(N_B)_y + \frac{4}{5}(750) = 0$   $(N_B)_y = -600 \text{ N}$ 

Ans

$$\Sigma F_z = 0;$$
  $(V_B)_z - 235.44 - 235.44 - \frac{3}{5}(750) = 0$   
 $(V_B)_z = 921 \text{ N}$  As

$$\Sigma M_x = 0;$$
  $(M_B)_x - 235.44(1) - 235.44(2)$ 

$$-\frac{3}{5}(750)(2)=0$$

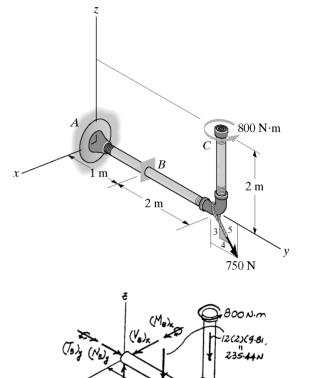
$$-\frac{3}{5}(750)(2) = 0$$

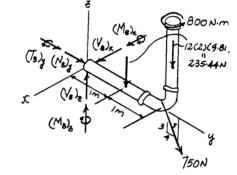
$$(M_B)_x = 1606 \text{ N} \cdot \text{m} \quad \text{Ans}$$

$$\Sigma M_y = 0;$$

$$(T_B)_{y} = 0$$

$$\Sigma M_z = 0;$$
  $(M_B)_z + 800 = 0$   $(M_B)_z = -800 \text{ N} \cdot \text{m}$  Ans





**1–31.** The curved rod has a radius r and is fixed to the wall at B. Determine the resultant internal loadings acting on the cross section through A which is located at an angle  $\theta$ from the horizontal.

## Equations of Equilibrium: For point A

$$+ \sum F_x = 0;$$
  $P\cos \theta - N_A = 0$   
 $N_A = P\cos \theta$ 

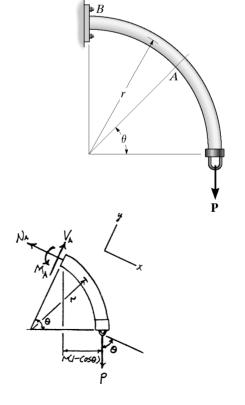
$$V_{A} = P\sin\theta = 0$$

$$V_{A} = P\sin\theta \qquad A$$

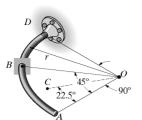
$$V_{A} = P\sin\theta \qquad A$$

$$V_{A} = P(1 - \cos\theta) = 0$$

$$M_{A} = P(1 - \cos\theta) \qquad A$$



\*1–32. The curved rod AD of radius r has a weight per length of w. If it lies in the horizontal plane, determine the resultant internal loadings acting on the cross section through point B. Hint: The distance from the centroid C of segment AB to point O is CO = 0.9745r.

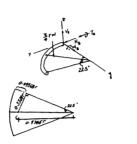


$$\Sigma F_z = 0;$$
  $V_B - \frac{\pi}{4} rw = 0;$   $V_B = 0.785 w r$ 

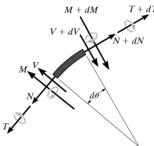
$$\Sigma F_{r} = 0; \qquad N_{R} = 0$$

$$\Sigma M_x = 0;$$
  $T_B - \frac{\pi}{4} rw(0.09968r) = 0;$   $T_B = 0.0783 \text{ w } r^2$  An

$$\Sigma M_y = 0;$$
  $M_B + \frac{\pi}{4} rw(0.3729 r) = 0;$   $M_B = -0.293 w r^2$  Am



**1–33.** A differential element taken from a curved bar is shown in the figure. Show that  $dN/d\theta = V$ ,  $dV/d\theta = -N$ ,  $dM/d\theta = -T$ , and  $dT/d\theta = M$ .



$$\Sigma F_{s} = 0;$$

$$N \cos \frac{d\theta}{2} + V \sin \frac{d\theta}{2} - (N + dN) \cos \frac{d\theta}{2} + (V + dV) \sin \frac{d\theta}{2} = 0$$

$$\Sigma F_{y} = 0;$$

$$N \sin \frac{d\theta}{2} - V \cos \frac{d\theta}{2} + (N + dN) \sin \frac{d\theta}{2} + (V + dV) \cos \frac{d\theta}{2} = 0$$

$$\Sigma M_{x} = 0;$$

$$T \cos \frac{d\theta}{2} + M \sin \frac{d\theta}{2} - (T + dT) \cos \frac{d\theta}{2} + (M + dM) \sin \frac{d\theta}{2} = 0$$

$$\sum M_{1} = 0;$$

$$T \sin \frac{d\theta}{2} - M \cos \frac{d\theta}{2} + (T + dT) \sin \frac{d\theta}{2} + (M + dM) \cos \frac{d\theta}{2} = 0$$

Since 
$$\frac{d\theta}{2}$$
 is small, then  $\sin \frac{d\theta}{2} = \frac{d\theta}{2}$ ,  $\cos \frac{d\theta}{2} = 1$ 

Eq. (1) becomes 
$$Vd\theta - dN + \frac{dVd\theta}{2} = 0$$

Neglecting the second order term,  $Vd\theta - dN = 0$ dN = 0

$$\frac{dN}{d\theta} = V$$
 QEI

Eq.(2) becomes 
$$Nd\theta + dV + \frac{dNd\theta}{2} = 0$$

Neglecting the second order term, 
$$Nd\theta + dV = 0$$

$$\frac{dV}{d\theta} = -N$$
 QED

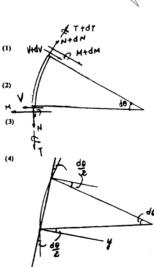
Eq.(3) becomes 
$$Md\theta - dT + \frac{dMd\theta}{2} = 0$$

Neglecting the second order term,  $Md\theta - dT = 0$ 

$$\frac{dT}{d\theta} = M$$
 QED

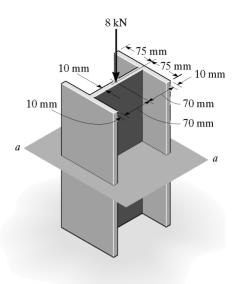
Eq. (4) becomes 
$$Td\theta + dM + \frac{dTd\theta}{2} = 0$$

Neglecting the second order term, 
$$Td\theta + dM = 0$$



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1-34. The column is subjected to an axial force of 8 kN at its top. If the cross-sectional area has the dimensions shown in the figure, determine the average normal stress acting at section a-a. Show this distribution of stress acting over the area's cross section.

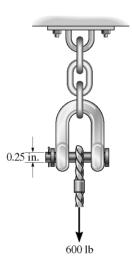


$$A = (2)(150)(10) + (140)(10)$$
  
= 4400 mm<sup>2</sup> = 4.4 (10<sup>-3</sup>) m<sup>2</sup>

$$\sigma = \frac{P}{A} = \frac{8(10^3)}{4A(10^3)} = 1.82 \text{ MPa}$$
 Ans

8 KM

**1–35.** The anchor shackle supports a cable force of 600 lb. If the pin has a diameter of 0.25 in., determine the average shear stress in the pin.



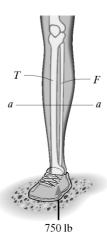
$$+ \uparrow \Sigma F_y = 0; \qquad 2V - 600 = 0$$

$$V = 300 \text{ lb}$$

$$\tau_{\text{avg}} = \frac{V}{A} = \frac{300}{\frac{\pi}{4}(0.25)^2} = 6.11 \text{ ksi}$$
 Ans



\*1–36. While running the foot of a 150-lb man is momentarily subjected to a force which is 5 times his weight. Determine the average normal stress developed in the tibia T of his leg at the mid section a–a. The cross section can be assumed circular, having an outer diameter of 1.75 in. and an inner diameter of 1 in. Assume the fibula F does not support a load.



$$P = 5(150 \text{ lb}) = 750 \text{ lb}$$

$$\sigma = \frac{P}{A} = \frac{750}{\frac{\pi}{4}((1.75)^2 - (1)^2)} = 463 \text{ psi}$$
 Ans

**1–37.** The thrust bearing is subjected to the loads shown. Determine the average normal stress developed on cross sections through points B, C, and D. Sketch the results on a

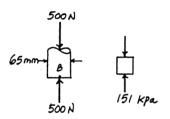
differential volume element located at each section.

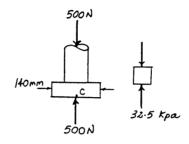
# Average Normal Stress:

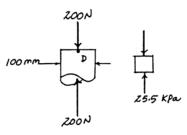
$$\sigma_B = \frac{500}{\frac{\pi}{4}(\frac{65}{1000})^2} = 151 \text{ kPa}$$
 Ans

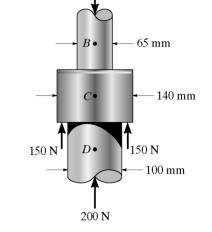
$$\sigma_C = \frac{500}{\frac{\pi}{6}(\frac{140}{1000})^2} = 32.5 \text{ kPa}$$
 Ans

$$\sigma_D = \frac{200}{\frac{\pi}{4}(\frac{100}{1000})^2} = 25.5 \text{ kPa}$$
 Ans



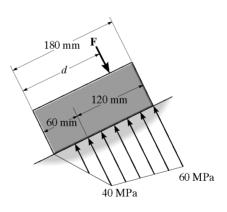






500 N

1-38. The small block has a thickness of 5 mm. If the stress distribution at the support developed by the load varies as shown, determine the force F applied to the block, and the distance d to where it is applied.



 $F = \int \sigma dA$  = volume under stress diagram

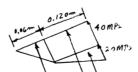
 $F = \frac{1}{2}(0.06)(40)(10^6)(0.005) + (0.120)(40)(10^6)(0.005) + \frac{1}{2}(0.120)(20)(10^6)(0.005)$ 

F = 36 kN

 $F d = \int x(\sigma dA)$ 

 $36.0(10^3)d = \frac{2}{3}(0.06)(\frac{1}{2})(0.06)(40)(10^6)(0.005) + (0.06 + \frac{1}{2}(0.120))(0.120)(40)(10^6)(0.005) + (0.06 + \frac{1}{2}(0.120))(0.120)(0.005) + (0.06 + \frac{1}{2}(0.120))(0.005)(0.005) + (0.06 + \frac{1}{2}(0.120))(0.005)(0.005) + (0.06 + \frac{1}{2}(0.120))(0.005)(0.005) + (0.06 + \frac{1}{2}(0.120))(0.005)(0.005)(0.005) + (0.06 + \frac{1}{2}(0.120))(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.005)(0.$ 

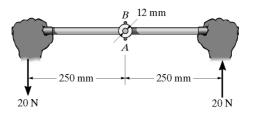
 $(0.06 + \frac{2}{3}(0.120))(\frac{1}{2})(0.120)(20)(10^6)(0.005)$ 



 $36.0(10^3)d = 3960$ 

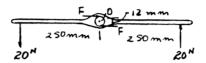
 $d = 0.110 = 110 \,\mathrm{mm}$ 

**1–39.** The lever is held to the fixed shaft using a tapered pin AB, which has a mean diameter of 6 mm. If a couple is applied to the lever, determine the average shear stress in the pin between the pin and lever.



$$\oint \Sigma M_O = 0;$$
  $F(12) - 20(500) = 0;$   $F = 833.33 \text{ N}$ 

$$\tau_{\text{avg}} = \frac{V}{A} = \frac{833.33}{\frac{\pi}{4}(\frac{6}{1000})^2} = 29.5 \text{ MPa}$$
 Ans



\*1–40. The cinder block has the dimensions shown. If the material fails when the average normal stress reaches 120 psi, determine the largest centrally applied vertical load  ${\bf P}$  it can support.

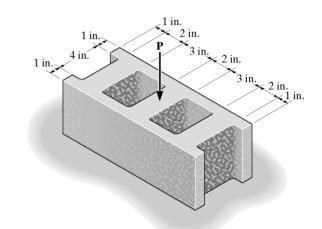
Cross section Area:

$$A = 6(14) - 2[4(1) + 3(4)] = 52 \text{ in}^2$$

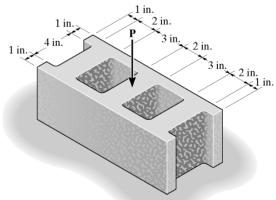
Average Normal Stress:

$$\sigma_{\text{allow}} = \frac{P_{\text{allow}}}{A}; \quad 120 = \frac{P_{\text{allow}}}{52}$$

$$P_{\text{allow}} = 6240 \text{ lb } = 6.24 \text{ kip}$$
 Ans



**1–41.** The cinder block has the dimensions shown. If it is subjected to a centrally applied force of  $P = 800 \, \text{lb}$ , determine the average normal stress in the material. Show the result acting on a differential volume element of the material.

