

# Modeling humans

For normal control systems, we already have good mathematical descriptions. (Think, for example, of linear control theory.) But when humans are involved, this becomes more complicated. One way to solve this problem is by trying to model humans as normal control systems. By doing this, we can explain and predict the behavior of the human and the system. In this chapter, we'll look at how it works.

## 1 A human as a linear controller

### 1.1 The human pilot

The human pilot is a **multimode**, **adaptive** and **learning** controller. It is capable of exhibiting an enormous variety of behavior, like...

- **System organization** – This means that the human controller can detect coherence between for example input and output signals. It then uses this coherence to control the system.
- **System adjustment** – The human controller can adjust the way in which it controls the system (i.e. its transfer function) such that the system is adequately controlled.

### 1.2 Display types

Let's examine a very basic single-input single-output feedback system. It is the task of the human controller to minimize the error  $e = y - y_{ref}$  between the output  $y$  and the reference signal  $y_{ref}$  of the system. The way in which the pilot can do this strongly depends on how the data is presented to him.

- In a **compensatory display**, only the error  $e$  is shown. The controller simply needs to compensate for this error.
- In the **pursuit display**, only the current output  $y$  and the reference signal  $y_{ref}$  are shown. This time the controller needs to let the output  $y$  pursuit the reference output  $y_{ref}$ .
- In the **preview display**, the controller doesn't only see  $y$  and  $y_{ref}$ . It also sees the future values of  $y_{ref}$ . In this way, the pilot can already keep in mind future changes of the reference signal.

Soon we will discuss the cross-over model of human behavior. This model only applies to the compensatory display. So it does not work for the other displays. Keep that in mind.

### 1.3 Human controllers as linear models

How can we model the human controller in the SISO feedback system with compensatory display? Of course, the human controller is nonlinear. However, research has shown that it can be modeled using a quasi-linear describing function. This function  $Y_p$  (known as the **causal model**) is basically a linear differential equation with constant coefficients and a time delay. It accounts for the portion of the controller's output  $c$  that is linearly related to the input  $e$ .

Of course, the describing function doesn't model the human behavior perfectly. The remaining inaccuracies are described by the **remnant model**  $n$ , which is a stationary noise process. Since this remnant model is not really understood well, we will focus on the describing function  $Y_p$ .

## 1.4 The ideal solution

If we don't use a human controller, what should our causal model  $Y_p$  be to ideally control the system? A good feedback control system provides a good relation between  $y$  and  $y_{ref}$ , it suppresses disturbances, is robust and provides adequate closed-loop stability margins. If we call the plant transfer function  $Y_c$ , then the system closed-loop transfer function is

$$\frac{y}{y_{ref}} = \frac{Y_p Y_c}{1 + Y_p Y_c} = \frac{Y_{OL}}{1 + Y_{OL}}, \quad (1.1)$$

with  $Y_{OL}$  the open loop transfer function. To have  $y/y_{ref} \approx 1$ , we can make  $|Y_{OL}|$  very big for the input bandwidth and very small for the other frequencies. This normally works well, unless some lag is also added. In this case, we often wind up with an unstable system.

To solve this problem, we must make sure that the gain crossover frequency  $\omega_c$  exceeds the maximum input frequency  $\omega_i$ . (Remember that  $\omega_c$  satisfies  $|Y_{OL}(j\omega_c)| = 1$ .) So within the bandwidth,  $|Y_{OL}|$  is roughly constant, and much bigger than 1. Near the crossover frequency,  $|Y_{OL}|$  will have a slope of  $-20\text{dB/decade}$ . Having this kind of open-loop transfer function will result in a stable system with small tracking errors. The value of  $Y_{OL}$  near  $\omega_c$  now determines the dominant closed loop modes. In fact, the system stability strongly depends on the phase at  $\omega_c$ . (That is, the phase margin.)

