

CHAPTER 4

$$4.1 \quad q_{\infty} = \frac{1}{2} \rho_{\infty} V_{\infty}^2 = \frac{1}{2} (0.002377)(50)^2 = 2.97 \text{ lb/ft}^2$$

$$c_{\ell} = 0.64 \quad \text{and} \quad c_{m_{c/4}} = -0.036$$

$$L' = q_{\infty} S c_{\ell} = (2.97)(2)(1)(0.64) = \boxed{3.80 \text{ lb per unit span}}$$

$$M'_{c/4} = q_{\infty} S c c_{m_{c/4}} = (2.97)(2)(1)(2)(-0.036) = \boxed{-0.428 \text{ ft/lb per unit span}}$$

$$4.2 \quad q_{\infty} = \frac{1}{2} \rho_{\infty} V_{\infty}^2 = \frac{1}{2} (1.23)(50)^2 = 1538 \text{ N/m}^2$$

$$c_{\ell} = \frac{L'}{q_{\infty} S} = \frac{1353}{(1538)(2)} = 0.44$$

From Fig. 4.5,

$$\boxed{\alpha = 2^{\circ}}$$

$$4.3 \quad \Gamma = \oint_c \vec{V} \cdot d\vec{s}$$

$$\frac{D\Gamma}{Dt} = \oint_c \frac{D\vec{V}}{Dt} \cdot d\vec{s} + \oint_c \vec{V} \cdot d\vec{s}$$

$$\frac{Dd\vec{s}}{Dt} = d\vec{V}$$

Hence, the second term in Eq. (1) becomes

$$\oint_c \vec{V} \cdot d\vec{V} = \oint_c d\left(\frac{V^2}{2}\right) = 0$$

From the momentum equation,

$$\frac{D\vec{V}}{Dt} = -\frac{1}{\rho} \nabla p \text{ (neglecting body forces)}$$

Hence, the first term in Eq. (1) becomes

$$\oint_c \frac{D\vec{V}}{Dt} \cdot \vec{ds} = - \oint_c \frac{1}{\rho} \nabla p \cdot \vec{ds} = - \oint_c \frac{dp}{\rho}$$

When $\rho = \text{const}$, or $\rho = \rho(p)$, then

$$- \oint_c \frac{dp}{\rho} = 0. \text{ Hence, from Eq. (3)}$$

$$\oint_c \frac{D\vec{V}}{Dt} \cdot \vec{ds} = 0 \tag{4}$$

Substituting Eqs. (2) and (4) into (1), we obtain

$$\boxed{\frac{D\Gamma}{Dt} = 0}$$

Note: See Karamcheti, Ideal-Fluid Aerodynamics, for more details (pp. 239-242).

$$\begin{aligned} 4.4 \quad M'_{LE} &= -\rho_\infty V_\infty \int_0^c \xi \gamma(\xi) d\xi \\ &= -\rho_\infty V_\infty \int_0^\pi \frac{c}{2} (1 - \cos\theta) \theta (\gamma) \frac{c}{2} \sin\theta d\theta \\ &= -\rho_\infty V_\infty \frac{c^2}{4} 2\alpha V_\infty \int_0^\pi (1 - \cos^2\theta) d\theta \\ &= -\rho_\infty V_\infty \frac{c^2}{2} \alpha \left[\frac{\pi}{2} \right] = -\left(\frac{1}{2} \rho_\infty V_\infty^2 \right) c^2 \frac{\pi\alpha}{2} \\ &= -q_\infty c^2 \frac{\pi\alpha}{2} \quad \text{This is Eq. (4.36).} \end{aligned}$$

4.5 $c_l = 2\pi\alpha$ where α is in radians. Hence

$$c_\ell = 2\pi \left(\frac{1.5}{57.3} \right) = \boxed{0.164}$$

$$c_{m,tc} = -c_\ell/r = \boxed{-0.041}$$

4.6 (a)

$$\text{For } 0 \leq \frac{x}{c} \leq 0.4: \left(\frac{dz}{dx} \right)_1 = 0.2 - 0.5 \left(\frac{x}{c} \right)$$

$$\text{For } 0.4 \leq \frac{x}{c} \leq 1: \left(\frac{dz}{dx} \right)_2 = 0.0888 - 0.2222 \left(\frac{x}{c} \right)$$