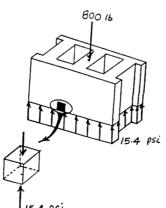


Cross section Area:

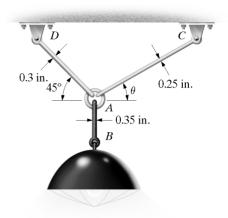
$$A = 14(6) - 4(8) = 52 \text{ in}^2$$

Average Normal Stress:

$$\sigma = \frac{P}{A} = \frac{800}{52} = 15.4 \text{ psi}$$
 And



**1–42.** The 50-lb lamp is supported by three steel rods connected by a ring at A. Determine which rod is subjected to the greater average normal stress and compute its value. Take  $\theta = 30^{\circ}$ . The diameter of each rod is given in the figure.



$$\begin{array}{c} \stackrel{+}{\to} \; \Sigma \; F_x = 0; \\ + \uparrow \; \Sigma \; F_y = 0; \end{array} \quad \begin{array}{c} F_{AC} \cos 30^{\circ} - F_{AD} \cos 45^{\circ} = 0 \\ F_{AC} \sin 30^{\circ} + F_{AD} \sin 45^{\circ} - 50 = 0 \end{array}$$

$$F_{AC} = 36.60 \text{ lb}, \qquad F_{AD} = 44.83 \text{ lb}$$

Rod AB:

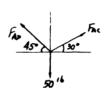
$$\sigma_{AB} = \frac{50}{\frac{\pi}{4} (0.35)^2} = 520 \text{ psi}$$

Rod AD

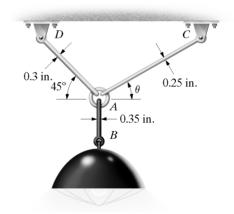
$$\sigma_{AD} = \frac{44.83}{\frac{\pi}{4}(0.3)^2} = 634 \text{ psi}$$

Rod AC:

$$\sigma_{AC} = \frac{36.60}{\frac{\pi}{4}(0.25)^2} = 746 \text{ psi}$$
 Ans



**1–43.** Solve Prob. 1–42 for  $\theta = 45^{\circ}$ .



$$\stackrel{\leftarrow}{\rightarrow} \Sigma F_x = 0; \qquad F_{AC} \cos 45^\circ - F_{AD} \cos 45^\circ = 0$$

$$+ \uparrow \Sigma F_y = 0; \qquad F_{AC} \sin 45^\circ + F_{AD} \sin 45^\circ - 50 = 0$$

$$F_{AC} = F_{AD} = 35.36 \text{ lb}$$

Rod AB:

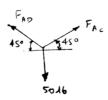
$$\sigma_{AB} = \frac{50}{\frac{\pi}{4}(0.35)^2} = 520 \text{ psi}$$

Rod AC

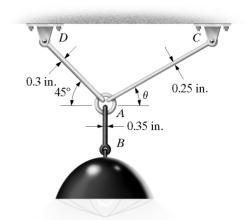
$$\sigma_{AC} = \frac{35.36}{\frac{\pi}{4}(0.25)^2} = 720 \text{ psi}$$
 Ans

Rod AD:

$$\sigma_{AD} = \frac{35.36}{\frac{\pi}{4}(0.3)^2} = 500 \text{ psi}$$



\*1–44. The 50-lb lamp is supported by three steel rods connected by a ring at A. Determine the angle of orientation  $\theta$  of AC such that the average normal stress in rod AC is twice the average normal stress in rod AD. What is the magnitude of stress in each rod? The diameter of each rod is given in the figure.



$$\sigma_{AD} = \frac{T_{AD}}{\frac{\pi}{4}(0.3)^2}; \qquad T_{AD} = (0.070686)\sigma_{AD}$$

$$\sigma_{AC} = 2\sigma_{AD} = \frac{T_{AC}}{\frac{\pi}{4}(0.25)^2}; \qquad T_{AC} = (0.098175)\sigma_{AD}$$

$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad -T_{AD} \cos 45^\circ + T_{AC} \cos \theta = 0 \qquad (1)$$

$$+\uparrow \Sigma F_v = 0;$$
  $T_{AC} \sin \theta + T_{AD} \sin 45^\circ - 50 = 0$  (2)

Thus

$$- (0.070686)\sigma_{AD}(\cos 45^{\circ}) + (0.098175)\sigma_{AD}(\cos \theta) = 0$$
  
$$\theta = 59.39^{\circ} = 59.4^{\circ}$$
 Ans

From Eq. (2):

$$(0.098175)\sigma_{AD} \sin 59.39^{\circ} + (0.070686)\sigma_{AD} \sin 45^{\circ} - 50 = 0$$
  
 $\sigma_{AD} = 371.8 \text{ psi} = 372 \text{ psi}$  Ans

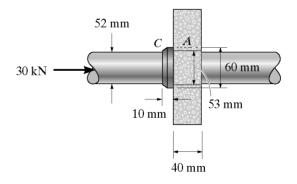
Hence,

$$\sigma_{AC} = 2(371.8) = 744 \text{ psi}$$

And.

$$\sigma_{AB} = \frac{T_{AB}}{\frac{\pi}{4}(0.35)^2} = \frac{50}{\frac{\pi}{4}(0.35)^2} = 520 \text{ psi}$$
 Ans

**1–45.** The shaft is subjected to the axial force of 30 kN. If the shaft passes through the 53-mm diameter hole in the fixed support A, determine the bearing stress acting on the collar C. Also, what is the average shear stress acting along the inside surface of the collar where it is fixed connected to the 52-mm diameter shaft?



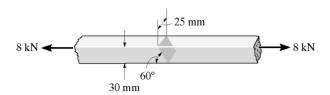
Bearing Stress:

$$\sigma_b = \frac{P}{A} = \frac{30(10^3)}{\frac{\pi}{4}(0.06^2 - 0.053^2)} = 48.3 \text{ MPa}$$
 Ans

Average Shear Stress:

$$\tau_{\text{avg}} = \frac{V}{A} = \frac{30(10^3)}{\pi (0.052)(0.01)} = 18.4 \text{ MPa}$$
 Ans

**1–46.** The two steel members are joined together using a  $60^{\circ}$  scarf weld. Determine the average normal and average shear stress resisted in the plane of the weld.



$$+ \sum F_x = 0;$$
  $N - 8 \sin 60^\circ = 0$ 

$$N = 6.928 \text{ kN}$$

$$\sum F_y = 0;$$
  $V - 8 \cos 60^\circ = 0$ 

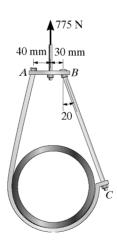
$$V = 4 \text{ kN}$$

$$A = (25) \left( \frac{30}{\sin 60^{\circ}} \right) = 866.03 \text{ mm}^2$$

$$\sigma = \frac{N}{A} = \frac{6.928 (10^3)}{0.8660 (10^{-3})} = 8 \text{ MPa}$$

$$\tau_{\text{avg}} = \frac{V}{A} = \frac{4(10^3)}{0.8660(10^{-3})} = 4.62 \text{ MPa}$$
 Ans

**1–47.** The J hanger is used to support the pipe such that the force on the vertical bolt is 775 N. Determine the average normal stress developed in the bolt BC if the bolt has a diameter of 8 mm. Assume A is a pin.



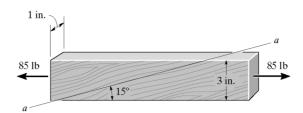
Support Reaction:

$$\Gamma + \Sigma M_A = 0;$$
 775(40)  $-F_{BC}\cos 20^{\circ}(70) = 0$   
 $F_{BC} = 471.28 \text{ N}$ 

Average Normal Stress:

$$\sigma_{BC} = \frac{F_{BC}}{A_{BC}} = \frac{471.28}{\frac{\pi}{4}(0.008^2)} = 9.38 \text{ MPa}$$

\*1–48. The board is subjected to a tensile force of 85 lb. Determine the average normal and average shear stress developed in the wood fibers that are oriented along section a–a at 15° with the axis of the board.



$$+ \sum F_x = 0;$$
  $V - 85 \cos 15^\circ = 0$ 

$$V = 82.10 \text{ lb}$$

$$+\Sigma F_y = 0; N - 85 \sin 15^\circ = 0$$

$$N = 22.00 \text{ lb}$$

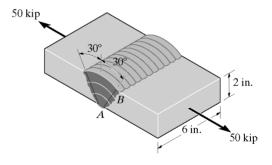
$$A = (1)(\frac{3}{\sin 15^{\circ}}) = 11.591 \text{ in}^2$$

$$\sigma = \frac{N}{A} = \frac{22.0}{11.591} = 1.90 \text{ psi}$$
 Ans

$$\tau_{avg} = \frac{V}{A} = \frac{82.10}{11.591} = 7.08 \text{ psi}$$
 Ans



**1–49.** The open square butt joint is used to transmit a force of 50 kip from one plate to the other. Determine the average normal and average shear stress components that this loading creates on the face of the weld, section AB.



# Equations of Equilibrium:

$$\Sigma + \Sigma F_y = 0;$$
  $N - 50 \cos 30^\circ = 0$   $N = 43.30 \text{ kip}$ 

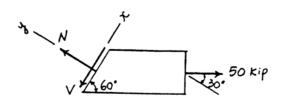
$$+\sum F_x = 0;$$
  $-V + 50 \sin 30^\circ = 0$   $V = 25.0 \text{ kip}$ 

# Average Normal and Shear Stress:

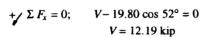
$$A' = \left(\frac{2}{\sin 60^{\circ}}\right)(6) = 13.86 \text{ in}^2$$

$$\sigma = \frac{N}{A'} = \frac{43.30}{13.86} = 3.125 \text{ ksi}$$
 Ans

$$\tau_{\text{avg}} = \frac{V}{A'} = \frac{25.0}{13.86} = 1.80 \text{ ksi}$$
 Ans



**1–50.** The specimen failed in a tension test at an angle of 52° when the axial load was 19.80 kip. If the diameter of the specimen is 0.5 in., determine the average normal and average shear stress acting on the area of the inclined failure plane. Also, what is the average normal stress acting on the *cross section* when failure occurs?





$$+$$
\(\Sigma \text{\$\Gamma\_{y} = 0\$}; \quad N - 19.80 \sin 52^{\text{o}} = 0\\ N = 15.603 \sin \text{\$\text{kip}}

Inclined plane:

$$\sigma' = \frac{P}{A};$$
  $\sigma' = \frac{15.603}{\frac{\pi(0.25)^2}{\sin 52^{\circ}}} = 62.6 \text{ ksi}$  Ans

$$\tau'_{avg} = \frac{V}{A};$$
  $\tau'_{avg} = \frac{12.19}{\frac{\pi(0.25)^2}{\sin 52^\circ}} = 48.9 \text{ ksi}$  Ans

Cross section:

$$\sigma = \frac{P}{A}$$
;  $\sigma = \frac{19.80}{\pi (0.25)^2} = 101 \text{ ksi}$  Ans

$$\tau_{a \vee g} = \frac{V}{A}; \qquad \tau_{a \vee g} = 0$$
Ans

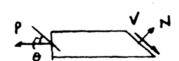
**1–51.** A tension specimen having a cross-sectional area A is subjected to an axial force  $\mathbf{P}$ . Determine the maximum average shear stress in the specimen and indicate the orientation  $\theta$  of a section on which it occurs.



$$\Delta \Sigma F_{v} = 0;$$
  $V - P \cos \theta = 0;$   $V = P \cos \theta$ 

$$\tau = \frac{P\cos\theta}{A/\sin\theta} = \frac{P\cos\theta\sin\theta}{A} = \frac{P\sin 2\theta}{2A}$$

$$\frac{d\tau}{d\theta} = \frac{P\cos 2\theta}{A} = 0$$



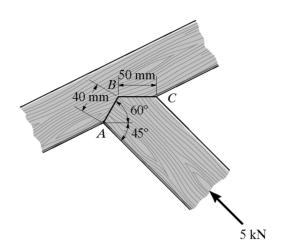
$$\cos 2\theta = 0$$

$$2\theta = 90^{\circ}$$

$$\theta = 45^{\circ}$$
 Ans

$$\tau_{\text{max}} = \frac{P}{2A} \sin 90^{\circ} = \frac{P}{2A} \qquad \text{Ans}$$

\*1–52. The joint is subjected to the axial member force of 5 kN. Determine the average normal stress acting on sections *AB* and *BC*. Assume the member is smooth and is 50-mm thick.



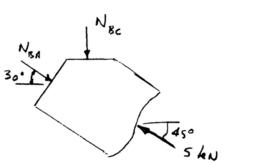
$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad N_{BA} \cos 30^\circ - 5 \cos 45^\circ = 0$$

$$N_{BA} = 4.082 \text{ kN}$$

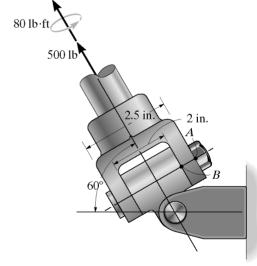
+ 
$$\uparrow \Sigma F_y = 0$$
;  $-N_{BC} - 4.082 \sin 30^\circ + 5 \sin 45^\circ = 0$   
 $N_{BC} = 1.494 \text{ kN}$ 

$$\sigma_{BA} = \frac{N_{BA}}{A_{BA}} = \frac{4.082(10^3)}{(0.04)(0.05)} = 2.04 \text{ MPa}$$
 Ans

$$\sigma_{BC} = \frac{N_{BC}}{A_{BC}} = \frac{1.494(10^3)}{(0.05)(0.05)} = 0.598 \text{ MPa}$$
 Ans



**1–53.** The yoke is subjected to the force and couple moment. Determine the average shear stress in the bolt acting on the cross sections through A and B. The bolt has a diameter of 0.25 in. *Hint:* The couple moment is resisted by a set of couple forces developed in the shank of the bolt.



At A force on bolt shank is zero, then  $\tau_A = 0$ 

Ans

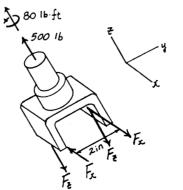
Equations of Equilibrium: Force on bolt shank at B.

$$\Sigma F_z = 0;$$
  $500 - 2F_z = 0$   $F_z = 250 \text{ lb}$ 

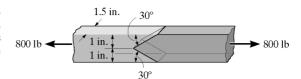
$$\Sigma M_z = 0;$$
 80 lb·ft  $\left(\frac{12 \text{ in}}{\text{ft}}\right) - F_x (2 \text{ in.}) = 0$   
 $F_x = 480 \text{ lb}$ 

Average Shear Stress: The bolt shank subjected to a shear force of  $V_B = F_B = \sqrt{250^2 + 480^2} = 541.2$  lb.

$$(\tau_B)_{avg} = \frac{541.2}{\frac{\pi}{4}(0.25)^2} = 11.0 \text{ ksi}$$
 Ans



**1–54.** The two members used in the construction of an aircraft fuselage are joined together using a 30° fish-mouth weld. Determine the average normal and average shear stress on the plane of each weld. Assume each inclined plane supports a horizontal force of 400 lb.