## 2 Ways of modeling the pilot

### 2.1 The crossover model

Previously, we haven't answered the question what  $Y_p$  will be if we use a human controller. This is where the crossover model theorem comes in. The **crossover model theorem** states that human controllers adjust their control behavior to the dynamics of the controlled element. They do this such that the open loop transfer function in the crossover region (i.e. with  $\omega$  near  $\omega_c$ ) can be described by

$$Y_{OL}(j\omega) = Y_p(j\omega)Y_c(j\omega) = \frac{\omega_c}{j\omega} e^{-j\omega\tau_e}. \quad (2.1)$$

Here,  $\omega_c$  is still the crossover frequency.  $\tau_e$  is an **effective time delay**, which represents the time lags of the human operator. It is now interesting to note that, if we know the dynamics of the system, then we can find the transfer function of the human operator near  $\omega_c$ . It is given by

$$Y_p(j\omega) = \frac{\omega_c}{Y_c(j\omega)j\omega} e^{-j\omega\tau_e}. \quad (2.2)$$

What does this imply for the performance and the stability of the system? Well, the **performance** is determined by  $\omega_c$ , while the **stability** is determined by the phase margin  $\phi_m$ . The crossover model now implies that  $\phi_m = \frac{\pi}{2} - \omega_c\tau_e$ . This means that, if the time lag  $\tau_e$  is high, we cannot have a high crossover frequency  $\omega_c$ . (Or instability can occur.) So big time lags lower the performance of the system.

### 2.2 A pilot describing function

Another way to model the pilot is by using the **simplified precision model**. In this model, we model the pilot as

$$Y_p(j\omega) = K_p \frac{1 + \tau_L j\omega}{1 + \tau_I j\omega} e^{-j\omega\tau_e}, \quad (2.3)$$

where  $K_p$  is the **pilot gain**,  $\tau_L$  is the **lead time constant**,  $\tau_I$  is the **lag time constant** and  $\tau_e$  is the **effective time delay**. Basically, the pilot gain  $K_p$  puts the crossover frequency  $\omega_c$  on the right position.

The time constants  $\tau_L$  and  $\tau_I$  are then used to give  $Y_{OL}$  a slope of  $-20\text{dB/decade}$  near the crossover frequency. The question remains how to set these parameters. There are 6 so-called **verbal adjustment rules** for that.

1. In **equalization selection and adjustment**, we want  $|Y_{OL}|$  to be very big for low frequencies. For frequencies near the crossover frequency, we want the slope to be  $-20\text{dB/decade}$ .
2. Within the limitations of the pilot, we should minimize errors. The most important way to do this is to minimize  $\tau_e$  as far as possible. How much this can be done depends on the input frequency  $\omega_i$ . If  $\omega_i \rightarrow 0$ , then  $\tau_e$  becomes the **basic time delay**  $\tau_0$ . Also, the corresponding crossover frequency is denoted by  $\omega_{c_0}$ .
3. We have to set the right crossover frequency. A good estimate is

$$\omega_{c_0} = \frac{\pi}{2\tau_0}. \quad (2.4)$$

4. If  $\omega_i < 0.8\omega_{c_0}$ , then changes in the input frequency  $\omega_i$  don't really influence the crossover frequency  $\omega_c$ . However, if  $\omega_i$  becomes bigger than  $0.8\omega_{c_0}$ , then resonance may occur. To prevent this, the pilot should choose  $\omega_c$  much lower than  $\omega_{c_0}$ . This is called **regression**. It simply means that the pilot does not try to follow the high-frequency signals. If he does, resonance would occur and the error would be bigger.
5. Once the parameters have been set, we can approximate the squared error using the **one-third law**

$$\frac{e^2}{y_{ref}^2} = \frac{1}{3} \left( \frac{\omega_i}{\omega_c} \right)^2. \quad (2.5)$$

However, this law is only valid when  $\omega_i$  is much smaller than  $\omega_c$ .

6. The time delay  $\tau_e$  and the optimal crossover frequency  $\omega_c$  depend on the bandwidth  $\omega_i$ . If  $\omega_i \approx 0$ , then the values mainly depend on the controlled system  $Y_c$ . However, the changes in  $\tau_e$  (if  $\omega_i$  is different from zero) mainly depend on  $\omega_i$ . So we can write

$$\tau_e(Y_c, \omega_i) \approx \tau_0(Y_c) - \Delta\tau_e(\omega_i). \quad (2.6)$$

Furthermore, we have

$$\omega_c(Y_c, \omega_i) \approx \omega_{c_0}(Y_c) + 0.18\omega_i. \quad (2.7)$$

With these verbal adjustment rules, we can get a pilot model without doing any experiments. This is very convenient for initial pilot-in-the-loop experiments.