Since $x = \frac{c}{2} (1 - \cos\theta)$, then

$$\left(\frac{dz}{dx} \right)_1 = -0.05 + 0.25 \cos\theta, \text{ for } 0 \leq \theta \leq 1.3694$$

$$\left(\frac{dz}{dx} \right)_2 = -0.0223 + 0.1111 \cos\theta, \text{ for } 1.3694 \leq \theta \leq \pi$$

$$\alpha_{L=0} = -\frac{1}{\pi} \int_0^\pi \frac{dz}{dx} (\cos\theta - 1) d\theta$$

$$= \frac{1}{\pi} \int_0^{1.3694} (-0.05 + 0.25 \cos\theta)(\cos\theta - 1) d\theta - \frac{1}{\pi} \int_{1.3694}^\pi$$

$$(-0.0223 + 0.1111 \cos\theta)(\cos\theta - 1) d\theta$$

$$= -\frac{1}{\pi} \int_0^{1.3694} (0.05 - 0.3 \cos\theta + 0.25 \cos^2\theta) d\theta$$

$$- \frac{1}{\pi} \int_{1.3694}^\pi (0.0223 - 0.13334 \cos\theta + 0.1111 \cos^2\theta) d\theta$$

$$= -\frac{1}{\pi} \left[0.05\theta - 0.3 \sin\theta + 0.25 \left(\frac{\theta}{2} + \frac{1}{4} \sin 2\theta \right) \right]_0^{1.3694}$$

$$\begin{aligned}
& -\frac{1}{\pi} [0.0223 \theta - 0.1334 \sin\theta + 0.111 \left(\frac{\theta}{2} + \frac{1}{4} \sin 2\theta\right)]_{1.3694}^{\pi} \\
& = -\frac{1}{\pi} [0.06847 - 0.2939 + 0.1712 + 0.0245] - \frac{1}{\pi} [0.0701 + 0.1745] \\
& \quad + \frac{1}{\pi} [0.0305 - 0.1307 + 0.0761 + 0.0109] \\
& = \frac{-0.2281}{\pi} = -0.0726 \text{ rad} = \boxed{-4.16^\circ}
\end{aligned}$$

(b)

$c_t = 2\pi(\alpha + \alpha_{l=0})$ where α is in radians

$$c_t = \frac{2\pi}{57.3} [3 - (-4.16)] = \boxed{0.782}$$

$$\begin{aligned}
4.7 \quad A_1 &= \frac{2}{\pi} \int_0^{\pi} \frac{dz}{dx} \cos\theta \, d\theta \\
&= \frac{2}{\pi} \int_0^{1.3694} (0.05 + 0.25 \cos\theta) \cos\theta \, d\theta \\
& \quad + \frac{2}{\pi} \int_{1.3694}^{\pi} (-0.0223 + 0.1111 \cos\theta) \cos\theta \, d\theta \\
&= \frac{2}{\pi} \int_0^{1.3694} (-0.05 \cos\theta + 0.25 \cos^2\theta) \, d\theta + \\
& \quad \frac{2}{\pi} \int_{1.3694}^{\pi} (-0.0223 \cos\theta + 0.1111 \cos^2\theta) \, d\theta \\
&= \frac{2}{\pi} [-0.05 \sin\theta + 0.25 \left(\frac{\theta}{2} + \frac{1}{4} \sin 2\theta\right)]_0^{1.3694} \\
& \quad + \frac{2}{\pi} [(-0.0233) \sin\theta + 0.1111 \left(\frac{\theta}{2} + \frac{1}{4} \sin 2\theta\right)]_{1.3694}^{\pi}
\end{aligned}$$

$$= \frac{2}{\pi} [-0.04899 + 0.25 (0.6847 + 0.09800) + 0.1745 \\ + 0.02185 - 0.1111 (0.6847 + 0.09800)]$$

$$A_1 = (0.2561) \frac{2}{\pi} = 0.1630$$

$$A_2 = \frac{2}{\pi} \int_0^{\pi} \frac{dz}{dx} \cos 2\theta \, d\theta$$

$$= \frac{2}{\pi} \int_0^{1.3694} (-0.05 + 0.25 \cos\theta) \cos 2\theta \, d\theta + \frac{2}{\pi} \int_{1.3694}^{\pi} (-0.0223$$