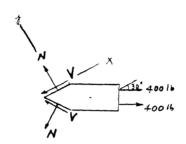
$$N - 400 \sin 30^{\circ} = 0;$$
  $N = 200 \text{ lb}$ 

$$400\cos 30^{\circ} - V = 0;$$
  $V = 346.41 \text{ lb}$ 

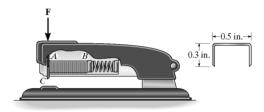
$$A' = \frac{1.5(1)}{\sin 30^{\circ}} = 3 \text{ in}^2$$

$$\sigma = \frac{N}{A'} = \frac{200}{3} = 66.7 \text{ psi}$$
 Ans

$$\tau = \frac{V}{A'} = \frac{346.41}{3} = 115 \text{ psi}$$
 Ans



**1–55.** The row of staples AB contained in the stapler is glued together so that the maximum shear stress the glue can withstand is  $\tau_{\rm max}=12$  psi. Determine the minimum force **F** that must be placed on the plunger in order to shear off a staple from its row and allow it to exit undeformed through the groove at C. The outer dimensions of the staple are shown in the figure. It has a thickness of 0.05 in. Assume all the other parts are rigid and neglect friction.



Average Shear Stress:

$$A = 0.5(0.3) - 0.4(0.25) = 0.05 \text{ in}^2$$

$$\tau_{\text{max}} = \frac{V}{A} \; ; \qquad 12 = \frac{V}{0.05}$$

$$F_{\min} = V = 0.60 \text{ lb}$$
 Ans

\*1–56. Rods AB and BC have diameters of 4 mm and 6 mm, respectively. If the load of 8 kN is applied to the ring at B, determine the average normal stress in each rod if  $\theta = 60^{\circ}$ .

$$+ \uparrow \Sigma F_{y} = 0;$$
  $T_{BC} \sin 60^{\circ} - 8 = 0$ 

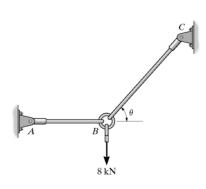
$$T_{BC} = 9.2376 \text{ kN}$$

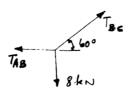
$$\rightarrow \Sigma F_x = 0;$$
 9.2376 cos 60° -  $T_{AB} = 0$ 

$$T_{AB} = 4.6188 \text{ kN}$$

$$\sigma_{AB} = \frac{T_{AB}}{A_{AB}} = \frac{4.6188(10^3)}{\frac{\pi}{4}(0.004)^2} = 368 \text{ MPa}$$
 Ans

$$\sigma_{BC} = \frac{T_{BC}}{A_{BC}} = \frac{9.2376(10^3)}{\frac{\pi}{4}(0.006)^2} = 327 \text{ MPa}$$
 Ans





1-57. Rods AB and BC have diameters of 4 mm and 6 mm. respectively. If the vertical load of 8 kN is applied to the ring at B, determine the angle  $\theta$  of rod BC so that the average normal stress in each rod is equivalent. What is this stress?

$$F_{AB} = \sigma A_{AB} = \sigma(\pi)(0.002)^2$$
  
 $F_{BC} = \sigma A_{BC} = \sigma(\pi)(0.003)^2$ 

$$\xrightarrow{+} \Sigma F_x = 0; \qquad \sigma(\pi)(0.003^2)\cos\theta - \sigma\pi(0.002^2) = 0 \qquad (1) 
+ \uparrow \Sigma F_y = 0; \qquad \sigma\pi(0.003^2)\sin\theta - 8(10^3) = 0 \qquad (2)$$

From Eq. (1):

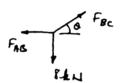
$$\cos \theta = (\frac{0.002}{0.003})^2$$

$$\theta = 63.6^{\circ}$$

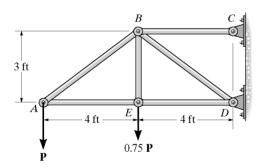
Ans

From Eq. (2):

$$\sigma = \frac{8(10^3)}{\pi (0.003)^2 \sin 63.6^\circ} = 316 \text{ MPa}$$
 Ans

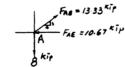


1-58. The bars of the truss each have a cross-sectional area of 1.25 in<sup>2</sup>. Determine the average normal stress in each member due to the loading P = 8 kip. State whether the stress is tensile or compressive.



$$\sigma_{AB} = \frac{F_{AB}}{A_{AB}} = \frac{13.33}{1.25} = 10.7 \text{ ksi}$$
 (T) Ans

$$\sigma_{AE} = \frac{F_{AE}}{A_{AE}} = \frac{1.25}{1.25} = 8.53 \text{ ksi}$$
 (C)



Joint E:

$$\sigma_{ED} = \frac{F_{ED}}{A_{ED}} = \frac{10.67}{1.25} = 8.53 \text{ ksi}$$
 (C) Ans

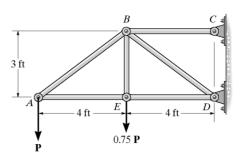
$$\sigma_{EB} = \frac{F_{EB}}{A_{CD}} = \frac{6.0}{1.25} = 4.80 \text{ ksi}$$
 (T) An

$$\sigma_{BC} = \frac{F_{BC}}{A_{BC}} = \frac{29.33}{1.25} = 23.5 \text{ ksi} \qquad \text{(T)} \qquad \text{Ans}$$

$$\sigma_{BD} = \frac{F_{BD}}{A_{BD}} = \frac{23.33}{1.25} = 18.7 \text{ ksi} \qquad \text{(C)} \qquad \text{Ans}$$

$$\sigma_{BD} = \frac{F_{BD}}{A_{BD}} = \frac{23.33}{1.25} = 18.7 \text{ ksi}$$
 (C) Ans

1-59. The bars of the truss each have a cross-sectional area of 1.25 in<sup>2</sup>. If the maximum average normal stress in any bar is not to exceed 20 ksi, determine the maximum magnitude P of the loads that can be applied to the truss.



Joint A:

$$+\uparrow \Sigma F_y = 0;$$
  $-P + (\frac{3}{5})F_{AB} = 0$ 

$$F_{AB} = (1.667)P$$

$$rightarrow \Sigma F_{x} = 0; -F_{AE} + (1.667)P(\frac{4}{5}) = 0$$

$$F_{AE} = (1.333)P$$

Joint E:

$$+ \uparrow \Sigma F_{r} = 0; \qquad F_{EB} - (0.75)P = 0$$

$$F_{EB} = (0.75)P$$

$$\stackrel{\bullet}{\rightarrow} \Sigma F_z = 0; \qquad (1.333)P - F_{ED} = 0$$

$$F_{ED} = (1.333)P$$

Joint B:

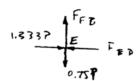
+ 
$$\uparrow \Sigma F_y = 0$$
;  $(\frac{3}{5})F_{BD} - (0.75)P - (1.667)P(\frac{3}{5}) = 0$ 

$$F_{BD} = (2.9167)P$$

$$\stackrel{+}{\to} \Sigma F_x = 0; \qquad F_{BC} - (2.9167)P(\frac{4}{5}) - (1.667)P(\frac{4}{5}) = 0$$

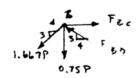
$$F_{BC} = (3.67)P$$





The highest stressed member is BC:

$$\sigma_{BC} = \frac{(3.67)P}{1.25} = 20$$

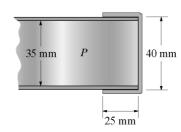


\*1-60. The plug is used to close the end of the cylindrical tube that is subjected to an internal pressure of  $p = 650 \,\mathrm{Pa}$ . Determine the average shear stress which the glue exerts on the sides of the tube needed to hold the cap in place.

Average Shear Stress:

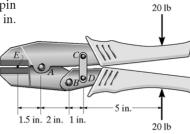
$$V = 650 \left[ \frac{\pi}{4} (0.035^2) \right] = 0.6254 \text{ N}$$

$$\tau_{\text{avg}} = \frac{V}{A} = \frac{0.6254}{\pi (0.04)(0.025)} = 199 \text{ Pa}$$
 Ans



1-61. The crimping tool is used to crimp the end of the wire E. If a force of 20 lb is applied to the handles, determine the average shear stress in the pin at A. The pin is subjected to double shear and has a diameter of 0.2 in.

Only a vertical force is exerted on the wire.



Support Reactions :

From FBD (a)

$$+ \sum M_D = 0;$$
  $20(5) - B_y(1) = 0$   $B_y = 100 \text{ Hz}$ 

$$\stackrel{+}{\rightarrow} \Sigma F_x = 0;$$

$$B_r = 0$$

From FBD (b)

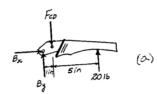
$$\xrightarrow{+} \Sigma F_x = 0;$$

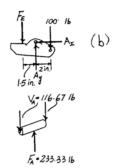
$$(+ \Sigma M_E = 0; A_y(1.5) - 100(3.5) = 0$$
  
 $A_y = 233.33 \text{ lb}$ 

Average Shear Stress: Pin A is subjected to double shear. Hence.

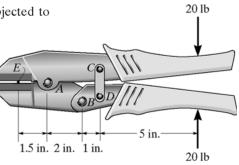
$$V_A = \frac{F_A}{2} = \frac{A_y}{2} = 116.67 \text{ lb}$$

$$(\tau_A)_{avg} = \frac{V_A}{A_A} = \frac{116.67}{\frac{\pi}{4}(0.2^2)}$$





**1–62.** Solve Prob. 1–61 for pin B. The pin is subjected to double shear and has a diameter of 0.2 in.



Support Reactions:

From FBD (a)

$$(+ \Sigma M_D = 0; 20(5) - B_y(1) = 0 B_y = 100 \text{ lb}$$

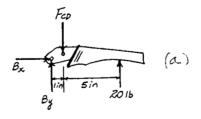
$$\stackrel{+}{\rightarrow} \Sigma F = 0$$

$$B = 0$$

Average Shear Stress: Pin B is subjected to double shear. Hence,

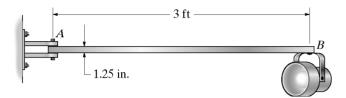
$$V_B = \frac{F_B}{2} = \frac{B_y}{2} = 50.0 \text{ lb}$$

$$(\tau_B)_{avg} = \frac{V_B}{A_B} = \frac{50.0}{\frac{\pi}{4}(0.2^2)}$$
  
= 1592 psi = 1.59 ksi Ans



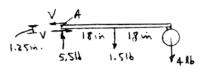


**1–63.** The railcar docklight is supported by the  $\frac{1}{8}$ -in-diameter pin at A. If the lamp weighs 4 lb, and the extension arm AB has a weight of 0.5 lb/ft, determine the average shear stress in the pin needed to support the lamp. *Hint:* The shear force in the pin is caused by the couple moment required for equilibrium at A.

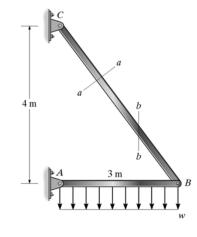


$$V(1.25) - 1.5(18) - 4(36) = 0$$
  
 $V = 136.8 \text{ lb}$ 

$$\tau_{\text{avg}} = \frac{V}{A} = \frac{136.8}{\frac{\pi}{4}(\frac{1}{8})^2} = 11.1 \text{ ksi}$$
 Ans



\*1–64. The two-member frame is subjected to the distributed loading shown. Determine the average normal stress and average shear stress acting at sections a–a and b–b. Member CB has a square cross section of 35 mm on each side. Take  $w = 8 \, \mathrm{kN/m}$ .



At setion a - a:

$$\sigma_{a-a} = \frac{15(10^3)}{(0.035)^2} = 12.2 \text{ MPa}$$
 Ans

 $\tau_{a-a}=0$  Ans

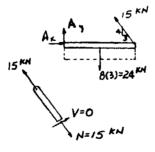
At section b-b:

$$\stackrel{+}{\to} \Sigma F_x = 0; \qquad N - 15(3/5) = 0; \qquad N = 9 \text{ kN}$$

$$+ \downarrow \Sigma F_y = 0;$$
  $V - 15(4/5) = 0;$   $V = 12 \text{ kN}$ 

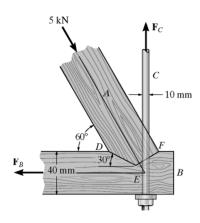
$$\sigma_{b-b} = \frac{9(10^3)}{(0.035)(0.035/0.6)} = 4.41 \text{ MPa}$$
 Ans

$$\tau_{b-b} = \frac{12(10^3)}{(0.035)(0.035/0.6)} = 5.88 \text{ MPa}$$
 Ans





**1–65.** Member A of the timber step joint for a truss is subjected to a compressive force of 5 kN. Determine the average normal stress acting in the hanger rod C which has a diameter of 10 mm and in member B which has a thickness of 30 mm.



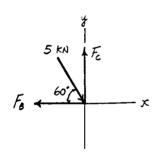
# Equations of Equilibrium:

$$\stackrel{+}{\rightarrow} \Sigma F_x = 0;$$
 5cos 60° -  $F_B = 0$   $F_B = 2.50 \text{ kN}$ 

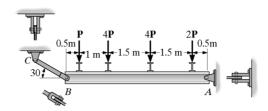
$$+ \uparrow \Sigma F_{\nu} = 0;$$
  $F_{c} - 5\sin 60^{\circ} = 0$   $F_{c} = 4.330 \text{ kN}$ 

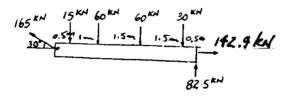
# Average Normal Stress:

$$\sigma_B = \frac{F_B}{A_B} = \frac{2.50(10^3)}{(0.04)(0.03)} = 2.08 \text{ MPa}$$
 Ans
$$\sigma_C = \frac{F_C}{A_C} = \frac{4.330(10^3)}{\frac{\pi}{4}(0.01^2)} = 55.1 \text{ MPa}$$
 Ans



**1–67.** The beam is supported by a pin at A and a short link BC. If P = 15 kN, determine the average shear stress developed in the pins at A, B, and C. All pins are in double shear as shown, and each has a diameter of 18 mm.





For pins B and C:

$$\tau_B = \tau_C = \frac{V}{A} = \frac{82.5 (10^3)}{\frac{\pi}{4} (\frac{18}{1000})^2} = 324 \text{ MPa}$$
 And



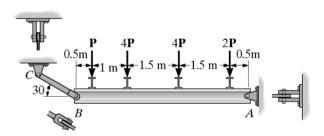
For pin A:

$$F_A = \sqrt{(82.5)^2 + (142.9)^2} = 165 \text{ kN}$$

$$\tau_A = \frac{V}{A} = \frac{82.5 (10^3)}{\frac{4}{3} (\frac{11}{1000})^2} = 324 \text{ MPa}$$
 Ans



\*1-68. The beam is supported by a pin at A and a short link BC. Determine the maximum magnitude P of the loads the beam will support if the average shear stress in each pin is not to exceed 80 MPa. All pins are in double shear as shown, and each has a diameter of 18 mm.



$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad A_x - 11P \cos 30^\circ = 0$$
$$A_x = 9.5263P$$

+ 
$$\uparrow \Sigma F_y = 0$$
;  $A_y - 11P + 11P \sin 30^\circ = 0$   
 $A_y = 5.5P$ 

$$F_A = \sqrt{(9.5263P)^2 + (5.5P)^2} = 11P$$

Require

$$\tau = \frac{V}{A};$$
  $80(10^6) = \frac{11P/2}{\frac{\pi}{4}(0.018)^2}$ 

$$P = 3.70 \text{ kN}$$
 Ans

**1–69.** The frame is subjected to the load of 200 lb. Determine the average shear stress in the bolt at A as a function of the bar angle  $\theta$ . Plot this function,  $0 \le \theta \le 90^{\circ}$ , and indicate the values of  $\theta$  for which this stress is a minimum. The bolt has a diameter of 0.25 in. and is subjected to single shear.

Support Reactions:  $(+\Sigma M_C = 0; F_{AB}\cos\theta (0.5) + F_{AB}\sin\theta (2) -200(3.5) = 0$ 

$$F_{AB}(0.5\cos\theta + 2\sin\theta) = 700$$

$$F_{AB} = \frac{700}{0.5\cos\theta + 2\sin\theta}$$

Average Shear Stress: Pin A is subjected to single shear. Hence,  $V_A = F_{AB}$ 

$$(\tau_A)_{avg} = \frac{V_A}{A_A} = \frac{\left(\frac{700}{0.5\cos\theta + 2\sin\theta}\right)}{\frac{\pi}{4}(0.25^2)}$$

$$= \left\{\frac{14260}{0.5\cos\theta + 2\sin\theta}\right\} \text{ psi}$$

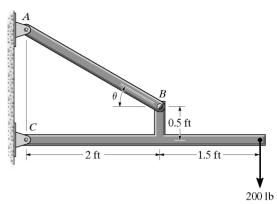
$$= \left\{\frac{14.3}{0.5\cos\theta + 2\sin\theta}\right\} \text{ ksi} \qquad \text{Ans}$$

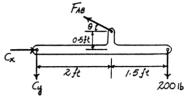
$$\frac{d\tau}{d\theta} = 0$$

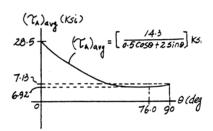
$$\frac{(0.5\cos\theta + 2\sin\theta)(0) - (-0.5\sin\theta + 2\cos\theta)(14260)}{(0.5\cos\theta + 2\sin\theta)^2} = 0$$

 $0.5\sin\theta - 2\cos\theta = 0$ 

$$\tan \theta = 4;$$
  $\theta_{min} = 76.0^{\circ}$  Ans







**1–70.** The jib crane is pinned at A and supports a chain hoist that can travel along the bottom flange of the beam, 1 ft  $\leq x \leq 12$  ft. If the hoist is rated to support a maximum of 1500 lb, determine the maximum average normal stress in the  $\frac{3}{4}$ -in.-diameter tie rod BC and the maximum average shear stress in the  $\frac{5}{8}$ -in.-diameter pin at B.

$$f + \sum M_A = 0;$$
  $T_{BC} \sin 30^{\circ} (10) - 1500(x) = 0$ 

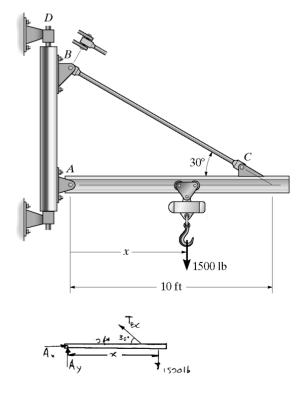
Maximum  $T_{BC}$  occurs when x = 12 ft

$$T_{BC} = 3600 \text{ lb}$$

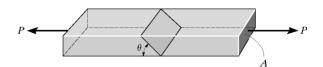
$$\sigma = \frac{P}{A} = \frac{3600}{\frac{\pi}{4}(0.75)^2} = 8.15 \text{ ksi}$$
 Ans

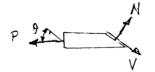
$$\tau = \frac{V}{A} = \frac{3600/2}{\frac{\pi}{4}(5/8)^2} = 5.87 \text{ ksi}$$
 Ans





**1–71.** The bar has a cross-sectional area A and is subjected to the axial load P. Determine the average normal and average shear stresses acting over the shaded section, which is oriented at  $\theta$  from the horizontal. Plot the variation of these stresses as a function of  $\theta$  ( $0 \le \theta \le 90^{\circ}$ ).





Equations of Equilibrium :

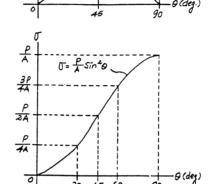
$$+\Sigma F_x = 0;$$
  $V - P \cos \theta = 0$   $V = P \cos \theta$   
 $V = P \cos \theta$ 

Average Normal Stress and Shear Stress: Area at  $\theta$  plane,  $A' = \frac{A}{\sin \theta}$ .

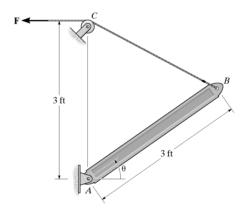
$$\sigma = \frac{N}{A'} = \frac{P \sin \theta}{\frac{A}{A + 1}} = \frac{P}{A} \sin^2 \theta$$

Ans

$$\tau_{avg} = \frac{V}{A'} = \frac{P \cos \theta}{\frac{A}{\sin \theta}}$$
$$= \frac{P}{A} \sin \theta \cos \theta = \frac{P}{2A} \sin 2\theta \qquad \text{Ans}$$



\*1–72. The boom has a uniform weight of 600 lb and is hoisted into position using the cable BC. If the cable has a diameter of 0.5 in., plot the average normal stress in the cable as a function of the boom position  $\theta$  for  $0^{\circ} \le \theta \le 90^{\circ}$ .



Support Reactions:

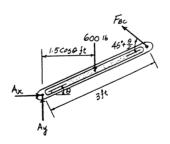
$$\left(+\Sigma M_A = 0; \qquad F_{BC} \sin\left(45^\circ + \frac{\theta}{2}\right)(3) - 600(1.5\cos\theta) = 0$$

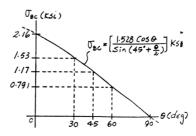
$$F_{BC} = \frac{300\cos\theta}{\sin\left(45^\circ + \frac{\theta}{2}\right)}$$

Average Normal Stress:

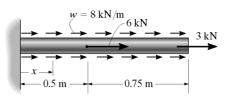
$$\sigma_{BC} = \frac{F_{BC}}{A_{BC}} = \frac{\frac{300\cos\theta}{\sin(45^{\circ} + \frac{\theta}{5})}}{\frac{\pi}{4}(0.5^{\circ})}$$

$$= \left\{ \frac{1.528\cos\theta}{\sin(45^{\circ} + \frac{\theta}{2})} \right\} \text{ ksi} \qquad \text{Ans}$$





**1–73.** The bar has a cross-sectional area of  $400 (10^{-6}) \text{ m}^2$ . If it is subjected to a uniform axial distributed loading along its length and to two concentrated loads as shown, determine the average normal stress in the bar as a function of x for  $0 < x \le 0.5 \text{ m}$ .

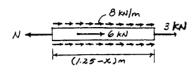


Equation of Equilibrium :

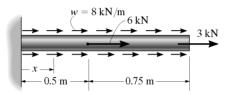
$$\stackrel{+}{\to} \Sigma F_x = 0; \qquad -N + 3 + 6 + 8(1.25 - x) = 0 
N = (19.0 - 8.00x) \text{ kN}$$

Average Normal Stress:

$$\sigma = \frac{N}{A} = \frac{(19.0 - 8.00x)(10^3)}{400(10^{-6})}$$
$$= (47.5 - 20.0x) \text{ MPa} \qquad \text{Ans}$$



**1–74.** The bar has a cross-sectional area of  $400(10^{-6})$  m<sup>2</sup>. If it is subjected to a uniform axial distributed loading along its length and to two concentrated loads as shown, determine the average normal stress in the bar as a function of x for 0.5 m  $< x \le 1.25$  m.



Equation of Equilibrium:

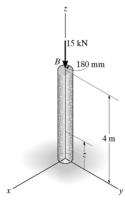
$$\stackrel{+}{\to} \Sigma F_x = 0;$$
  $-N+3+8(1.25-x)=0$   
 $N = (13.0-8.00x) \text{ kN}$ 

 $N \leftarrow \longrightarrow \longrightarrow \longrightarrow 3 \text{ kr}$ 

Average Normal Stress:

$$\sigma = \frac{N}{A} = \frac{(13.0 - 8.00x)(10^3)}{400(10^{-6})}$$
$$= (32.5 - 20.0x) \text{ MPa} \quad \text{Ans}$$

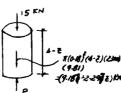
**1–75.** The column is made of concrete having a density of  $2.30 \,\mathrm{Mg/m^3}$ . At its top B it is subjected to an axial compressive force of 15 kN. Determine the average normal stress in the column as a function of the distance z measured from its base. *Note:* The result will be useful only for finding the average normal stress at a section removed from the ends of the column, because of localized deformation at the ends.



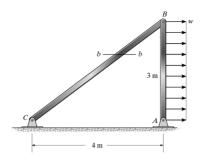
$$+ \uparrow \Sigma F_y = 0$$
  $P - 15 - 9.187 + 2.297 z = 0$ 

$$P = 24.187 - 2.297 z$$

$$\sigma = \frac{P}{A} = \frac{24.187 - 2.297 z}{\pi (0.18)^2} = (238 - 22.6 z) \text{ kPa} \quad \text{Ans}$$



\*1-76. The two-member frame is subjected to the distributed loading shown. Determine the largest intensity w of the uniform loading that can be applied to the frame without causing either the average normal stress or the average shear stress at section b-b to exceed  $\sigma = 15$  MPa and  $\tau = 16$  MPa, respectively. Member CB has a square cross-section of 30 mm on each side.



Support Reactions : FBD (a)  $(+ \sum M_A = 0;$   $\frac{4}{5}F_{BC}(3) - 3w(1.5) = 0$   $F_{BC} = 1.875w$ 

Equations of Equilibrium: For section b-b, FBD (b)

$$\stackrel{+}{\to} \Sigma F_x = 0;$$
  $\frac{4}{5}(1.875w) - V_{b-b} = 0$   $V_{b-b} = 1.50w$ 

$$+\uparrow \Sigma F_{y} = 0;$$
  $\frac{3}{5}(1.875w) - N_{b-b} = 0$   $N_{b-b} = 1.125w$   
Average Normal Stress And Shear Sress: The cross-sectional area of section  $b - b$ ,  $A' = \frac{5A}{3}$ ; where  $A = (0.03)(0.03) = 0.90(10^{-3}) \text{ m}^{2}$ .

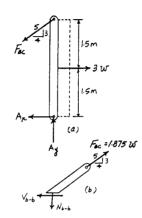
Then 
$$A' = \frac{5}{3}(0.90)(10^{-3}) = 1.50(10^{-3}) \text{ m}^2$$
.

Assume failure due to normal stress. 
$$(\sigma_{b-b})_{\text{Allow}} = \frac{N_{b-b}}{A'}; \qquad 15(10^6) = \frac{1.125w}{1.50(10^{-3})}$$

$$w = 20000 \text{ N/m} = 20.0 \text{ kN/m}$$

Assume failure due to shear stress. 
$$(\tau_{b-b})_{Allow} = \frac{V_{b-b}}{A'} \; ; \qquad 16(10^6) = \frac{1.50w}{1.50(10^{-3})}$$

$$w = 16000 \text{ N/m} = 16.0 \text{ kN/m} (Controls !)$$
 Ans



1-77. The pedestal supports a load P at its center. If the material has a mass density  $\rho$ , determine the radial dimension r as a function of z so that the average normal stress in the pedestal remains constant. The cross section is circular.