$$+ 0.1111 \cos\theta) \cos\theta \, d\theta$$

$$= \frac{2}{\pi} \left[\frac{1}{2} (-0.05) \sin 2\theta + 0.25 \left(\frac{\sin \theta}{2} + \frac{\sin 3\theta}{6} \right) \right]_0^{1.3694}$$

$$+ \frac{2}{\pi} \left[\frac{1}{2} (-0.0223) \sin 2\theta + 0.1111 \left(\frac{\sin \theta}{2} + \frac{\sin 3\theta}{6} \right) \right]_{1.3694}^{\pi}$$

$$= \frac{2}{\pi} [-0.009800 + 0.25 (0.4899 - 0.1372) + 0.004371$$

$$- 0.1111 (0.4899 - 0.1372)]$$

$$= (0.0436) \frac{2}{\pi} - 0.0277$$

$$c_{m_{c_f}} = \frac{\pi}{4} (A_2 - A_1) = \frac{\pi}{4} (0.0277 - 0.1630) = \boxed{-0.1063}$$

$$\frac{x_{cp}}{c} = \frac{1}{4} \left[1 + \frac{\pi}{c_f} (A_1 - A_2) \right] = \frac{1}{4} \left[1 + \frac{\pi}{0.782} (0.1630 - 0.0277) \right] = \boxed{0.386}$$

4.8

	<u>Experiment (Ref. 11)</u>	<u>Theory</u>	<u>% Difference</u>
$\alpha_{L=0}$	-3.9°	-4.16°	6.25%
c_t	0.76	0.782	2.8%
$c_{m_{c/4}}$	-0.095	-0.1063	10.6%

$$4.9 \quad M'_{LE} = -\rho_{\infty} V_{\infty} \int_0^c \xi \gamma(\xi) d\xi$$

$$c_{m_{c/4}} = \frac{M'_{LE}}{\frac{1}{2} \rho_{\infty} V_{\infty}^2 c^2} = \frac{-2}{V_{\infty} c^2} \int_0^c \xi \gamma(\xi) d\xi \quad (1)$$

$$\xi = \frac{c}{2} (1 - \cos\theta)$$

$$d\xi = \frac{c}{2} \sin\theta d\theta$$

$$\gamma(\theta) = 2 V_{\infty} \left[A_0 \frac{(1 + \cos\theta)}{\sin\theta} + \sum_{n=1}^8 A_n \sin n\theta \right]$$

With the above, Eq. (1) becomes

$$c_{m_{c/4}} = - \int_0^{\pi} A_0 (1 - \cos^2\theta) d\theta - \sum_{n=1}^8 \int_0^{\pi} A_n (1 - \cos\theta) \sin\theta \sin n\theta d\theta \quad (2)$$

Note the following definite integrals:

$$\int_0^{\pi} \cos^2\theta d\theta = \frac{\pi}{2}$$

$$\int_0^{\pi} \sin^2\theta d\theta = \frac{\pi}{2}$$

$$\int_0^{\pi} \cos\theta \sin^2\theta d\theta = 0$$

$$\int_0^{\pi} \sin\theta \sin n\theta \, d\theta = 0 \quad \text{for } n = 2, 3, \dots$$

$$\int_0^{\pi} \cos\theta \sin\theta \sin 2\theta \, d\theta = \frac{\pi}{4}$$

$$\int_0^{\pi} \cos\theta \sin\theta \sin n\theta \, d\theta = 0 \quad \text{for } n = 3, 4, \dots$$

Hence, Eq. (2) becomes:

$$c_{m_{te}} = -\left[\pi A_0 - \frac{\pi}{2} A_0 + \frac{\pi}{2} A_1 - \frac{\pi}{4} A_2\right]$$

$$c_{m_{te}} = -\frac{\pi}{2} \left(A_0 + A_1 - \frac{A_2}{2}\right)$$

4.10 The slope of the lift curve is

$$a_0 = \frac{0.65 - (-0.39)}{4 - (-6)} = 0.104 \text{ per degree}$$

The slope of the moment coefficient curve is

$$m_0 = \frac{-0.037 - (-0.045)}{4 - (-6)} = 8 \times 10^{-4} \text{ per degree}$$

From Eq. (4.71),

$$x_{ac} = -\frac{m_0}{a_0} + 0.25 = -\frac{8 \times 10^{-4}}{0.104} + 0.25 = \boxed{0.242}$$