Require:  

$$\sigma = \frac{P + W_1}{A} = \frac{P + W_1 + dW}{A + dA}$$

$$P dA + W_1 dA = A dW$$

$$\frac{dW}{dA} = \frac{P + W_1}{A} = \sigma \tag{1}$$

$$dA = \pi (r + dr)^2 - \pi r^2 = 2\pi r dr$$

$$dW = \pi r^2(\rho g) dz$$

From Eq. (1),  

$$\frac{\pi r^2(\rho g) dz}{\sigma} = \sigma$$

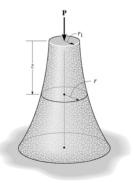
$$\frac{r\rho g\,dz}{2\,dr}=\sigma$$

$$\frac{\rho g}{2\sigma} \int_{0}^{z} dz = \int_{0}^{r} \frac{dr}{r}$$

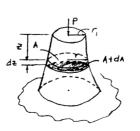
$$\frac{\rho g z}{2\sigma} = \ln \frac{r}{r_1}; \qquad r = r_1 e^{(\frac{\rho z}{2\sigma})z}$$
However,  $\sigma = \frac{P}{\pi r_1^2}$ 

However, 
$$\sigma = \frac{P}{\pi r^2}$$

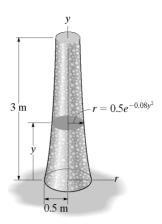
$$r = r_1 e^{\left(\frac{\pi r_1^2 \rho_4}{2P}\right)t}$$
 Ans







**1–78.** The radius of the pedestal is defined by  $r = (0.5e^{-0.08y^2})$  m, where y is given in meters. If the material has a density of 2.5 Mg/m<sup>3</sup>, determine the average normal stress at the support.



$$A = \pi (0.5)^2 = 0.7854 \text{ m}^2$$

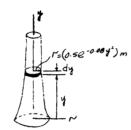
$$dV = \pi (r^2) dy = \pi (0.5)^2 (e^{-0.08y^2})^2$$

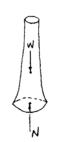
$$V = \int_0^3 \pi (0.5)^2 (e^{-0.08y^2})^2 dy = 0.7854 \int_0^3 (e^{-0.08y^2})^2 dy$$

$$W = \rho g V = (2500)(9.81)(0.7854) \int_0^3 (e^{-0.08y^2})^2 dy$$

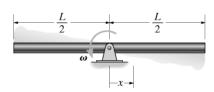
$$W = 19.262(10^3) \int_0^3 (e^{-0.08y^2})^2 dy = 38.849 \text{ kN}$$

$$\sigma = \frac{W}{A} = \frac{38.849}{0.7854} = 49.5 \,\mathrm{kPa} \qquad \text{Ans}$$





**1–79.** The uniform bar, having a cross-sectional area of A and mass per unit length of m, is pinned at its center. If it is rotating in the horizontal plane at a constant angular rate of  $\omega$ , determine the average normal stress in the bar as a function of x.

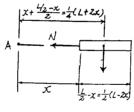


Equation of Motion:

$$\stackrel{+}{\leftarrow} \Sigma F_x = ma_N; \qquad N = m \left[ \frac{1}{2} (L - 2x) \right] \omega^2 \left[ \frac{1}{4} (L + 2x) \right] \\
= \frac{m\omega^2}{8} \left( L^2 - 4x^2 \right)$$

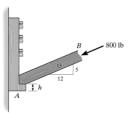
Average Normal Stress:

$$\sigma = \frac{N}{A} = \frac{m\omega^2}{8A} \left( L^2 - 4x^2 \right)$$
 Ans



$$a_1 = \omega^2 r = \omega^2 \left[ \frac{1}{4} (L + 2x) \right]$$

\*1-80. Member B is subjected to a compressive force of 800 lb. If A and B are both made of wood and are  $\frac{3}{8}$  in. thick, determine to the nearest  $\frac{1}{4}$  in. the smallest dimension h of the support so that the average shear stress does not exceed  $\tau_{\rm allow} = 300$  psi.

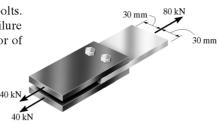


$$\tau_{\text{allow}} = 300 = \frac{307.7}{(\frac{3}{8})h}$$

$$h = 2.74 \text{ ir}$$

Use 
$$h = 2\frac{3}{4}$$
 in. Ans

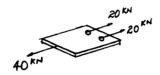
**1–81.** The joint is fastened together using two bolts. Determine the required diameter of the bolts if the failure shear stress for the bolts is  $\tau_{\text{fail}} = 350 \text{ MPa}$ . Use a factor of safety for shear of F.S. = 2.5.



$$\frac{350(10^6)}{2.5} = 140(10^6)$$

$$\tau_{\text{allow}} = 140(10^6) = \frac{20(10^3)}{\frac{\pi}{4}d^2}$$

$$d = 0.0135 \text{ m} = 13.5 \text{ mm}$$
 A



**1–82.** The rods AB and CD are made of steel having a failure tensile stress of  $\sigma_{\text{fail}} = 510 \text{ MPa}$ . Using a factor of safety of F.S. = 1.75 for tension, determine their smallest diameter so that they can support the load shown. The beam is assumed to be pin connected at A and C.

Support Reactions:

$$+\Sigma M_A = 0;$$
  $F_{CD}(10) - 5(7) - 6(4) - 4(2) = 0$   $F_{CD} = 6.70 \text{ kN}$ 

$$+\Sigma M_C = 0;$$
  $4(8) + 6(6) + 5(3) - F_{AB}(10) = 0$   
 $F_{AB} = 8.30 \text{ kN}$ 

Allowable Normal Stress: Design of rod sizes

For rod AB

$$\sigma_{\text{allow}} = \frac{\sigma_{\text{fail}}}{\text{F.S}} = \frac{F_{AB}}{A_{AB}};$$

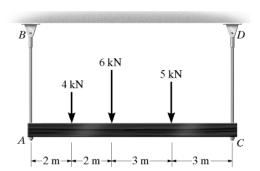
$$\frac{510(10^6)}{1.75} = \frac{8.30(10^3)}{\frac{\pi}{4}d_{AB}^2}$$

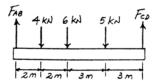
$$d_{AB} = 0.006022 \text{ m} = 6.02 \text{ mm} \quad \text{Ans}$$

For rod CD

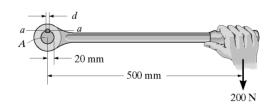
$$\sigma_{\text{allow}} = \frac{\sigma_{\text{fail}}}{\text{F.S}} = \frac{F_{CD}}{A_{CD}}; \qquad \frac{510(10^6)}{1.75} = \frac{6.70(10^3)}{\frac{\pi}{4} d_{CD}^2}$$

$$d_{CD} = 0.005410 \text{ m} = 5.41 \text{ mm}$$
 Ans





**1–83.** The lever is attached to the shaft A using a key that has a width d and length of 25 mm. If the shaft is fixed and a vertical force of 200 N is applied perpendicular to the handle, determine the dimension d if the allowable shear stress for the key is  $\tau_{\text{allow}} = 35 \text{ MPa}$ .



$$(+ \sum M_A = 0; F_a)$$

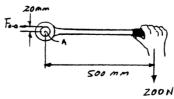
$$\xi + \sum M_A = 0;$$
  $F_{a-a}(20) - 200(500) = 0$ 

$$F_{a-a} = 5000 \text{ N}$$

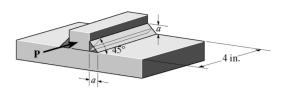
$$\tau_{\text{allow}} = \frac{F_{a \cdot a}}{A_{a \cdot a}}; \quad 35 (10^6) = \frac{5000}{d(0.025)}$$

$$d = 0.00571 \,\mathrm{m} = 5.71 \,\mathrm{mm}$$

Ans



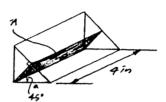
\*1-84. The fillet weld size a is determined by computing the average shear stress along the shaded plane, which has the smallest cross section. Determine the smallest size a of the two welds if the force applied to the plate is P=20 kip. The allowable shear stress for the weld material is  $\tau_{\rm allow}=14$  ksi.



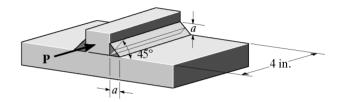
Shear plane 
$$A = a \sin 45^{\circ}(4) = 2.8284 a$$

$$\tau_{\text{allow}} = \frac{V}{A}; \qquad 14(10^{3}) = \frac{\frac{20(10^{3})}{2}}{2.8284 a}$$

$$a = 0.253 \text{ in.} \qquad \text{Ans}$$



**1–85.** The fillet weld size a = 0.25 in. If the joint is assumed to fail by shear on both sides of the block along the shaded plane, which is the smallest cross section, determine the largest force P that can be applied to the plate. The allowable shear stress for the weld material is  $\tau_{\rm allow} = 14$  ksi.

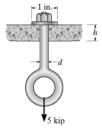


Area = 
$$(2)[(4)(0.707)(0.25)] = 1.414 \text{ in}^2$$

$$\tau_{\text{allow}} = \frac{V}{A}; \qquad 14 = \frac{P}{1.414}$$

 $P = 19.8 \,\mathrm{kip}$  Ans

**1–86.** The eye bolt is used to support the load of 5 kip. Determine its diameter d to the nearest  $\frac{1}{8}$  in. and the required thickness h to the nearest  $\frac{1}{8}$  in. of the support so that the washer will not penetrate or shear through it. The allowable normal stress for the bolt is  $\sigma_{\text{allow}} = 21$  ksi and the allowable shear stress for the supporting material is  $\tau_{\text{allow}} = 5$  ksi.



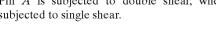
Allowable Normal Stress: Design of bolt size

$$\sigma_{\text{allow}} = \frac{P}{A_b}$$
;  $21.0(10^3) = \frac{5(10^3)}{\frac{5}{4}d^2}$   
 $d = 0.5506 \text{ in.}$   
 $Use d = \frac{5}{8} \text{ in.}$  Ans

Allowable Shear Stress: Design of support thickness

$$\tau_{\text{allow}} = \frac{V}{A}$$
;  $5(10^3) = \frac{5(10^3)}{\pi(1)(h)}$   
 $h = 0.3183 \text{ in.}$   
 $Use h = \frac{3}{8} \text{ in.}$  Ans

1-87. The frame is subjected to the load of 1.5 kip. Determine the required diameter of the pins at A and B if the allowable shear stress for the material is  $\tau_{\text{allow}} = 6 \text{ ksi.}$ Pin A is subjected to double shear, whereas pin B is subjected to single shear.





$$F_{BC} = 0;$$
  $F_{BC}(\sin 45^{\circ})(5) - 1.5(7) = 0$   
 $F_{BC} = 2.970 \text{ kip}$ 

From FBD (b),

$$\int_{C} + \sum M_A = 0;$$
  $D_y(10) - 1.5(7) = 0$   $D_y = 1.05 \text{ kip}$ 

$$\stackrel{+}{\leftarrow} \Sigma F_x = 0;$$
  $A_x - 1.5 = 0$   $A_x = 1.50 \text{ kip}$ 

$$+ \uparrow \Sigma F_{y} = 0;$$
 1.05  $-A_{y} = 0$   $A_{y} = 1.05 \text{ kip}$ 

Allowable Shear Stress: Design of pin sizes

## For pin A

Pin A is subjected to double shear and

$$F_A = \sqrt{1.50^2 + 1.05^2} = 1.831 \text{ kip.}$$

Therefore, 
$$V_A = \frac{F_A}{2} = 0.9155 \text{ kip}$$

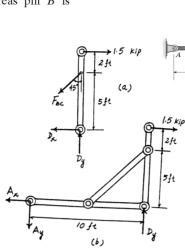
$$\tau_{\text{allow}} = \frac{V_A}{A_A}$$
;  $6 = \frac{0.9155}{\frac{\pi}{4}d_A^2}$ 

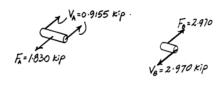
For pin B

Pin B is subjected to single shear. Therefore,

$$V_B = F_B = F_{BC} = 2.970 \text{ kip}$$

$$au_{\rm allow} = rac{V_B}{A_B} \; ; \qquad 6 = rac{2.970}{\frac{\pi}{4} d_B \, 2} \ d_B = 0.794 \; {
m in}. \qquad {
m An}$$





## \*1–88. The two steel wires AB and AC are used to support the load. If both wires have an allowable tensile stress of $\sigma_{\rm allow} = 200$ MPa, determine the required diameter of each wire if the applied load is P = 5 kN.

$$+\Sigma F_x = 0;$$
  $\frac{4}{5}F_{AC} - F_{AB}\sin 60^\circ = 0$  (1)

$$+ \uparrow \Sigma F_y = 0;$$
  $\frac{3}{5} F_{AC} + F_{AB} \cos 60^\circ - 5 = 0$  (2)

Solving Eqs. (1) and (2) yields:

$$F_{AB} = 4.3496 \text{ kN}; \qquad F_{AC} = 4.7086 \text{ kN}$$

Applying 
$$\sigma_{allow} = \frac{P}{A}$$

For wire AB,

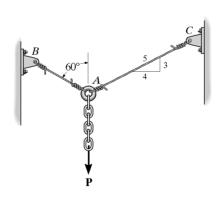
$$200(10^6) = \frac{4.3496(10^3)}{\frac{\pi}{4}(d_{AB})^2}$$

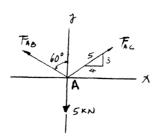
$$d_{AB} = 0.00526 \text{ m} = 5.26 \text{ mm}$$
 Ans

For wire AC,

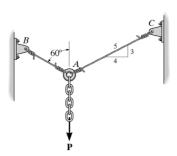
$$200(10^6) = \frac{4.7086(10^3)}{\frac{\pi}{6}(d_{AC})^2}$$

$$d_{AC} = 0.00548 \text{ m} = 5.48 \text{ mm}$$
 Ans





**1–89.** The two steel wires AB and AC are used to support the load. If both wires have an allowable tensile stress of  $\sigma_{\rm allow} = 180$  MPa, and wire AB has a diameter of 6 mm and AC has a diameter of 4 mm, determine the greatest force P that can be applied to the chain before one of the wires fails.



$$+\sum_{A} F_{x} = 0;$$
  $\frac{4}{5}F_{AC} - F_{AB} \sin 60^{\circ} = 0$  (1)

$$+ \uparrow \Sigma F_y = 0;$$
  $\frac{3}{5} F_{AC} + F_{AB} \cos 60^\circ - P = 0$  (2)

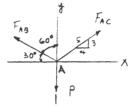
Assume failure of AB:

$$\sigma_{\text{allow}} = \frac{F_{AB}}{A_{AB}}; \quad 180(10^6) = \frac{F_{AB}}{\frac{\pi}{4}(0.006)^2}$$

$$F_{AB} = 5089.38 \text{ N} = 5.089 \text{ kN}$$

Solving Eqs.(1) and (2) yields:

 $F_{AC} = 5.509 \text{ kN}$ ; P = 5.85 kN



Assume failure of AC:

$$\sigma_{\text{allow}} = \frac{F_{AC}}{A_{AC}}; \quad 180(10^6) = \frac{F_{AC}}{\frac{\pi}{4}(0.004)^2}$$

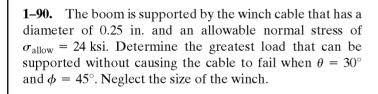
$$F_{AC} = 2261.94 \,\mathrm{N} = 2.262 \,\mathrm{kN}$$

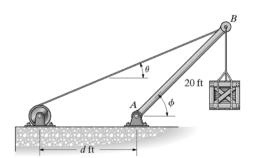
Solving Eqs. (1) and (2) yields:

 $F_{AB} = 2.089 \text{ kN}$ ; P = 2.40 kN

Choose the smallest value

$$P = 2.40 \text{ kN}$$
 Ans





$$\sigma = \frac{P}{A};$$
  $24(10^3) = \frac{T}{\frac{\pi}{4}(0.25)^2};$ 

$$T = 1178.10 \text{ lb}$$

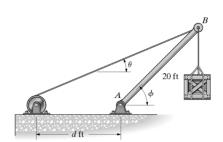
$$\rightarrow \Sigma F_x = 0;$$
 -1178.10 cos 30° +  $F_{AB}$  sin 45° = 0

$$+ \uparrow \Sigma F_v = 0;$$
  $-W + F_{AB} \cos 45^\circ - 1178.10 \sin 30^\circ = 0$ 

$$W = 431 \text{ lb}$$
 Ans

$$F_{AB} = 1442.9 \text{ lb}$$

**1–91.** The boom is supported by the winch cable that has an allowable normal stress of  $\sigma_{\rm allow} = 24$  ksi. If it is required that it be able to slowly lift 5000 lb, from  $\theta = 20^{\circ}$  to  $\theta = 50^{\circ}$ , determine the smallest diameter of the cable to the nearest  $\frac{1}{16}$  in. The boom AB has a length of 20 ft. Neglect the size of the winch. Set d = 12 ft.



Maximum tension in cable occurs when  $\theta = 20^{\circ}$ ,

$$\frac{\sin 20^{\circ}}{20} = \frac{\sin \psi}{12}$$

$$\psi = 11.842^{\circ}$$

$$\stackrel{+}{\rightarrow} \Sigma F_x = 0; \qquad -T \cos 20^\circ + F_{AB} \cos 31.842^\circ = 0$$

$$+ \uparrow \Sigma F_y = 0;$$
  $F_{AB} \sin 31.842^{\circ} - T \sin 20^{\circ} - 5000 = 0$ 

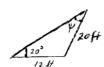
T = 20698.3 lb

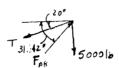
$$F_{AB} = 22896 \text{ lb}$$

$$\sigma = \frac{P}{A};$$
  $24(10^3) = \frac{20698.3}{\frac{\pi}{4}(d)^2}$ 

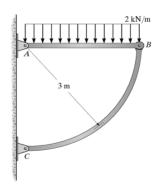
$$d = 1.048$$
 in.

Use 
$$d = 1\frac{1}{16}$$
 in. Ans





\*1–92. The frame is subjected to the distributed loading of 2 kN/m. Determine the required diameter of the pins at A and B if the allowable shear stress for the material is  $\tau_{\rm allow} = 100$  MPa. Both pins are subjected to double shear.



Support Reactions: Member BC is a two force member.

$$(+\Sigma M_A = 0; F_{BC} \sin 45^{\circ}(3) - 6(1.5) = 0$$
  
 $F_{BC} = 4.243 \text{ kN}$ 

$$+ \uparrow \Sigma F_y = 0;$$
  $A_y + 4.243 \sin 45^{\circ} - 6 = 0$   $A_y = 3.00 \text{ kN}$ 

$$\stackrel{\leftarrow}{\leftarrow} \Sigma F_x = 0; \qquad A_x - 4.243 \cos 45^\circ = 0$$

$$A_x = 3.00 \text{ kN}$$

Allowable Shear Stress: Pin A and pin B are subjected to double shear.

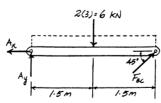
$$F_A = \sqrt{3.00^2 + 3.00^2} = 4.243 \text{ kN}$$
 and

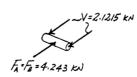
$$F_B = F_{BC} = 4.243 \text{ kN}.$$

Therefore,

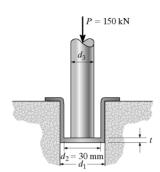
$$V_A = V_B = \frac{4.243}{2} = 2.1215 \text{ kN}$$

$$\tau_{\text{allow}} = \frac{V}{A}$$
;  $100(10^6) = \frac{2.1215(10^3)}{\frac{\pi}{4}d^2}$   
 $d = 0.005197 \text{ m} = 5.20 \text{ mm}$   
 $d_A = d_B = d = 5.20 \text{ mm}$  Ans





**1–93.** Determine the smallest dimensions of the circular shaft and circular end cap if the load it is required to support is  $P=150\,\mathrm{kN}$ . The allowable tensile stress, bearing stress, and shear stress is  $(\sigma_t)_{\mathrm{allow}}=175\,\mathrm{MPa}$ ,  $(\sigma_b)_{\mathrm{allow}}=275\,\mathrm{MPa}$ , and  $\tau_{\mathrm{allow}}=115\,\mathrm{MPa}$ .



Allowable Normal Stress: Design of end cap outer diameter

$$(\sigma_t)_{\text{allow}} = \frac{P}{A};$$
  $175(10^6) = \frac{150(10^3)}{\frac{\pi}{4}(d_1^2 - 0.03^2)}$   
  $d_1 = 0.04462 \text{ m} = 44.6 \text{ mm}$  Ans

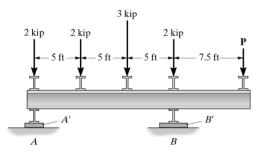
Allowable Bearing Stress: Design of circular shaft diameter

$$(\sigma_b)_{\text{allow}} = \frac{P}{A};$$
  $275(10^6) = \frac{150(10^3)}{\frac{g}{4}d_3^2}$   $d_3 = 0.02635 \text{ m} = 26.4 \text{ mm}$  Ans

Allowable Shear Stress: Design of end cap thickness

$$\tau_{\text{allow}} = \frac{V}{A}$$
; 115 (10<sup>6</sup>) =  $\frac{150(10^3)}{\pi (0.02635) t}$   
 $t = 0.01575 \text{ m} = 15.8 \text{ mm}$  Ans

**1–94.** If the allowable bearing stress for the material under the supports at A and B is  $(\sigma_b)_{\text{allow}} = 400$  psi, determine the size of *square* bearing plates A' and B' required to support the loading. Take P = 1.5 kip. Dimension the plates to the nearest  $\frac{1}{2}$  in. The reactions at the supports are vertical.

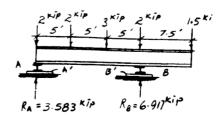


For Plate A:

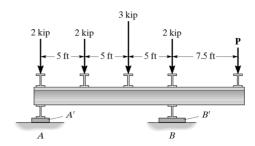
$$\sigma_{\text{allow}} = 400 = \frac{3.583 (10^3)}{a_A^2}$$
 $a_{A'} = 2.99 \text{ in.}$ 
Use a 3 in. x 3 in. plate Ans

For Plate B:

$$\sigma_{\text{allow}} = 400 = \frac{6.917 (10^3)}{a_B^2}$$
 $a_B = 4.16 \text{ in.}$ 
Use a  $4\frac{1}{2}$  in. x  $4\frac{1}{2}$  in. plate Ans



**1–95.** If the allowable bearing stress for the material under the supports at A and B is  $(\sigma_b)_{\text{allow}} = 400$  psi, determine the maximum load **P** that can be applied to the beam. The bearing plates A' and B' have square cross sections of 2 in.  $\times$  2 in. and 4 in.  $\times$  4 in., respectively.



$$\{+ \Sigma M_A = 0; \quad B_y(15) - 2(5) - 3(10) - 2(15) - P(225) = 0\}$$

$$B_{\rm v} = 1.5P + 4.667$$

$$+ \uparrow \Sigma F_y = 0;$$
  $A_y + 1.5P + 4.667 - 9 - P = 0$ 

$$A_{y} = 4.333 - 0.5P$$

**Α**ι **A** :

$$0.400 = \frac{4.333 - 0.5P}{2(2)}$$

$$P = 5.47 \text{ kip}$$

s' s' s' zs'

At R

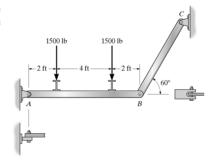
$$0.400 = \frac{1.5P + 4.667}{4(4)}$$

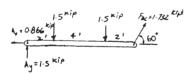
$$P = 1.16 \text{ kip}$$

Thus,

$$P_{\rm allow} = 1.16 \, {\rm kip}$$
 Ans

\*1–96. Determine the required cross-sectional area of member BC and the diameter of the pins at A and B if the allowable normal stress is  $\sigma_{\rm allow}=3$  ksi and the allowable shear stress is  $\tau_{\rm allow}=4$  ksi.





Member BC:

$$\sigma_{\text{allow}} = 3 (10^3) = \frac{1.732 (10^3)}{400}$$

$$A_{BC} = 0.577 \text{ in}^2$$
 Ans

Pin A

$$F_A = \sqrt{(0.866)^2 + (1.5)^2} = 1.732 \text{ kip}$$

$$\tau_{\text{allow}} = 4(10^3) = \frac{1.732(10^3)}{\frac{\pi}{3}(d_A)^2}$$

$$d_{\rm A} = 0.743 \, {\rm in.}$$
 Ans

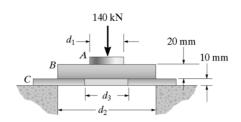
.844 hip 1.732 hip

Pin B:

$$\tau_{\text{allow}} = 4(10^3) = \frac{0.866(10^3)}{\frac{\pi}{5}(d_B)^2}$$

$$d_0 = 0.525 \text{ in.}$$
 An

**1–97.** The assembly consists of three disks A, B, and C that are used to support the load of 140 kN. Determine the smallest diameter  $d_1$  of the top disk, the diameter  $d_2$  within the support space, and the diameter  $d_3$  of the hole in the bottom disk. The allowable bearing stress for the material is  $(\sigma_{\rm allow})_b = 350$  MPa and allowable shear stress is  $\tau_{\rm allow} = 125$  MPa.



## Solution

Allowable Shear Stress: Assume shear failure for disk C.

$$\tau_{\text{allow}} = \frac{V}{A}$$
;  $125(10^6) = \frac{140(10^3)}{\pi d_2(0.01)}$   
  $d_2 = 0.03565 \text{ m} = 35.7 \text{ mm}$  Ans

Allowable Bearing Stress: Assume bearing failure for disk C.

$$(\sigma_b)_{\text{allow}} = \frac{P}{A};$$
 350  $(10^6) = \frac{140(10^3)}{\frac{\pi}{4}(0.03565^2 - d_3^2)}$   
 $d_3 = 0.02760 \text{ m} = 27.6 \text{ mm}$  Ans

Allowable Bearing Stress: Assume bearing failure for disk B.

$$(\sigma_b)_{\text{allow}} = \frac{P}{A};$$
 350  $(10^6) = \frac{140(10^3)}{\frac{\kappa}{4}d_1^2}$   
 $d_1 = 0.02257 \text{ m} = 22.6 \text{ mm}$ 

Since  $d_3 = 27.6 \text{ mm} > d_1 = 22.6 \text{ mm}$ , disk B might fail due to shear.

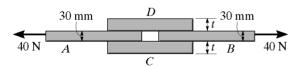
$$\tau = \frac{V}{A} = \frac{140(10^3)}{\pi(0.02257)(0.02)} = 98.7 \text{ MPa} < \tau_{\text{allow}} = 125 \text{ MPa} (O.K!)$$

Therefore,

$$d_1 = 22.6 \text{ mm}$$

Ans

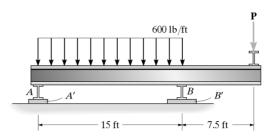
**1–98.** Strips A and B are to be glued together using the two strips C and D. Determine the required thickness t of C and D so that all strips will fail simultaneously. The width of strips A and B is 1.5 times that of strips C and D.



Average Normal Stress: Requires,

$$\sigma_A = \sigma_B = \sigma_C;$$
  $\frac{40}{(0.03)(1.5w)} = \frac{20}{wt}$   
 $t = 0.0225 \text{ m} = 22.5 \text{ mm}$  Ans

**1–99.** If the allowable bearing stress for the material under the supports at A and B is  $(\sigma_b)_{\text{allow}} = 400 \text{ psi}$ , determine the size of *square* bearing plates A' and B' required to support the load. Dimension the plates to the nearest  $\frac{1}{2}$  in. The reactions at the supports are vertical. Take P = 1500 lb.



Support Reactions:

$$F_B = 6.75 \text{ kip}$$
  $F_B = 6.75 \text{ kip}$ 

$$+ \uparrow \Sigma F_y = 0;$$
  $F_A + 6.75 - 9.00 - 1.50 = 0$   
 $F_A = 3.75 \text{ kip}$ 

Allowable Bearing Stress: Design of bearing plates

For plate A'

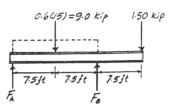
$$(\sigma_b)_{\rm allow} = \frac{F_A}{A_{A'}}$$
;  $400 = \frac{3.75(10^3)}{L_{A'}^2}$   $L_{A'} = 3.06$  in.

Use 
$$3\frac{1}{2}$$
 in.  $\times$   $3\frac{1}{2}$  in. plate **Ans**

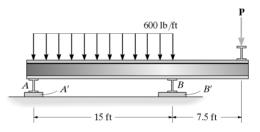
for plate B'

$$(\sigma_b)_{\text{allow}} = \frac{F_B}{A_{B'}}; \quad 400 = \frac{6.75(10^3)}{L_{B'}^2} \quad L_{B'} = 4.11 \text{ in.}$$

Use 
$$4\frac{1}{2}$$
 in.  $\times$   $4\frac{1}{2}$ in. plate Ans



\*1–100. If the allowable bearing stress for the material under the support at A and B is  $(\sigma_b)_{\rm allow} = 400 \, \rm psi$ , determine the maximum load **P** that can be applied to the beam. The bearing plates A' and B' have square cross sections of 2 in.  $\times$  2 in. and 4 in.  $\times$  4 in., respectively.



Support Reactions:

$$\oint_{A} + \Sigma M_{A} = 0; \qquad F_{B}(15) - 9.00(7.5) - P(22.5) = 0$$

$$15F_{B} - 22.5P = 67.5$$
 [1]

$$\int_{A} + \Sigma M_{B} = 0; \qquad 9.00(7.5) - P(7.5) - F_{A}(15) = 0$$

$$15F_{A} + 7.5P = 67.5 \qquad [2]$$

Allowable Bearing Stress: Assume failure of material occurs under plate A'

$$(\sigma_b)_{\text{allow}} = \frac{F_A}{A_{A'}}$$
;  $400 = \frac{F_A}{2(2)}$   
 $F_A = 1600 \text{ lb} = 1.60 \text{ kip}$ 

From Eq. [2]

$$P = 5.80 \text{ kip}$$

Assume failure of material occurs under B'

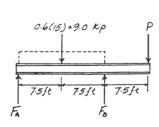
$$(\sigma_b)_{\text{allow}} = \frac{F_B}{A_{B'}};$$
  $400 = \frac{F_B}{4(4)}$   
 $F_B = 6400 \text{ lb} = 6.40 \text{ kip}$ 

From Eq. [1]

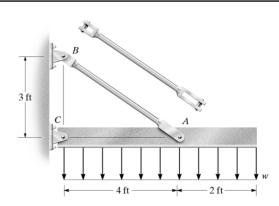
$$P = 1.27 \text{ kip}$$

Choose the *smallest* value P = 1.27 kip

Ans



**1–101.** The hanger assembly is used to support a distributed loading of w = 0.8 kip/ft. Determine the average shear stress in the 0.40-in.-diameter bolt at A and the average tensile stress in rod AB, which has a diameter of 0.5 in. If the yield shear stress for the bolt is  $\tau_y = 25$  ksi, and the yield tensile stress for the rod is  $\sigma_y = 38$  ksi, determine the factor of safety with respect to yielding in each case.



For holt 4 :

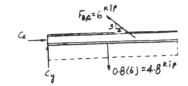
$$\tau = \frac{V}{A} = \frac{3}{\frac{\pi}{4}(0.4^2)} = 23.9 \text{ ksi}$$
 Ans  
F. S.  $= \frac{\tau_y}{\tau} = \frac{25}{23.9} = 1.05$  Ans

For rod AB:

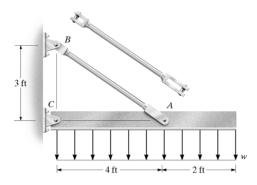
r rod 
$$AB$$
:  

$$\sigma = \frac{P}{A} = \frac{6}{\frac{\pi}{4}(0.5^2)} = 30.6 \text{ ksi} \quad \text{Ans}$$
F. S. =  $\frac{\sigma_y}{\sigma} = \frac{38}{30.6} = 1.24 \quad \text{Ans}$ 





**1–102.** Determine the intensity w of the maximum distributed load that can be supported by the hanger assembly so that an allowable shear stress of  $\tau_{\rm allow} = 13.5$  ksi is not exceeded in the 0.40-in.-diameter bolts at A and B, and an allowable tensile stress of  $\sigma_{\rm allow} = 22$  ksi is not exceeded in the 0.5-in.-diameter rod AB.

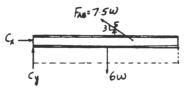


Assume failure of pin A or B:

$$\tau_{\text{allow}} = 13.5 = \frac{3.75w}{\frac{\pi}{4}(0.4^2)}$$

w = 0.452 kip/ft (controls) Ans

7.5W 3.75W



Assuming failure of rod AB:

$$\sigma_{\text{allow}} = 22 = \frac{7.5w}{\frac{\pi}{4}(0.5^2)}$$

 $w = 0.576 \, \text{kip/ft}$ 

**1–103.** The bar is supported by the pin. If the allowable tensile stress for the bar is  $(\sigma_t)_{\rm allow}=21\,{\rm ksi}$ , and the allowable shear stress for the pin is  $\tau_{\rm allow}=12\,{\rm ksi}$ , determine the diameter of the pin for which the load P will be a maximum. What is this maximum load? Assume the hole in the bar has the same diameter d as the pin. Take  $t=\frac{1}{4}$  in. and w=2 in.



Allowable Normal Stress: The effective cross-sectional area A' for the bar must be considered here by taking into account the reduction in cross sectional area introduced by the hole. Here  $A' = (2-d)(\frac{1}{4})$ .

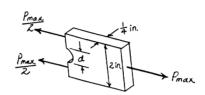
$$(\sigma_t)_{\text{allow}} = \frac{P}{A'}; \quad 21(10^3) = \frac{P_{\text{max}}}{(2-d)(\frac{1}{4})}$$
 [1]

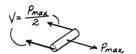
Allowable Shear Stress: The pin is subjected to double shear and therefore,  $V = \frac{P_{\text{max}}}{2}$ 

$$\tau_{\text{allow}} = \frac{V}{A}; \quad 12(10^3) = \frac{P_{\text{max}}/2}{\frac{\pi}{d}d^2}$$
 [2]

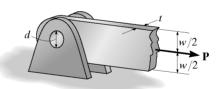
Solving Eq.[1] and[2] yields:

$$d = 0.620$$
 in. Ans  $P_{\text{max}} = 7.25$  kip Ans



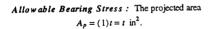


\*1–104. The bar is connected to the support using a pin having a diameter of d=1 in. If the allowable tensile stress for the bar is  $(\sigma_t)_{\text{allow}} = 20$  ksi, and the allowable bearing stress between the pin and the bar is  $(\sigma_b)_{\text{allow}} = 30$  ksi, determine the dimensions w and t such that the gross area of the cross section is wt = 2 in and the load P is a maximum. What is this maximum load? Assume the hole in the bar has the same diameter as the pin.

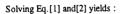


Allowable Normal Stress: The effective cross-sectional area A' for the bar must be considered here by taking into account the reduction in cross-sectional area introduced by the hole. Here  $A' = (w-1)t = wt - t = (2-t) \ln^2 w$  where  $wt = 2 \ln^2$ .

$$(\sigma_t)_{\text{allow}} = \frac{P}{A'}; \quad 20(10^3) = \frac{P_{\text{max}}}{2 - t}$$
 [1]



$$(\sigma_b)_{\text{allow}} = \frac{P}{A_P}; \quad 30(10^3) = \frac{P_{\text{max}}}{t}$$
 [2]

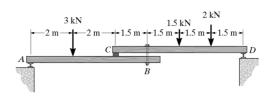


$$t = 0.800$$
 in. Ans  $P_{max} = 24.0$  kip Ans

And 
$$w = 2.50$$
 in. Ans



1-105. The compound wooden beam is connected together by a bolt at B. Assuming that the connections at A, B, C, and D exert only vertical forces on the beam, determine the required diameter of the bolt at B and the required outer diameter of its washers if the allowable tensile stress for the bolt is  $(\sigma_t)_{\text{allow}} = 150 \text{ MPa}$  and the allowable bearing stress for the wood is  $(\sigma_b)_{\text{allow}} = 28 \text{ MPa}$ . Assume that the hole in the washers has the same diameter as the bolt.



From FBD (a):

$$(+ \Sigma M_D = 0;$$
  $F_B(4.5) + 1.5(3) + 2(1.5) - F_C(6) = 0$   
 $4.5 F_B - 6 F_C = -7.5$  (1

From FBD (b):

$$\text{ (+ } \Sigma M_A = 0; \qquad F_B(5.5) - F_C(4) - 3(2) = 0$$

$$5.5 F_B - 4 F_C = 6$$
(2)

Solving Eqs. (1) and (2) yields

$$F_B = 4.40 \text{ kN}; \quad F_C = 4.55 \text{ kN}$$

For holt

$$\sigma_{\text{allow}} = 150 (10^6) = \frac{4.40(10^3)}{\frac{\pi}{4}(d_g)^2}$$
 $d_g = 0.00611 \text{ m}$ 
 $= 6.11 \text{ mm}$  Ans

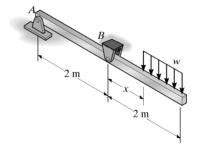


$$\sigma_{\text{allow}} = 28 (10^6) = \frac{4.40(10^3)}{\frac{\pi}{4}(d_{\pi}^2 - 0.00611^2)}$$

$$d_{-} = 0.0154 \,\mathrm{m} = 15.4 \,\mathrm{mm}^{-}$$
 Ans

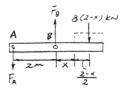
1–106. The bar is held in equilibrium by the pin supports at A and B. Note that the support at A has a single leaf and therefore it involves single shear in the pin, and the support at B has a double leaf and therefore it involves double shear. The allowable shear stress for both pins is  $\tau_{\rm allow} = 150 \, \text{MPa}$ . If a uniform distributed load of w = 8 kN/m is placed on the bar, determine its minimum

allowable position x from B. Pins A and B each have a diameter of 8 mm. Neglect any axial force in the bar.



$$f = \sum M_A = 0;$$
  $F_B(2) - 8(2 - x)(3 + \frac{x}{2}) = 0$ 

$$\begin{cases} + \Sigma M_A = 0; & F_B(2) - 8(2 - x)(3 + \frac{x}{2}) = 0 \\ 2F_B - 48 + 16x + 4x^2 = 0 \end{cases}$$
 (1) 
$$\begin{cases} + \Sigma M_B = 0; & F_A(2) - 8(2 - x)(\frac{x}{2} + 1) = 0 \\ 2F_A - 16 + 4x^2 = 0 \end{cases}$$
 (2)



Assume failure of pin A

$$\tau_{\text{allow}} = \frac{F_A}{A_A}; \quad 150(10^6) = \frac{F_A}{\frac{\pi}{4}(0.008)^2}$$

$$F_A = 7539.8 \,\mathrm{N} = 7.5398 \,\mathrm{kN}$$



Substitute  $F_A = 7.5398 \text{ kN}$  into Eq. (2), x = 0.480 m

Assume failure of pin B

$$\tau_{\text{allow}} = \frac{\frac{F_B}{2}}{A_B}; \quad 150(10^6) = \frac{\frac{F_B}{2}}{\frac{\pi}{4}(0.008)^2}$$

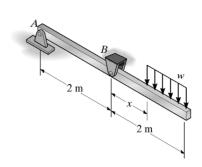
$$F_B = 15079.6 \,\mathrm{N} = 15.0796 \,\mathrm{kN}$$

Substitute 
$$F_B = 15.0796 \text{ kN}$$
 into Eq. (1),  $x = 0.909 \text{ m}$ 

Choose the larger 
$$x = 0.909 \text{ m}$$
 Ans



**1–107.** The bar is held in equilibrium by the pin supports at A and B. Note that the support at A has a single leaf and therefore it involves single shear in the pin, and the support at B has a double leaf and therefore it involves double shear. The allowable shear stress for both pins is  $\tau_{\text{allow}} = 125 \text{ MPa}$ . If x = 1 m, determine the maximum distributed load w the bar will support. Pins A and B each have a diameter of 8 mm. Neglect any axial force in the bar.



For pin A,

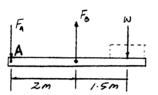
$$\tau_{\text{allow}} = \frac{F_A}{A_A}$$
;  $125(10^6) = \frac{0.75w}{\frac{\pi}{4}(0.008)^2}$ 

$$w = 8377 \text{ N/m} = 8.38 \text{ kN/m}$$

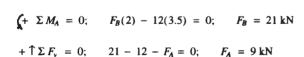


$$\tau_{\text{allow}} = \frac{\frac{F_B}{2}}{A_B}; \qquad 125(10^6) = \frac{\frac{1.75 \, \text{w}}{2}}{\frac{\pi}{4}(0.008)^2}$$

$$w = 7181 \text{ N/m} = 7.18 \text{ kN/m} \text{ (controls)}$$
 Ans



\*1–108. The bar is held in equilibrium by the pin supports at A and B. Note that the support at A has a single leaf and therefore it involves single shear in the pin, and the support at B has a double leaf and therefore it involves double shear. The allowable shear stress for both pins is  $\tau_{\text{allow}} = 125 \text{ MPa}$ . If x = 1 m and w = 12 kN/m, determine the smallest required diameter of pins A and B. Neglect any axial force in the bar.



For pin A

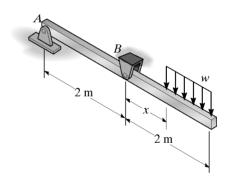
$$\tau_{\text{allow}} = \frac{F_A}{A_A}; \quad 125(10^6) = \frac{9(10^3)}{\frac{\pi}{4}(d_A)^2}$$

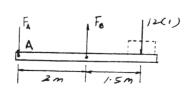
$$d_A = 0.00957 \text{ m} = 9.57 \text{ mm}$$
 Ans



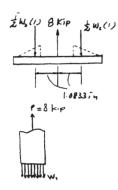
$$\tau_{\text{allow}} = \frac{\frac{F_B}{2}}{A_B}; \qquad 125(10^6) = \frac{\frac{21(10^3)}{2}}{\frac{\pi}{4}(d_B)^2}$$

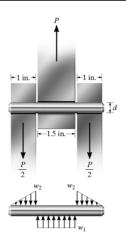
$$d_B = 0.0103 \text{ m} = 10.3 \text{ mm}$$
 Ans





1-109. The pin is subjected to double shear since it is used to connect the three links together. Due to wear, the load is distributed over the top and bottom of the pin as shown on the free-body diagram. Determine the diameter d of the pin if the allowable shear stress is  $\tau_{\rm allow} = 10\,{\rm ksi}$  and the load P = 8 kip. Also, determine the load intensities  $w_1$  and  $w_2$ .





Pin:

$$+ \uparrow \Sigma F_y = 0;$$
  $8 - 1.5 w_1 = 0$   
 $w_1 = 5.33 \text{ kip/in.}$  Ans

$$+ \uparrow \Sigma F_y = 0;$$
  $-2(\frac{1}{2}w_2)(1) + 8 = 0$   
 $w_2 = 8 \text{ kip/in.}$  Ans

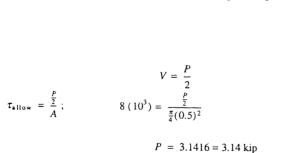
Shear stress

$$\tau_{\text{allow}} = \frac{\frac{P}{2}}{\frac{\pi}{4}(d)^2}; \qquad 10 = \frac{\frac{8}{2}}{\frac{\pi}{4}(d)^2}$$

d = 0.714 in.Ans



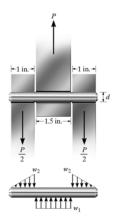
**1–110.** The pin is subjected to double shear since it is used to connect the three links together. Due to wear, the load is distributed over the top and bottom of the pin as shown on the free-body diagram. Determine the maximum load P the connection can support if the allowable shear stress for the material is  $\tau_{\rm allow} = 8$  ksi and the diameter of the pin is 0.5 in. Also, determine the load intensities  $w_1$  and  $w_2$ .



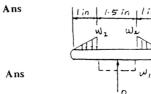
3.1416 kip = 
$$w_1$$
 (1.5)  
 $w_1 = 2.09$  kip/in.

$$\frac{3.1416}{2} = \frac{1}{2} w_2 (1)$$

$$w_2 = 3.14 \text{ kip/in.}$$

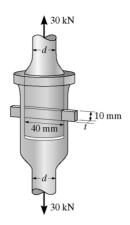






Ans

**1–111.** The cotter is used to hold the two rods together. Determine the smallest thickness t of the cotter and the smallest diameter d of the rods. All parts are made of steel for which the failure tensile stress is  $\sigma_{\text{fail}} = 500 \text{ MPa}$  and the failure shear stress is  $\tau_{\text{fail}} = 375 \text{ MPa}$ . Use a factor of safety of  $(\text{F.S.})_t = 2.50$  in tension and  $(\text{F.S.})_s = 1.75$  in shear.



Allowable Normal Stress: Design of rod size

$$\sigma_{\text{allow}} = \frac{\sigma_{\text{fail}}}{\text{F.S}} = \frac{P}{A};$$

$$\frac{500(10^6)}{2.5} = \frac{30(10^3)}{\frac{\pi}{4}d^2}$$

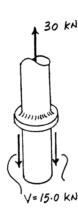
$$d = 0.01382 \text{ m} = 13.8 \text{ mm} \quad \text{Ans}$$

Allowable Shear Stress: Design of cotter size.

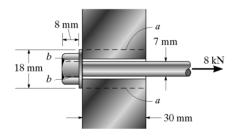
$$\tau_{\text{allow}} = \frac{\tau_{\text{fail}}}{\text{F.S}} = \frac{V}{A};$$

$$\frac{375(10^6)}{1.75} = \frac{15.0(10^3)}{(0.01)t}$$

$$t = 0.0070 \text{ m} = 7.00 \text{ mm}$$
Ans



\*1–112. The long bolt passes through the 30-mm-thick plate. If the force in the bolt shank is 8 kN, determine the average normal stress in the shank, the average shear stress along the cylindrical area of the plate defined by the section lines a–a, and the average shear stress in the bolt head along the cylindrical area defined by the section lines b–b.

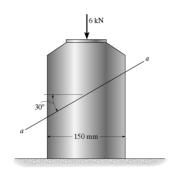


$$\sigma_s = \frac{P}{A} = \frac{8 (10^3)}{\frac{\pi}{4} (0.007)^2} = 208 \text{ MPa}$$
 Ans

$$(\tau_{avg})_a = \frac{V}{A} = \frac{8(10^3)}{\pi(0.018)(0.030)} = 4.72 \text{ MPa}$$
 Ans

$$(\tau_{avg})_b = \frac{V}{A} = \frac{8(10^3)}{\pi(0.007)(0.008)} = 45.5 \text{ MPa}$$
 Ans

**1–113.** The bearing pad consists of a 150 mm by 150 mm block of aluminum that supports a compressive load of 6 kN. Determine the average normal and shear stress acting on the plane through section a–a. Show the results on a differential volume element located on the plane.

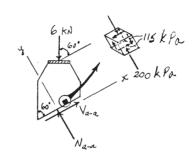


Equations of Equilibrium:

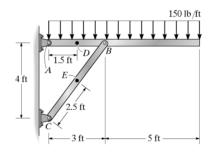
$$V_{a-a} = 0;$$
  $V_{a-a} = 6\cos 60^{\circ} = 0$   $V_{a-a} = 3.00 \text{ kN}$   
 $V_{a-a} = 0;$   $V_{a-a} = 6\sin 60^{\circ} = 0$   $V_{a-a} = 5.196 \text{ kN}$ 

Average Normal Stress And Shear Stress: The cross sectional Area at section a- a is  $A = \left(\frac{0.15}{\sin 60^{\circ}}\right)(0.15) = 0.02598 \text{ m}^2$ .

$$\sigma_{a-a} = \frac{N_{a-a}}{A} = \frac{5.196(10^3)}{0.02598} = 200 \text{ kPa}$$
 Ans
$$\tau_{a-a} = \frac{V_{a-a}}{A} = \frac{3.00(10^3)}{0.02598} = 115 \text{ kPa}$$
 Ans



**1–114.** Determine the resultant internal loadings acting on the cross sections located through points D and E of the frame.



Segment AD:

$$\stackrel{+}{\to} \Sigma F_x = 0;$$
  $N_D - 1.2 = 0;$   $N_D = 1.20 \text{ kip}$  Ans

$$+\downarrow \Sigma F_y = 0;$$
  $V_D + 0.225 + 0.4 = 0;$   $V_D = -0.625 \text{ kip}$  Ans

$$(+ \Sigma M_D = 0; M_D + 0.225(0.75) + 0.4(1.5) = 0$$

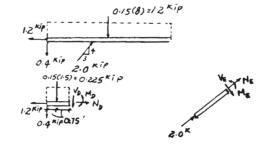
$$M_D = -0.769 \text{ kip} \cdot \text{ft}$$
 Ans

Segment CE:

$$/+\Sigma F_x = 0;$$
  $N_E + 2.0 = 0;$   $N_E = -2.00 \text{ kip}$  Ans

$$V_E = 0; V_E = 0 Ans$$

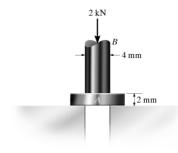
$$(+\Sigma M_E = 0; M_E = 0)$$
 Ans



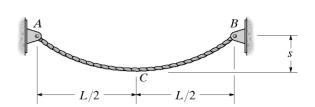
**1–115.** The circular punch B exerts a force of 2 kN on the top of the plate A. Determine the average shear stress in the plate due to this loading.

Average Shear Stress: The shear area  $A = \pi (0.004)(0.002) = 8.00(10^{-6})\pi \text{ m}^2$ 

$$\tau_{\text{avg}} = \frac{V}{A} = \frac{2(10^3)}{8.00(10^{-6})\pi} = 79.6 \text{ MPa}$$
 Ans



\*1–116. The cable has a specific weight  $\gamma$  (weight/volume) and cross-sectional area A. If the sag s is small, so that its length is approximately L and its weight can be distributed uniformly along the horizontal axis, determine the average normal stress in the cable at its lowest point C.

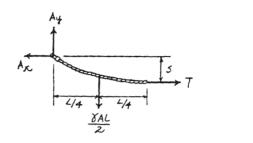


Equation of Equilibrium:

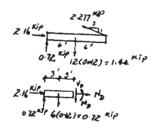
$$\begin{split} \underbrace{\left(+\Sigma M_A=0;\right.} &Ts-\frac{\gamma AL}{2} \Big(\frac{L}{4}\Big)=0 \\ &T=\frac{\gamma AL^2}{8\,s} \end{split}$$

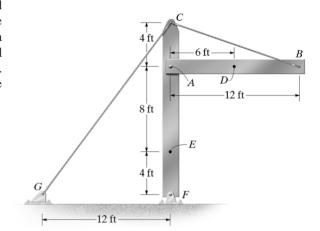
Average Normal Stress:

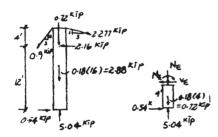
$$\sigma = \frac{T}{A} = \frac{\frac{\gamma A L^2}{8 \, r}}{A} = \frac{\gamma L^2}{8 \, s} \qquad \text{Ans}$$



**1–117.** The beam AB is pin supported at A and supported by a cable BC. A separate cable CG is used to hold up the frame. If AB weighs 120 lb/ft and the column FC has a weight of 180 lb/ft, determine the resultant internal loadings acting on cross sections located at points D and E. Neglect the thickness of both the beam and column in the calculation.







Segment BD:

$$\stackrel{\bullet}{\to} \Sigma F_x = 0; N_D + 2.16 = 0; N_D = -2.16 \text{ kip} Ans$$

$$+ \downarrow \Sigma F_y = 0; V_D + 0.72 - 0.72 = 0; V_D = 0 Ans$$

 $M_D - 0.72(3) = 0;$ 

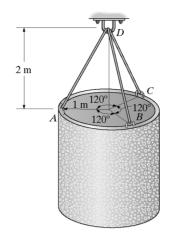
 $(+\Sigma M_D = 0;$ Segment FE:

$$\leftarrow \Sigma F_x = 0; \qquad V_E - 0.54 = 0; \qquad V_E = 0.540 \text{ kip}$$

$$+ \downarrow \Sigma F_y = 0; \qquad N_E + 0.72 - 5.04 = 0; \qquad N_E = 4.32 \text{ kip}$$

$$+ \Delta M_E = 0; \qquad -M_E + 0.54(4) = 0; \qquad M_E = 2.16 \text{ kip} \cdot \text{ft}$$
Ans

**1–118.** The 3-Mg concrete pipe is suspended by the three wires. If BD and CD have a diameter of 10 mm and AD has a diameter of 7 mm, determine the average normal stress in each wire.



Equations of Equilibrium:

$$\Sigma M_x = 0;$$
  $F_{BD} (1\sin 60^\circ) - F_{CD} (1\sin 60^\circ) = 0$   $F_{BD} = F_{CD} = F$ 

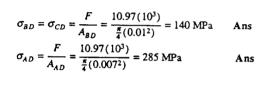
$$\Sigma M_y = 0;$$
  $2F(1\cos 60^\circ) - F_{AD}(1) = 0$ 

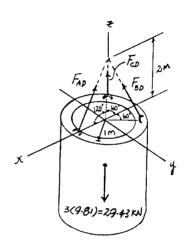
$$F_{AD} = F$$

$$\Sigma F_z = 0; \qquad 3 \left[ F \left( \frac{2}{\sqrt{5}} \right) \right] - 29.43 = 0$$

$$F = 10.97 \text{ kN}$$

Average Normal Stress:





**1–119.** The yoke-and-rod connection is subjected to a tensile force of 5 kN. Determine the average normal stress in each rod and the average shear stress in the pin A between the members.

For the 40 - mm - dia. rod:

$$\sigma_{40} = \frac{P}{A} = \frac{5 (10^3)}{\frac{\pi}{4} (0.04)^2} = 3.98 \text{ MPa}$$
 Ans

For the 30-mm-dia. rod:

$$\sigma_{30} = \frac{V}{A} = \frac{5 (10^3)}{\frac{\pi}{4} (0.03)^2} = 7.07 \text{ MPa}$$
 Ans

Average shear stress for pin A:

$$\tau_{\text{avg}} = \frac{P}{A} = \frac{2.5 (10^3)}{\frac{\pi}{4} (0.025)^2} = 5.09 \,\text{MPa}$$
 Ans

